

AD A 053561

2

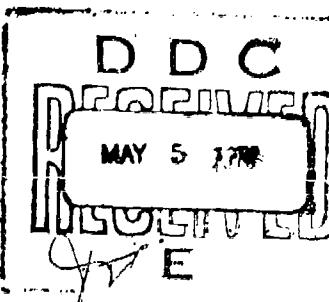


RADC-TR-78-55  
Final Technical Report  
March 1978

ELECTRONIC EQUIPMENT SCREENING AND DEBUGGING  
TECHNIQUES

R. E. Schafer  
S. P. Gray  
L. E. James  
E. A. McMillan

Hughes Aircraft Company



AU NO. 200 FILE COPY

Approved for public release; distribution unlimited.

ROME AIR DEVELOPMENT CENTER  
Air Force Systems Command  
Griffiss Air Force Base, New York 13441

This report has been reviewed by the RADC Information Office (OI) and is releasable to the National Technical Information Service (NTIS). At NTIS it will be releasable to the general public, including foreign nations.

RADC-TR-78-55 has been reviewed and is approved for publication.

APPROVED: *Eugene Fiorentino*

EUGENE FIORENTINO  
Project Engineer

APPROVED: *Joseph J. Naresky*

JOSEPH J. NARESKY  
Chief, Reliability & Compatibility Division

ADDRESSING FORM	
PERIOD	ONE MONTH
MAIL	<input checked="" type="checkbox"/>
TELEGRAMS	<input type="checkbox"/>
TELETYPE	<input type="checkbox"/>
NOTIFICATION	<input type="checkbox"/>
DISTRIBUTOR/AVAILABILITY BOARD	
MAIL	AT&T, NY/US MAIL
A	

FOR THE COMMANDER:

*John P. Huss*

JOHN P. HUSS  
Acting Chief, Plans Office

If your address has changed or if you wish to be removed from the RADC mailing list, or if the addressee is no longer employed by your organization, please notify RADC (RBRT) Griffiss AFB NY 13441. This will assist us in maintaining a current mailing list.

Do not return this copy. Retain or destroy.

(18) RADC

(19) TR-78-55

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

REPORT DOCUMENTATION PAGE		READ INSTRUCTIONS BEFORE COMPLETING FORM	
1. REPORT NUMBER RADC-TR-78-55	2. GOVT ACCESSION NO.	3. RECIPIENT'S CATALOG NUMBER	
4. TITLE (and subtitle) ELECTRONIC EQUIPMENT SCREENING AND DEBUGGING TECHNIQUES		5. TYPE OF REPORT & PERIOD COVERED Final Technical Report. Aug 1976 - Sep 1977	
6. AUTHOR(s) R. E. Schafer, E. A. McMillan S. P. Gray, L. E. James		7. PERFORMING ORGANIZATION REPORT NUMBER FR-77-16-995	
8. CONTRACT OR GRANT NUMBER(s) F30602-76-C-0395		9. PERFORMING ORGANIZATION NAME AND ADDRESS Hughes Aircraft Company, Ground Systems Group P.O. Box 3310 Fullerton CA 92634	
10. CONTROLLING OFFICE NAME AND ADDRESS Rome Air Development Center (RBRT) Griffiss AFB NY 13441		11. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBER 62702F 23380205	
12. REPORT DATE Mar 1978		13. NUMBER OF PAGES 112	
14. MONITORING AGENCY NAME & ADDRESS (if different from Controlling Office) Same		15. SECURITY CLASS. (of this report) UNCLASSIFIED	
		15a. DECLASSIFICATION/DOWNGRADING SCHEDULE N/A	
16. DISTRIBUTION STATEMENT (of this Report) Approved for public release; distribution unlimited.			
17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different from Report) Same			
18. SUPPLEMENTARY NOTES RADC Project Engineer: Eugene Fiorentino (RBRT)			
19. KEY WORDS (Continue on reverse side if necessary and identify by block number) Screening/Debugging Tests Reliability Operations Research Statistical Analysis			
20. ABSTRACT (Continue on reverse side if necessary and identify by block number) This report presents the results of a quantitative investigation into the optimum selection of screening and debugging methods for the purpose of removing defects, and hence improving reliability, throughout the development process.  A computerized optimization model has been developed which can be used to address three problem areas;			

172 370

Hue

**UNCLASSIFIED**

SECURITY CLASSIFICATION OF THIS PAGE(When Data Entered)

- 1) Select an optimum sequence of screens for a fixed dollar expenditure,
- ii) Select a sequence of screens which minimizes costs in the face of a fixed desired reliability improvement, and
- iii) Provide quantitative relationships for performing screening/ debugging tradeoffs.

In order to implement the model it was necessary to obtain data to compute test strengths and test costs. The model includes four assembly levels, five types of screens (temperature cycling, constant temperature, vibration constant power and power cycling), provides for introduction of defects at each assembly level and includes the rework cycle.

**UNCLASSIFIED**

SECURITY CLASSIFICATION OF THIS PAGE(When Data Entered)

## TABLE OF CONTENTS

	Page
<b>SECTION 0.0 INTRODUCTION . . . . .</b>	<b>1</b>
<b>SECTION 1.0 SUMMARY OF RESULTS . . . . .</b>	<b>2</b>
<b>SECTION 2.0 DATA SOURCES . . . . .</b>	<b>5</b>
2.1 Literature Survey . . . . .	5
2.2 Summary of the Results of the Literature Survey . . . . .	7
2.3 Industry Survey . . . . .	9
2.4 Internal Data . . . . .	18
<b>SECTION 3.0 DESCRIPTION OF IMPORTANT MODEL VARIABLES . . . . .</b>	<b>22</b>
3.1 Role of Test Strength in the Screening and Debugging Optimization Model . . . . .	22
3.2 Interval Estimates for the Number of Defects Remaining – Monitoring/Controlling the Screening Process . . . . .	24
3.3 Role of Part Level Screens . . . . .	25
3.4 Determination of Test Strength . . . . .	27
3.5 Determination of Initial and Final Defects . . . . .	29
<b>SECTION 4.0 DATA ANALYSIS FOR MODEL PARAMETERS . . . . .</b>	<b>31</b>
4.1 Computation of Test Strength for Power Applied . . . . .	31
4.2 Test Strength for Constant Temperature, Temperature Cycling, Vibration, and Combined Screens . . . . .	37
4.3 Conversion of MTBF to Defects . . . . .	38
4.4 Cost Data and Cost Equations . . . . .	41
4.5 Determining Relationship of Number of Subcontractors and Percent Non-Standard Parts to Test Strength . . . . .	42
<b>SECTION 5.0 COST/STRENGTH MODELS . . . . .</b>	<b>44</b>
5.1 Manufacturing Process . . . . .	44
5.2 Optimization Algorithm . . . . .	48
<b>SECTION 6.0 PROCEDURE AND EXAMPLES . . . . .</b>	<b>54</b>
6.1 Procedure for Using the SDO Model . . . . .	54
6.2 Fixed Cost - Optimum Screen . . . . .	61
6.3 Fixed Reliability - Minimum Cost . . . . .	66
<b>SECTION 7.0 RECOMMENDATIONS AND CONCLUSIONS . . . . .</b>	<b>79</b>

TABLE OF CONTENTS (Continued)

	Page
SECTION 8.0 BIBLIOGRAPHY AND REFERENCES.....	80
APPENDIX A.....	84
A.1 UCLA Biomedical Computer Program.....	84
A.2 General Curve Plotting.....	84
A.3 Multiple Linear Regression: Least Squares Fit and Analysis....	84
A.4 Least Squares Curve-fit.....	84
A.5 SDO Computer Program Printout.....	85

## LIST OF FIGURES

Figure		Page
2.1	Guidelines for Screening/Debugging Form . . . . .	10
2.2	Screening and Debugging Test Form . . . . .	11
4.1	Power Test Strength. . . . .	33
5.1	Generalized Manufacturing Assembly - Test-Rework Process . . . . .	45
5.2	General Manufacturing Process Equations . . . . .	47
5.3	General Manufacturing Process Equations - (Alternate Retest Policy) . . . . .	49
5.4	Computation Procedure for the Optimization Algorithm . . . . .	51
5.5	Flow Diagram of Optimization Procedure. . . . .	53

## LIST OF TABLES

Table	Page
1. 1 Input Data Requirements/Options for SDO Model . . . . .	3
2. 1 Survey Data Distribution . . . . .	13
2. 2 Companies Contacted . . . . .	14
2. 3 Industry Survey Data . . . . .	17
2. 4 Hughes Internal Data . . . . .	19
4. 1 Questionnaire Data Points . . . . .	32
4. 2 Internal Data Points . . . . .	34
4. 3 Reliability Growth Data . . . . .	35
4. 4 Test Strength as a Function of Several Variables . . . . .	37
4. 5 Conversion of MTBF to Defects Using Internal Data . . . . .	39
4. 6 Conversion of MTBF to Defects Using Reliability Growth Data . . . . .	40
4. 7 Man-hours Per Defect . . . . .	42
4. 8 Determining Relationship of Non-Standard Parts and Subcontractors to Test Strength . . . . .	43
6. 1 Input Data Requirements/Options . . . . .	56
6. 2 Table A - Program Data . . . . .	67
6. 3 Table B - Test and Assembly Data . . . . .	67
6. 4 Table C - Detection Probabilities for Test Equipment . . . . .	67
6. 5 Table D - Fraction of Defects Requiring Primary Level (Card Assy) Rework . . . . .	67
6. 6 Table E - Fraction of Defects Removed at the Primary Level Per Rework/Retest Cycle . . . . .	68
6. 7 Table F - Fraction of Defects Removed at the Higher Assembly Level Per Rework/Retest Cycle . . . . .	68
6. 8 Table G - Primary Level Rework Cost Per Cycle . . . . .	68
6. 9 Table H - Higher Assembly Level Rework Cost Per Cycle . . . . .	69
6. 10 Test Description . . . . .	69
6. 11 Table A - Program Data . . . . .	72
6. 12 Table B - Test and Assembly Data . . . . .	72
6. 13 Table C - Detection Probabilities for Test Equipment . . . . .	72
6. 14 Table D - Fraction of Defects Requiring Primary Level (Card Assy) Rework . . . . .	72
6. 15 Table E - Fraction of Defects Removed at the Primary Level Per Rework/Retest Cycle . . . . .	73

LIST OF TABLES (Continued)

Table		Page
<b>6.16</b>	Table F -- Fraction of Defects Removed at the Higher Assembly Level Per Rework/Retest Cycle .....	73
<b>6.17</b>	Table G -- Primary Level Rework Cost Per Cycle .....	73
<b>6.18</b>	Table H -- Higher Assembly Level Rework Cost Per Cycle .....	73
<b>6.19</b>	Test Description .....	74
<b>6.20</b>	Table A -- Program Data .....	75
<b>6.21</b>	Table B -- Test and Assembly Data .....	75
<b>6.22</b>	Table C -- Detection Probabilities for Test Equipment .....	75
<b>6.23</b>	Table D -- Fraction of Defects Requiring Primary Level (Card Assy) Rework .....	75
<b>6.24</b>	Table E -- Fraction of Defects Removed at the Primary Level Per Rework/Retest Cycle .....	76
<b>6.25</b>	Table F -- Fraction of Defects Removed at the Higher Assembly Level Per Rework/Retest Cycle .....	76
<b>6.26</b>	Table G -- Primary Level Rework Cost Per Cycle .....	76
<b>6.27</b>	Table H -- Higher Assembly Level Rework Cost Per Cycle .....	76
<b>6.28</b>	Test Description .....	77

## EVALUATION

1. The objective of this study was to develop techniques which permit cost/effectiveness trade-off analyses among various screening test approaches that may be used during the development and production of electronic equipment.
2. The objectives have been satisfactorily achieved. A computerized optimization model was developed for use as a decision tool in selecting the most cost-effective test sequence from among the many possible alternatives that are presented in a given program. Although the model is rather complex, the computer program allows for optional user input and relative ease in performing sensitivity and trade-off analyses.
3. Use of the model is recommended as an aid in selecting optimum test sequences in terms of minimum expected test and rework costs and maximum screening effectiveness. In addition, users of the model are encouraged to provide feedback information which will permit further refinements in the model structure, input parameters and application procedures.

*Eugene Fiorentino*

EUGENE FIORENTINO  
R&M Engineering Techniques Section

## SECTION 0.0 - INTRODUCTION

One of the key ideas developed in reliability engineering is the realization that electronic parts, cards, assemblies, equipments and systems contain defects, due to a variety of causes such as poor parts, design errors and manufacturing errors among others, which are not readily detectable by the usual electrical tests of functional performance. Moreover, the later in system development and system use that the defects are uncovered the more costly the discovery becomes. Of course this last statement must be balanced by the fact the removal of all defects at one level (e.g., parts) does not guarantee no defects at the next (e.g., card) level because in each step of the assembly process new defect causes are introduced.

It is now a truism that by subjecting parts, cards, assemblies, equipments and systems to extremes of environment such as temperature or vibration, these previously mentioned undetectable (by common means) defects can be degraded to a detectable level and with an appropriate test setup can be detected and eliminated. These kinds of tests with the purpose of eliminating undetectable or latent defects are generally called screening or debugging (SD) tests.

Unfortunately there are a large number of possible tests that can be run in any particular situation and the methodology for selecting an optimum sequence of tests is not well developed: irrespective of whether one's standard of methodology is good applied science or ever good engineering.

The purpose of this present study is to develop a computerized screening and debugging optimization (SDO) model which includes test strength (TS) and test costs among the primary variables and which can be used to solve three problems:

- 1) For a fixed dollar amount available what is the optimum sequence of screens which yields the highest final MTBF,  $\theta_F$ .
- 2) In the face of an initial MTBF,  $\theta_I$  and a  $\theta_F$  requirement determine the sequence of screens which moves the "system" from  $\theta_I$  to  $\theta_F$ ,  $\theta_I < \theta_F$ , at least cost.
- 3) For a fixed sequence of screens, determine the cost and the test strength.

In order to solve these problems it was necessary to:

- 1) Perform a literature search to determine what types of screens are useful and feasible.
- 2) Build a data base in order to compute test strengths (TS) and test costs.
- 3) Find a way of converting MTBF to defects.
- 4) Construct a flexible, useful computerized model which truly reflects the process of and cost of removing defects from a system at its various stages or levels of assembly.

The balance of this report discusses how these four tasks were accomplished and what results were achieved.

## SECTION 1.0 - SUMMARY OF RESULTS

The basic result of the study is the model itself. It is a fairly sophisticated model yet, hopefully, it will be useful. Its utility derives partly from the fact that it is a faithful representation of the screening process: it includes rework cycles, test strengths, test cost models, allows for introduction of defects when a new level of assembly is reached and accommodates imperfect rework at the card and assembly levels. It includes four levels of assembly: card, unit, equipment and system. The reasons for omitting the part level are thoroughly discussed in Section 3.3. It accommodates temperature cycling, constant temperature, vibration, constant power and cycled power screens.

Perhaps the key feature of the model is that practically all the required input variables are already included in the model as fall-back options. That is, if the user has better information for some variables he can enter it; otherwise the model uses what we have put in it. The model is described in considerable detail in Section 5.0. However, the accompanying Table 1.1 gives a good idea of the parameters needed in the model and how the fall-back option is implemented.

There are three major test characteristics: test costs, screening strength and detection probability. The equations yielding screening strength (SS) are incorporated in the model. No fall-back option is provided. If the user desires to change the SS equations the model should be changed. Because the fixed costs of test are so highly user dependent (e.g., how is the test equipment being amortized) this variable is a user input with a zero fall-back option. The variable test costs are also a user input with functions provided in the model as a fall-back option. Detection probability is also a user input with a fall-back option of 0.75.

For the manufacturing data required all inputs are user inputs with the exception of the fixed portion of rework cost. This cost is programmed as a function of variable costs and no fall-back option is provided.

The program data portion of the input parameters consists of all user inputs with fall-back options provided for each parameter. Here "tolerance" is necessary to decide when a solution has been reached.

The output reports consist of the optional test sequence, including test parameters (e.g. 14 temperature cycles) total cost broken down by level and by test and the total TS.

TABLE 1.1. Input Data Requirements/Options for SDO Model

Data Requirements	Input	Fall-Back Option
<u>Test Characteristics</u>		
- Fixed Test Cost	• User	• Assume zero
- Variable Test Cost	• User	• Provided as function of test
- Screening Strength	Computed	
- Detection Probability	• User	• 0.75 for all tests
<u>Manufacturing Data</u>		
- Fixed Rework Cost	• Function of Variable Cost	• None
- Variable Rework Cost	• User	• Provided as function of rework level
- Fraction Defectives to "Card" Rework	• User	• Provided as function of rework level
- Fraction Detectives Removed at Card Rework	• User	• 0.5
- Fraction Defectives Removed at Assy Rework	• User	• 0.5 All levels
<u>Program Data</u>		
- Total Quantity of Parts	• User	• 1.0 percent of total parts
- Percentage Parts - Defects	• User	• 2.3 percent of total parts
- Percentage Assy - Defects	• User	• (4 - levels max/min)
- Number of Assy Levels	• User	• 1 percent
- Tolerance (E)		

TABLE 1.1. Input Data Requirements/Options and Output Report for SDO Model (Cont)

- Input Data Summarized
- Output Report
  - Optimal Test Selection
  - Test Parameter Values
  - Total Cost by Level by Test
  - Total Test Strength

Test Sequence	Parameter Value				Cost
	No. 1	No. 2	No. 3	No. 4	
Level 1					
Test 1					
•					
•					
•					
Level 2					
Test 1					
•					
•					
•					
Total Cost					

## SECTION 2.0 - DATA SOURCES

### 2.1 Literature Survey

The literature survey was conducted in order to ascertain state-of-the-art popular screens, costs, and test strengths. Little data was obtained for the variables test costs and test strengths because the data available was stated in qualitative rather than quantitative terms.

The literature survey itself consisted of obtaining literature searches from the Hughes library, National Aeronautics and Space Administration Scientific and Technical Information facilities, and the Department of Defense Documentation Center. Articles thought to be applicable to the study were reviewed and further segregated by their relevancy. The final results yielded approximately fifty (50) articles/reports of particular relevance to this study. A complete listing of the references and bibliography can be found in Section 8.0.

Several reports were detailed enough, i. e., references 51 and 52, to be incorporated into the study's main data base.

Other reports suggested guidelines to follow for specific screening procedures. For example, reference 8 supplied the following proposed guidelines for temperature cycling acceptance testing of electronic assemblies:

<u>Type of Equipment</u>	<u>No. of Temperature Cycles</u>
Simple (100 electronic parts)	1
Moderately complex (500 electronic parts)	3
Complex (2000 electronic parts)	6
Very complex (4000 electronic parts)	10

#### Temperature Range

The suggested range is  $-65^{\circ}\text{F}$  to  $131^{\circ}\text{F}$ , or as a minimum, a temperature range of at least  $160^{\circ}\text{F}$  is recommended.

#### Temperature Rate of Change

The rate of change of internal parts should fall within  $1^{\circ}\text{F}$  and  $40^{\circ}\text{F}$  per minute. The higher rates provide the best screening.

#### Temperature Soak Times

The next temperature ramp may be started when the internal parts have stabilized within  $5^{\circ}\text{F}$  of the specified temperature and the functional checks have been completed.

#### Equipment Operation

Equipment should be energized and operated during temperature cycling, except the equipment should be turned off during chamber cool-down to permit internal parts to become cold.

## Equipment Monitoring

While it is desirable to continuously monitor the equipment during the temperature cycling, cost considerations may dictate otherwise. In such cases, periodic checks plus close monitoring of the final cycles is appropriate.

## Failure Criteria

The last cycle shall be failure free. Each repair should be reviewed for the possibilities of introducing new defects into the hardware and additional temperature cycles added when appropriate. If repairs are complex or difficult to make and inspect, or many unscreened parts are used as replacements, additional cycles should be implemented as appropriate to the individual case.

Reference 51 gives a hypothetical circuit card screening program as follows, based on its experiments with a single card type:

Rated power would be applied to the cards while they were thermally cycled between temperature extremes at a rate of  $\pm 20^{\circ}\text{C}/\text{min}$ . The temperature profile would be sawtooth (i.e., no dwell). Power would not be cycled and the cards would not be vibrated. The temperature rate chosen ( $20^{\circ}\text{C}/\text{min}$ ) would cause the least solder-cracking while permitting completion of screening in the fastest possible time. The number of temperature cycles chosen would depend upon the estimated number of quality and reliability failures contained in the population, and the subsequent reliability requirement.

Reference 24 drew the following conclusions concerning the relationships between screens and field reliability.

- Those items that had the more effective burn-in testing on production units tended to have the better reliability agreement. This indicates the necessity for adequately specifying a production unit burn-in both in duration and in environmental exposure.
- Relationships were found between reliability differences and several temperature related measures for ambient cooled WRA's, including:
  - minimum ambient temperature
  - operating time at low temperature
  - maximum ambient temperature
  - temperature rate of change

indicating that the current MIL-STD-781 tests of only requiring dwells at the temperature extremes with moderate rates of change between the limits is not an adequate test. No provisions exist for evaluating the item under conditions of rapid and frequent temperature cycling.

- The significant relationships between vibration measures and reliability differences included:
  - level
  - duration

indicating that the MIL-STD-781 vibration test of requiring 10 minutes of sinusoidal vibration each hour at one non-resonant frequency between 20-60 Hz is not representative of the field environment. The test article is never exposed to those frequencies occurring in the field, that produce failures. The vibration test duration was found to be a poor representation of the accumulated field vibration time. The lack of reliability agreement was more pronounced in WRA's installed in jet aircraft where the field environment is random.

The majority of the articles dealt in generalities concerning screen effectiveness. The screens discussed varied, but could be classified into operating, environmental, e.g., temperature, vibration, and a combination of the two. The conclusions and experiences of the articles were used accordingly in the development of the concepts utilized in this report.

General conclusions can be expressed in the words of Reference 1: "The one central theme that emerges ... is that opinions and case histories vary widely within and between the product lines ...".

A summary of the literature search can be found in the following section.

## **2.2 Summary of the Results of the Literature Survey**

At the outset we must distinguish between accelerated tests in general and the screening/debugging tests of interest in this present study. Generally, accelerated tests have as their purpose the extrapolation of some reliability measure "back" from accelerated conditions to normal conditions, the motivation being that testing at normal conditions takes too long to provide statistically reliable estimates of whatever measure of reliability is selected. The purpose of screening and debugging tests is the elimination of (previously) undetectable defects.

As previously indicated the purpose of this present study is to create a model with which to optimize the screens used. For this reason the key issues for any particular screen for this study are:

- the test strength (TS) of a screen: the probability that a (previously) undetectable failure will be detected by the screen.
- the costs of the screen.

While there are literally tons of material on screening and debugging tests the literature search turned up very little data regarding the above two factors. This is not surprising since if much data existed this present study would probably not be needed. A mathematical model is a very demanding beast and cannot accept inputs like: "about 10-12 thermal cycles should do it" (see reference 8). It demands inputs like: the TS for this screen is 0.73.

However, the literature search was valuable in pointing out what screens are likely to be of use in eliminating defects and actually there is little disagreement on the matter. The useful screens appear to be:

- i) temperature cycling
- ii) constant temperature "soak"

- iii) power applied continuously
- iv) power cycled
- v) vibration
- vi) various combinations of these.

Before discussing these tests an important issue must be put in perspective: do screening tests damage good items (parts, cards, assemblies, etc.)? The general consensus from the literature search is that screens do not damage good items if the screens are used within reasonable limits. It is obvious, of course, that beyond "reasonable" limits damage could occur although the literature is, more or less, silent on what the absolute limits are. For this present study the matter is of little consequence. The SDO model is concerned with the removal of defects by screens. Test strength is defined as the (conditional) probability of detecting a latent defect. The fact that good parts may also be removed affects only the costs of the screen and if the user (of the SDO model) has a feel for this probability (of damaging a good item) he can factor it into his cost model.

The overwhelming majority of the literature regards temperature cycling as the most powerful (without regard to cost) and versatile screen; it being claimed that temperature cycling screens can detect part defects, workmanship defects and design errors. The three key variables (all of which affect TS and costs) of a temperature cycling test are

- i) number of cycles
- ii) rise time: rate of change of temperature with respect to time
- iii) temperature extremes

The literature indicates that on the order of 10 cycles are used for parts/cards and or the order of 25 cycles for assemblies and equipments. Rise times vary from about  $1^{\circ}\text{C}/\text{minute}$  to something like  $10^{\circ}\text{C}/\text{minute}$ . The temperature extremes were commonly on the order of  $-60^{\circ}\text{C}$  to  $125^{\circ}\text{C}$ .

Another popular screen is the constant (usually high) temperature soak. The two key variables of a constant temperature screen which relate to the TS and costs are the length of the soak and the temperature of the soak. The literature agrees that the same kind of defects can be eliminated by a constant temperature screen as can be eliminated by a temperature cycle screen although for equal times at equal temperatures the temperature cycle screen is thought to have a larger TS. There was no agreement at all in the literature regarding length of the constant temperature screen although 100 to 300 hours is common. Also, there is a belief, though no explicit expression of TS was found (except for the Hughes Aircraft internal work called CREDIT, reference 42) that time of soak and temperature level can be "traded off." For example, 1000 hours at  $70^{\circ}\text{C}$  is equivalent to 200 hours at  $125^{\circ}\text{C}$  in TS. Obviously there is a tradeoff since TS is a function of soak time and temperature used. That is, TS is a function of two variables and in 3-dimensional space (TS, soak time, temperature) the relationship is a surface which might have many common points TS. Generally it is agreed that for electronic items defects are not detected, that is, TS is low, until about  $50^{\circ}\text{C}$  is used. On the other hand, anything over about  $125^{\circ}\text{C}$  and certainly  $200^{\circ}\text{C}$  begins to damage good parts.

Power application is another popular screen even though by itself long times or number of cycles are required to obtain even a moderate TS. Also it is agreed cycled power is more effective than continuously applied power for equal amounts of time applied. Notwithstanding the low TS, many system level screens like RPM or Duane growth control programs are essentially power screens on large systems. Thus power screens are most effective in detecting the failure of an item to perform a function because of a design error rather than detecting bad parts or poor (as against incorrect) workmanship.

One of the more useful applications of power screens is in conjunction with temperature screens. When a part or card is operated at high temperature, probable surface contamination problems are often forced to a detectable level. This is an Arrhenius type effect. When a bias voltage is applied during high temperature screens, latent defects can be identified. This is the so-called Eyring effect.

Vibration is the final popular screen that is often used. The consensus seems to be that vibration should be random and that vibration is extremely useful for detecting solder joint, lead, connection and bond problems. There was no agreement at all on the length of the cycle or number of cycles but on the order of about 1-20 cycles was common. The levels ranged from 1g or 2g's to about 6g's.

Various combinations of the above screens can be used very effectively to locate just about any kind of latent defect. In fact, the screens used and the levels of the variables is a cost/TS tradeoff and this is what the SDO model does.

### 2.3 Industry Survey

#### 2.3.1 Questionnaire

Several approaches were discussed concerning the industry questionnaire. The final approach taken was that of a two (2) page survey which could develop the data necessary to the purposes of this study in a cost-effective manner. A copy of the questionnaire, along with the guidelines, Figures 2.1, 2.2, follow.

The response to the survey was good. A total of forty-two (42) data points was obtained from 64% of the respondents to the questionnaire. The total response to the two hundred (200) questionnaires distributed was 17%. This response is considered good in comparison to the results of other surveys trying to obtain similar information, i.e., Reliability Study Circular Electrical Connectors, RADC-TR-73-171, which obtained 21% response out of 600 surveyed, and the Bayesian Reliability Demonstration, RADC TR-71-209.

Cost data was obtained in 67% of the responses, which was good considering its proprietary nature, while some form of screen effectiveness data was found in 93% of the cases. Data was gathered at the part, card, and higher assembly levels. The distribution of the data is shown in Table 2.1.

- THE MAIN POINTS OF INTEREST ARE:
  - describe the screen as well as possible (section 4)
  - describe the effectiveness of the screen (section 5)
  - estimate costs (section 6)
- DO NOT BE RELUCTANT TO MAKE ESTIMATES
- DO NOT BE RELUCTANT TO LEAVE BLANKS

For example, in section 6 if only estimated total costs are available, do not worry about fixed and variable costs.

Figure 2.1. Guidelines For Screening/Debugging Form

HUGHES AIRCRAFT COMPANY

FULLERTON CALIFORNIA 92634

SCREENING & DEBUGGING TEST DATA FORM

1. NAME OF PERSON COMPLETING FORM \_\_\_\_\_ DATE COMPLETED \_\_\_\_\_  
ORGANIZATION (WITHIN COMPANY) \_\_\_\_\_
2. SCREENED/DEBUGGED (S/D) ITEM DESCRIPTION:
  - 2.1 PART \_\_\_\_ CARD \_\_\_\_ ASSEMBLY \_\_\_\_ EQUIPMENT \_\_\_\_ OTHER \_\_\_\_\_
  - 2.2 PART QUALITY LEVEL PROCURED: COMMERCIAL \_\_\_\_ JAN \_\_\_\_ 883 \_\_\_\_ OTHER \_\_\_\_\_
  - 2.3 ITEM NAME(e.g., I.C., Diode, Motor, etc.) \_\_\_\_ TYPE OR FUNCTION \_\_\_\_\_
  - 2.4 QUANTITY TESTED \_\_\_\_ COMPLEXITY OF ITEM; NO. OF PARTS \_\_\_\_\_
  - 2.5 PARENT "SYSTEM" NAME/TYPE \_\_\_\_\_
3. TEST BACKGROUND
  - 3.1 WHY WAS THIS ITEM SELECTED FOR SCREEN \_\_\_\_\_
  - 3.2 PROGRAM PHASE: DEVELOPMENT \_\_\_\_ PRODUCTION \_\_\_\_ OTHER \_\_\_\_\_
  - 3.3 WAS A FORMAL RELIABILITY DEMO TEST REQUIRED ON THE OVERALL PROGRAM
  - 3.4 PURPOSE OF SCREEN DETECT:  
QUALITY DEFECTS \_\_\_\_\_ MANUFACTURING DEFECTS \_\_\_\_\_  
DESIGN ERRORS \_\_\_\_\_ OTHER \_\_\_\_\_
4. TEST DESCRIPTION
  - 4.1 TEST DATES: START \_\_\_\_\_ END \_\_\_\_\_
  - 4.2 TEST CONDITIONS:
    - 4.2.1 SCREENING STRESS(ES) USED \_\_\_\_\_  
(E.G. TEMPERATURE, VIBRATION ETC.)  
IF MORE THAN ONE TYPE OF STRESS, GIVE SEQUENCE \_\_\_\_\_
    - 4.2.2 MAX STRESS, MIN STRESS, RISE/DWELL TIME WHERE APPLICABLE  
FOR EACH STRESS USED \_\_\_\_\_

FIGURE 2.2

HUGHES AIRCRAFT COMPANY

FULLERTON, CALIFORNIA 92634

4.2.3 LENGTH OF STRESS CYCLE (HOURS) \_\_\_\_\_ NUMBER OF CYCLES \_\_\_\_\_

4.2.4 STRESS EXCEED RATED REQUIRED: YES \_\_\_\_\_ NO \_\_\_\_\_

% EXCEED RATED/NOMINAL \_\_\_\_\_

4.2.5 POWER APPLIED DURING TEST: CONTINUOUSLY \_\_\_\_\_ PERIODICALLY \_\_\_\_\_

NO POWER \_\_\_\_\_

4.2.6 APPLIED POWER EXCEED RATED: YES \_\_\_\_\_ NO \_\_\_\_\_

% EXCEED RATED/NOMINAL \_\_\_\_\_

4.3 PREDOMINANT FAILURE MODES/MECHANISMS OBSERVED: \_\_\_\_\_

5. EFFECTIVENESS OF SCREEN

5.1 PASS/FAIL CRITERIA: EXCEED RATINGS \_\_\_\_\_ DRIFT \_\_\_\_\_ OTHER \_\_\_\_\_

5.2 ITEM ELECTRICALLY TESTED BEFORE SCREEN: YES \_\_\_\_\_ NO \_\_\_\_\_ % DEFECTIVE \_\_\_\_\_

5.3 % DEFECTIVE (REJECT RATE) AFTER SCREEN \_\_\_\_\_

5.4 ESTIMATE % IMPROVEMENT (OVER UNSCREENED MTBF) IN ITEM MTBF \_\_\_\_\_

5.5 ESTIMATE % IMPROVEMENT (OVER UNSCREENED) IN % DEFECTIVE \_\_\_\_\_

5.6 DO YOU FEEL THE SCREEN WAS WORTH THE MONEY: YES \_\_\_\_\_ NO \_\_\_\_\_

5.7 DO YOU FEEL THE SCREEN DESTROYED A SIGNIFICANT % OF GOOD ITEMS:

NO \_\_\_\_\_ YES \_\_\_\_\_ % GOOD ITEMS DESTROYED \_\_\_\_\_

6. COST OF SCREEN (DOLLARS)

6.1 VARIABLE \_\_\_\_\_ FIXED \_\_\_\_\_ TOTAL \_\_\_\_\_

6.2 COST PER ITEM SCREENED: VARIABLE \_\_\_\_\_ FIXED \_\_\_\_\_ TOTAL \_\_\_\_\_

6.3 COST PER DEFECTIVE FOUND: VARIABLE \_\_\_\_\_ FIXED \_\_\_\_\_ TOTAL \_\_\_\_\_

7. SEND ME A COPY OF THE FINAL REPORT (ADDRESS) \_\_\_\_\_

8. REMARKS

FIGURE 2.2 (cont'd)

TABLE 2.1 Survey Data Distribution

Total Response 17% (34/200)			
Test Level		Test Data	Cost Data
Part	24%	100%	60%
Card/Module	12%	80%	20%
Equip/System	64%	93%	78%

The responses of the questionnaire were of major import in the developing and modeling of test strengths as defined in Section 3.4, as well as supplying "state-of-art" information concerning screening and debugging techniques.

#### 2.3.2 Companies Contacted

Distribution of the questionnaire was made by obtaining a list of responsible contacts who could supply the study with reliable data. It was decided that the GIDEP (Government-Industry Data Exchange Program) Roster of Representatives, in addition to several personal contacts of study personnel, could provide such required data.

The companies chosen were felt to be representative of the "industry". Several divisions of the larger corporations were contacted in hopes of comparing the data for standardization of costs, etc., however, results in this regard were insufficient.

Table 2.2 is the listing of companies chosen for distribution of the questionnaire.

#### 2.3.3 Survey Data

Out of the forty-two (42) data points supplied by the industry survey, thirty-four (34) were found to be sufficient to determine test strengths for their respective screens.

Due to several referenced screen parameters being identical, a total of seventeen (17) different test strengths were determined. The following Table 2.3 presents the data obtained from the industry survey and their respective test strengths. These test strengths were computed from the equations given in Section 4.2.

TABLE 2.2 Companies Contacted

Aerojet Electrosystems Co.	General Atomic Co.
Aeronutronic-Ford Corp.	General Dynamics Corp., Electro
Aerospace Corp.	Dynamics Div.
Ail Cutler-Hammer	General Dynamics Corp., Electronics Div.
Avco Corp., Systems Div.	General Electric Co., Research &
	Development Center
Beckman Instruments	Goodyear Aerospace Corp.
Bell Aerospace Co.	Grumman Aerospace Corp.
Bendix Corp., Guidance Systems Div.	GTE Sylvania Inc. Elect Systems Group,
Bendix Corp., Aerospace Sys Div.,	Western Div.
Mishawaka Operation	GTE Sylvania Inc. Elect Systems Group,
Bendix Corp., Aerospace Sys Div.	Eastern Div.
Booz-Allen Applied Research Inc.	
Boeing Aerospace Company	Harris Corp.
Bulova Watch Co.	Harris RF Communications Inc.
Bunker-Ramo Corp.	Hartman Systems Div. A-T-O Inc.
Celestec Industries Inc.	Hazeltine Corp., Electro-Acoustic Lab
Chrysler Corp.	Hazeltine Corp., Electronics Div.
Cincinnati Electronics Corp.	Hercules Inc.
Control Data Corp.	Hermes Electronics LTD.
Crane Co.	Hewlett Packard Co.
Cubic Corp.	Hittman Assoc. Inc.
Curtiss Wright Corp.	Hoffman Electronics Corp.
Dalmo-Victor Co.	Homes & Narver Inc.
Delco Electronics	Honeywell Inc., Marine Systems Center
Delco Electronics Div., General	Honeywell Inc.
Motors Corp.	Honeywell Inc. Aerospace Div.
Digital Equipment Corp.	Honeywell Information Sys Inc.
Douglas Aircraft Co.	Honeywell Inc., Govt & Aeronaut Prod Div.
	Honeywell Radiation Center
Electro Optical Systems Inc.	IBM Corp.
Electronic Communications Inc.	IBM Corp., Electronics Systems Center
Electrospace Systems Inc.	Interstate Electronics Corp.
Emerson Electric Co.	Itek Corp.
EMR-Telemetry, Weston Instr.	ITT Aerospace/Optical Div.
EPSCO Inc.	ITT Avionics
E-Systems Inc	ITT Federal Electric Corp
	ITT Gilfillan Inc.
Fairchild Hiller Corp.	Jet Propulsion Laboratory
Fairchild Republic Co.	
Fairchild Space & Defense Systems Div.	Kaiser Aerospace & Elec Corp.
Fairchild Stratos Div.	Klauder, Louis T & Assoc.
FMC Corp.	Kollsman Instrument Corp.
Foxboro Co.	

TABLE 2.2 Companies Contacted (Cont)

Lawrence Livermore Lab.	Rockwell International Collins Radio Group
Leeds & Northrup Co.	Electronics Oper DI
Life Systems Inc.	Rockwell International Collins Radio Group
Little. Arthur D. Inc.	Electronics Oper DI
Litton Data System Div.	Rockwell International Columbus Aircraft Civ.
Litton Systems Inc.	Rockwell International Collins Radio Group
Lockheed California Co.	Rohr Industries
Lockheed Electronics Co.	
Lockheed Georgia Co.	Sanders Associates
Lockheed Missiles & Space Co., Space Systems Div.	Sandia Laboratories
Lockheed Missiles & Space Co., Missile Systems Div.	Sangamo Electric Co.
Loral Electronics Corp.	Science Applications Inc.
Martin Marietta Corp.	Science Applications Inc.
Martin Marietta Aerospace	Scott Electronics Corp.
Martin Marietta Aerospace	Sedco Systems Inc.
McDonnell Douglas Astronautics Co, West	Sierra Research Corp.
McDonnell Douglas Electronics Co.	Singer Co.
Mechanics Research Inc.	Singer-Kearfott Div.
Motorola Inc.	Solar Div.
National Water Lift Co.	Sperry Marine Systems Div.
Nature-Crafts	Sperry Microwave Electronics
Northrop Corp., Electro/Mech Div.	Sperry Rand Corp., Sperry Flight Sys. Div.
Northrop Corp., Electronics Div.	Sperry Rand Corp. Systems Mgt. Div.
Northrop Corp., Aircraft Div.	Sperry Rand Corp., Sperry Systems Mgt.
Pacific Car & Foundry Co.	Stencel Aero Engineering Corp.
Parker Hannifin Corp.	Stone & Webster Engineering Corp.
Parsons Co.	Stromberg Carlson Corp.
Perkin-Elmer Corp.	Stromberg Datagraphix Inc.
Plessey Industries Inc.	Systematics General Corp.
RCA Astro Electronics Div.	Systems Associates Inc.
RCA, Aerospace Systems Div.	Systems Evaluation Inc.
RCA, Govt & Comm. Sys.	
RCA, Missile & Surface Radar Div.	Tektronix Inc.
Raytheon Co., Electro Magnetic Sys. Div.	Teledyne Cae
Raytheon Co., Missile Systems Div.	Teledyne Ryan Aeronautical
Raytheon Co., Lever Bldg.	Teledyne Systems Co.
Raytheon Co. Equipment Div.	Texas Instruments Inc.
Reflectone Inc.	Tracor Inc.
Rel-Reeves Inc.	TRW Equipment Group
Rexnord Inc.	TRW Systems Group
Rockwell International Electronics Operations	TRW Colorado Electronics Inc.
Rockwell International Rocketdyne Div.	
Rockwell International Space Div.	Unidynamics
	Union Carbide Corp.
	Union Switch & Signal Div.
	United Engineers & Constructors Inc.
	United Nuclear Industries Inc.
	United Technologies Corp., Norden Div.
	United Technologies Corp., Hamilton Standard Div.
	Univac, Div of Sperry Rand

TABLE 2.2 Companies Contacted (Cont)

Value Engineering  
Vega Precision Laboratories Inc.  
Vitok Engineers Inc.  
Vought Corp., Michigan Div.  
Vought Corp., Systems Div.  
  
Westinghouse Electric Corp., Marine Div.  
Westinghouse Electric Corp., Def & Elect  
    Sys. Center  
Westinghouse Electric Corp., Industrial  
    Equipment Div.  
Westinghouse Electric Corp., Astronuclear Lab  
Weston Instruments Inc.  
Weston Instruments Inc.  
Wiggins EB  
  
Zimmer - USA

TABLE 2.3 Industry Survey Data  
 Time is Expressed in Hours  
 Temp is Expressed in °C

Respondent Number*	Constant Temp C; c**, Temp, Time	Cycled Temp Cyc, Temp Rate, Hi, Lo	Vibration Cyc, g's, Time Temp	Constant Power Cyc. Time	Cycled Power Cyc. Time	Test Strength
2; 30	1, 125, 168			1, 168		.3231
3; 10	1, 25, 168			1, 168		.2177
6	2, 55, 2			2, 8		.0278
8A; B	1, 25, 48			1, 48		.0713
29	1, 40, 144		12, 1, 1.3, 40	12, 12		.2163
38A	5, 125, .16"	5, 12.5, 125, -40		1, 96		.3224
38B	1, 25, 96					.1348
4	14, 72, 5.5	14, 5, 72, -54				.3512
5	14, -54, 2.6	60, 5, 100, -40				.5746
7	1, 60, 2	10, 20, 60, -40				.4869
	10, 60, .25					
	10, -40, .25					
8C; 12-25	8, 55, 2	8, 5, 55, -54	8, 2.2, .33, 55	8, 2.73		.2080
28	18, 55, 2	18, 5, 55, -54	18, 2, .33, 55	18, 3		.3890
31	55, 80, 2	55, 3.8, 80, -55				.5742
	55, -55, .5					
34	12, 55, 5.5	12, 5, 55, -54				.2751
	12, -54, 2.5					
36	4, 60, 8	4, 3, 60, 0				.0666
	4, 0, 1					
37	24, 71, 24	24, 5, 71, -55				.6169
	24, -55, 24					
39	3, 55, 2	3, 5, 55, -55	3, 2.2, .33, 55	3, 2.37		.0863

\*These numbers are the code used for the questionnaire responses.

\*\*Cycle at this level indicates the number of times the Constant Temp is executed.

2.4      Internal Data

2.4.1    Hughes - Fullerton

Much of the cost data used in the study was extracted from Hughes-Fullerton facilities. Due to the many and varied factors contributing to this cost data, an attempt was made to ascertain the needed information on a "typical" card, module, unit, and system level. This was accomplished through an averaging of costs associated with particular programs.

System data was collected from fourteen (14) on-going Hughes-Fullerton programs. Systems Effectiveness Department personnel supplied the study with data in the following fields; MTBF (specified, predicted, initial, final), operating time, environmental time/conditions, percent of non-standard to total number of parts, total number of subcontractors (on assembly level), complexity (part count) and Operations and Maintenance Reports (OMR's). Additional system level data was extracted from The Reliability Growth Study, Ref. 47, conducted at Hughes-Fullerton.

The in-house systems contributing the above mentioned data are:

CVTSC - Display Console  
IPD - Shipboard Radar  
LFR - Low Frequency Receiver  
PRC 104 - Portable Radio  
SID - Display Console  
SLQ 31 - Electronic Warfare Sui.  
TDMA - Communications Terminal  
TPQ 36 - Mortar Locator Radar  
TPQ 37 - Artillery Locator Radar  
UYQ - Console  
COMBAT GRANDE - Air Defense System  
MK 31 - Weapon Control Console  
MK 82 - Weapon Data Converter  
SURTASS - Towed Sonar Segment

2.4.2    Additional Hughes Aircraft Sites

The Hughes sites that contributed to the study's main data base in addition to Hughes-Fullerton were Culver City, El Segundo, and Tucson. Supplementary cost data was obtained from El Segundo manufacturing personnel, screening strength equations were contributed by Culver City, while test effectiveness data was contributed by all three of these additional sites.

The programs/experiments contributed by these Hughes Aircraft Company sites consist of: F-15 Radar Control; Maverick; B-52; and F-15.

The following table 2.4 represents the test data obtained from these sources, combined with the test strengths computed from the test strength equations, Section 4.2.

TABLE 2.4 Hughes Internal Data  
Time is Expressed in Hours  
Temp is Expressed in °C

Reference	Constant Temp Cyc, Temp, Time	Cycled Temp Cyc, Temp Rate, Hi, Lo	Vibration Cyc, g's, Time, Temp	Constant Power Cyc, Time	Cycled Power Cyc, Time	Test Strength
F-15 Radar Control 4 different tests - all at card level	24, 6, 115, -60 72, 6, 115, -60 120, 6, 115, -60 168, 6, 115, -60					.5084 .5979 .6000 .6000
Maverick 16 tests - First twelve at card level, remaining four at unit level	12, 5, 75, -40 12, 10, 75, -40 12, 20, 75, -40 12, 5, 75, -40 12, 10, 75, -40 12, 20, 75, -40 48, 5, 75, -40 48, 10, 75, -40 48, 20, 75, -40 48, 5, 75, -40 48, 10, 75, -40 48, 20, 75, -40 3, 10, 75, -40 12, 10, 75, -40 3, 20, 75, -40 12, 20, 75, -40				.2514 .3975 .5316 .2515 .3976 .5317 .5317 .5922 .5999 .5318 .5923 .6000 .1438 .4006 .5317	
E-52 53 tests - all at card level	64, 5, 80, -40 64, 5, 80, -40 64, 5, 80, -40 64, 5, 80, -40 64, 10, 80, -40 64, 10, 80, -40 64, 10, 80, -40 64, 10, 80, -40					.5904 .5786 .5906 .5697 .6124 .6068 .6124 .5984
				33, 6, .25, 25 33, 6, .25, 25 3, .383 12, .383	64, .40 64, .40 64, .20 64, .20	

TABLE 2.4 Hughes Internal Data (Cont)

Reference	Constant Temp Cyc, Temp, Time	Cycled Temp Cyc, Temp Rate, Hi, Lo	Vibration Cyc, g's, Time Temp	Constant Power Cyc, Time	Cycled Power Cyc, Time	Test Strength
<b>B-52 (Cont)</b>						
53 tests - all at card level i						
	64, 20, 80, -40		33, 6, .25, 25	64, .10		.6111
	64, 20, 80, -40		33, 6, .25, 25	64, .10		.6082
	64, 20, 80, -40		33, 6, .25, 25	64, .10		.6111
	64, 20, 80, -40		33, 6, .25, 25	64, .10		.6000
	168, 5, 80, -40		33, 6, .25, 25	168, .40		.6356
	168, 5, 80, -40		33, 6, .25, 25	168, .40		.6080
	168, 5, 80, -40		33, 6, .25, 25	168, .40		.6366
	168, 5, 80, -40		33, 6, .25, 25	168, .40		.5998
	168, 10, 80, -40		33, 6, .25, 25	168, .20		.3226
	168, 10, 80, -40		33, 6, .25, 25	168, .20		.6083
	168, 10, 80, -40		33, 6, .25, 25	168, .20		.6228
	168, 10, 80, -40		33, 6, .25, 25	168, .20		.6000
	168, 20, 80, -40		33, 6, .25, 25	168, .10		.6155
	168, 20, 80, -40		33, 6, .25, 25	168, .10		.6083
	168, 20, 80, -40		33, 6, .25, 25	168, .10		.6156
	168, 20, 80, -40		33, 6, .25, 25	168, .10		.6000
	64, 15, 80, -40		64, .133	64, .233		.6037
	64, 80, .05					
	64, -40, .05					
	1, 25, 100					
	1, 25, 17.07					
	64, 5, 80, -40		33, 6, .25, 25	64, .40		.5904
	64, 5, 80, -40		33, 6, .25, 25	64, .40		.5786
	64, 5, 80, -40		33, 6, .25, 25	64, .40		.5818
	64, 5, 80, -40		33, 6, .25, 25	64, .20		.5697
	64, 10, 80, -40		33, 6, .25, 25	64, .20		.6124
	64, 10, 80, -40		33, 6, .25, 25	64, .20		.6068
	64, 10, 80, -40		33, 6, .25, 25	64, .20		.5042
	64, 10, 80, -40		33, 6, .25, 25	64, .10		.5985
	64, 20, 80, -40		33, 6, .25, 25	64, .10		.6111
	64, 20, 80, -40		33, 6, .25, 25	64, .10		.6082
	64, 20, 80, -40		33, 6, .25, 25	64, .10		.6029
	64, 20, 80, -40					.6000

TABLE 2.4 Hughes Internal Data (Cont)

Reference	Constant Temp Cyc, Temp, Time	Cycled Temp Cyc, Temp Rate, Hi, Lo	Vibration Cyc, g's, Time Temp	Constant Power Cyc, Time	Cycled Power Cyc, Time	Test Strength
F-15 3 tests - card level unit level equip. level	168, 5, 115, -55 33, 5, 70, -40 30, 5, 71, -55			33, .367 30, .367	33, .367 30, .367	.6000 .4664 .4662

SECTION 3.0 – DESCRIPTION OF IMPORTANT  
MODEL VARIABLES

3.1

Role of Test Strength in the Screening and Debugging Optimization (SDO) Model

The idea in a sequence of screens is the successive removal of defects. For screens that do not damage good items the probability of "catching" a defect is generally not one. Thus if  $D_I$  defects are submitted to the first of  $k \geq 1$  screens the number of defects caught by the first screen will be (say)  $\bar{D}_1 \leq D_I$ . The number of defects entering the second screen (neglecting rework cycles to simplify the illustration) is the  $D_I - \bar{D}_1$  and if the number caught by the second screen is  $\bar{D}_2 \leq D_I - \bar{D}_1$  the number of defects entering the third screen is  $D_I - \bar{D}_1 - \bar{D}_2$  and so on.

Test strength, the probability that a given defect will be caught by the  $k^{\text{th}}$  screen, is called  $TS_k$  and  $TS_k$  assists in providing a mathematical model for the screening process. The actual number of defects removed solely by the  $k^{\text{th}}$  screen is a random variable whose domain is the set of positive integers

$$0, \dots, D_I - \sum_{i=1}^{k-1} \bar{D}_i$$

where of course  $\bar{D}_i$ , the number removed at the  $i^{\text{th}}$  screen,  $1 \leq i \leq k$  is also a random variable. To carry the analysis throughout the process on the basis of random variables is quite complicated so we deal with expected values. Thus, if  $D_I$  defects enter screen one, two quantities are of interest

i) the expected number of defects entering the second screen:

$D_I - TS_1 D_I = (1 - TS_1) D_I$ . That is, this quantity is the expected number of defects not caught.

ii) the expected number of defects removed by the first screen:

$$(TS_1) D_I.$$

The expected number of defects removed by the second screen is  $(TS_2) \times (1 - TS_1) D_I$  and hence the expected number of defects entering the third screen is  $(1 - TS_1)(1 - TS_2) D_I$ . The expected number of defects removed by the third screen is  $TS_3 (1 - TS_1)(1 - TS_2) D_I$  so that the expected number of defects entering the fourth screen is  $(1 - TS_1)(1 - TS_2)(1 - TS_3) D_I$ . In general then the expected number of defects entering the  $k^{\text{th}}$  successive screen is

$$\prod_{i=1}^{k-1} (1 - TS_i) D_I$$

and if the sequence is stopped at, but including, the  $k^{\text{th}}$  screen the final defects remaining

$$D_F = \prod_{i=1}^k (1 - TS_i) D_I.$$

In the next section we obtain interval estimates of the random variable  $D_I$ .

Interval Estimates for the Number of Defects Remaining,  $D_F$  - Monitoring/  
Controlling the Screening Process.

As noted in the previous section the entire process of starting with  $D_I$  initial defects and removing defects by a sequence of screens until there are only  $D_F$  defects left after the last screen is a random process. The SDO model works, necessarily, with expected values; to do otherwise would result in enormous costs.

However, the fact remains  $D_I$ ,  $D_F$  are random unknown quantities and the model works with the expected values of  $D_I$  and  $D_F$ . Now suppose there are  $k \geq 1$  screens in the entire process with  $TS_i$ ,  $i=1, \dots, k$  the test strength of the  $i^{\text{th}}$  screen. The only observable (random) quantities are the number of defects caught at each screen,  $\bar{D}_i$ , where  $\bar{D}_i$  represents the number caught at the  $i^{\text{th}}$  screen. It would be of value to have confidence interval estimates of  $D_I$  and  $D_F$ ; neither of which is observable.  $D_F$  is particularly important because it is the number of defects remaining in the "system" after the last ( $k^{\text{th}}$ ) screen is completed. We note again that

$$D_F = D_I - \sum_{i=1}^k \bar{D}_i$$

That is, the number of defects remaining after the last ( $k^{\text{th}}$ ) screen is the initial number present minus the total of those removed, caught or detected.

It is fairly easy to show that the probability distribution of  $\bar{D}_k$ , the number of defects caught by the last screen is binomial with parameters  $D_I$  and

$$\prod_{i=1}^k (TS_i)$$

where

$$\bar{TS}_i = 1 - TS_i, \text{ i.e.,}$$

$$P(\bar{D}_k = X) = \binom{D_I}{X} \left[ \prod_{i=1}^k \bar{TS}_i \right]^X \left[ 1 - \prod_{i=1}^k \bar{TS}_i \right]^{D_I-X}$$

In this expression there are three quantities of importance:  $D_I$  (not observable),

$$\prod_{i=1}^k \bar{TS}_i$$

(computable from the test strength equations) and  $X$  the observable number of defects actually caught at the last ( $k^{\text{th}}$ ) screen. Hence the known quantity

$$\prod_{i=1}^k \bar{TS}_i$$

and the observable quantity  $X$  can be used for inferences on  $D_I$ .

Once having obtained inferences on  $D_I$  we can obtain inferences on

$$D_F = D_I - \sum_{i=1}^k \bar{D}_i$$

since

$$\sum_{i=1}^k \bar{D}_i$$

is observable (it is the total defects detected). Of course only upper confidence bounds on  $D_F$  are of interest. If not all the parts are tested, the value of  $D_F$  must be adjusted according to what fraction is tested.

#### Example

Suppose  $k = 3$  screens with test strengths  $TS_1 = 0.50$ ,  $TS_2 = 0.50$ ,  $TS_3 = 0.20$ . Suppose also that at screen one 6 defects are caught, at screen two 2 defects are caught and at screen three 1 defect is caught. Then  $\bar{D}_1 = 6$ ,  $\bar{D}_2 = 2$ ,  $\bar{D}_3 = 1$ . Thus

$$\sum_{i=1}^3 \bar{D}_i = 9 \text{ and } \prod_{i=1}^3 \bar{TS}_i = (0.50)(0.50)(0.80) = 0.20, \bar{TS}_i = (1 - TS_i).$$

Hence,

$$P(\bar{D}_3 = X) = \binom{D_I}{X} (0.20)^X (0.80)^{D_I-X}.$$

But we have observed  $X = 1$ . Proceeding to tables of the binomial probability distribution with "p" = 0.20 we find

$$P(X \leq 1 \mid D_I = "n" = 20) = 0.07$$

Hence the confidence is 0.93 that  $9 \leq D_I \leq 19$ , and hence the confidence is 0.93 that

$$D_I - \sum_{i=1}^3 \bar{D}_i = D_F$$

is less than or equal to  $19 - 9 = 10$ .

These interval estimates can be prepared AFTER EACH SUCCESSIVE SCREEN to monitor/control the evolution of the entire process. That is, the interval estimate of  $D_F$  at the  $m^{\text{th}}$  stage can be compared with the expected results (at the  $m^{\text{th}}$  stage):

$$D_F^m = E(D_I) \left[ \prod_{i=1}^m \bar{TS}_i \right]$$

where  $(D_I)$  is expected value of  $D_I$  used in the SDO model. Discrepancies between observed and expected results must be resolved between two items: was the initial  $(D_I)$  incorrect or is one (or more) of the  $TS_i$ 's incorrect. In any case the observed  $D_F^m$ 's can be projected ahead to see what  $D_F$  might finally be by using the  $TS_{M+1}$ ,  $TS_{M+2}$ , ...,  $TS_k$  to predict the final  $D_F^k$  from the observed results up to the  $m^{th}$  test and the expected results from the  $(M+1)^{st}$  stage to the  $k^{th}$  stage.

#### Example

We continue with the previous example and suppose there is one more last test to be run with  $TS_4 = 0.15$  and after all four tests we had hoped to have  $D_F \leq 8$ . We already have any upper confidence limit of 10 on  $D_F^3$  and  $D_F = D_F^4 = TS_4 (D_F^3) = (0.85) 10 = 8.5$  which is an upper 0.93 confidence limit on  $D_F$  and the achievement of  $D_F \leq 8$  appears to be entirely feasible.

### 3.3

#### Role of Part Level Screens

The SDO model developed as a result of this study does not include, explicitly, a part level screen. This was done for two reasons. As will be described in Section 5.0 the initial number of defects,  $D_I$ , is an entry variable in the model.  $D_I$  is a function of the number of parts, say  $N$ , in the system (a quantity known to a careful reliability engineer), and the fraction defective, say  $p$ , of the incoming parts. Thus,  $D_I = N_p S$  (where  $S$  is the number of systems) and  $p$  is a function of quality level (e.g., commercial, B-level) of part purchased. In order to keep the CPU time and core requirements of the SDO model within bounds it was felt that the user can easily price the various quality grades and compute  $D_I$  as a function of quality ( $p$ ) and price then for each  $D_I$  run the SDO model to see, as  $D_I$  varies, which is the most economical part quality level to purchase. For example, suppose the final tolerable defects  $D_F = 20$  and that "to go" from  $D_I = 500$  to  $D_F = 20$  costs \$100,000 worth of screens and that to go from  $D_I = 250$  to  $D_F = 20$  costs \$63,000 worth of screens. Now suppose  $N = 25,000$  and  $p = 0.01$  for B-level and  $p = 0.02$  for commercial parts. If the incremental cost of the B-level parts is less than \$37,000 it is cheaper to start with B-level parts.

The second reason part level screens have been omitted from the model is the "exploding" effect of part level defects when the parts are assembled on cards. What this means is that part screening, while probably relatively inexpensive, must have a tremendously high (near one) TS to prevent defective cards. For example, suppose a large (the parts are used on many programs not just "yours") part population which has  $p = 0.02$  and that the cards will have 150 parts/card. Then the expected number of defects per card is  $0.02 \times 150 = 3$  and since it takes only one or more bad parts to make a bad card it is clear, using the Poisson approximation, the fraction of bad cards (a card with one or more defects) is on the order of 0.95. Thus to keep the fraction bad cards low the parts must be screened with high TS. For example, suppose TS (at the part level) is so high that  $p = 0.02$  can be reduced to 0.001, a 20-1 reduction. Then the expected number of defects/card is  $0.001 \times 150 = 0.15$  and the fraction of bad cards is on the order of 0.14 - a reduction by a factor of only 7-1.

A more accurate analysis requires some notation and a small amount of probability calculus. Let the following definitions be made:

- $N$  = number of parts in lot
- $M$  = number of cards to be made
- $m$  = number of parts per card
- $p$  = fraction defective of the lot
- $X_i$  = number of defective parts on the  $i^{\text{th}}$  card  $i = 1, \dots, M$

Then  $Np$  is the total number of defective parts in the lot, the  $X_i$  are random variables and .

$$P(X_1 = x_1, \dots, X_M = x_M) = \left[ \frac{\binom{Np}{x_1} \binom{N-Np}{m-x_1}}{\binom{N}{m}} \right] \left[ \frac{\binom{Np-x_1}{x_2} \binom{N-m-(Np-x_1)}{m-x_2}}{\binom{N-m}{m}} \right] \times \dots \\ \dots \times \left[ \frac{\binom{Np - \sum_{i=1}^{M-1} x_i}{x_M} \binom{N - (M-1)m - (Np - \sum_{i=1}^{M-1} x_i)}{m-x_M}}{\binom{N - (M-1)m}{m}} \right]$$

where

$$\binom{x}{y} = \frac{x!}{y!(x-y)!} \quad x \geq y$$

The card fraction defective is then (since it is clear that it is immaterial which  $1 \leq i \leq M$  of the  $M$  cards is defective and zero defective cards contributes nothing to the sum, i.e.,  $0/M = 0$ ).

$$\sum_{i=1}^M \binom{i}{M} \binom{M}{i} P(X_1 = 0, \dots, X_{M-i} = 0, X_{M-i+1} \geq 1, \dots, X_M \geq 1)$$

where

$$\begin{aligned} P(X_1 = 0, \dots, X_{M-i} = 0, X_{M-i+1} \geq 1, \dots, X_M \geq 1) \\ = \sum_{\substack{X_{M-i+1}, \dots, X_M \geq 1 \\ X_1, \dots, X_{M-i} = 0}} P(X_1 = x_1, \dots, X_M = x_M) \\ \sum x_i \leq mN \end{aligned}$$

The following examples give a very few results since the computations involved can quickly out-distance even a large high-speed digital computer.

#### Example 1

$N = 1000, p = 0.01, m = 50, M = 10$ : the card fraction defective is about 0.40.

#### Example 2

$N = 5000, p = 0.01, m = 100, M = 25$ : the card fraction defective is about 0.63.

### 3.4 Determination of Test Strength (TS)

The ability of particular screen to detect incipient/latent defects will be called test strength (TS) and is represented formally as a probability

TS = the probability that a given screen, including the test set-up, will detect an incipient/latent defect.

The portion of TS relating solely to the ability of the screen to degrade the defect to a detectable level is called screening strength, SS, and the portion relating to the ability of the test equipment/set-up to detect the defect once it has been degraded to a detectable level is called  $P_d$ .

Thus,  $TS = SS \times P_d$ .

Ordinarily, TS is not computable or estimable. For any given screen (with test set-up) TS could be estimated by

$$TS = \frac{\text{number of defects detected after screen}}{\text{number of defects entering screen}}$$

Unfortunately, the denominator of this fraction is usually unknown. It would be possible to estimate TS by placing a known number of bad items on the screen and observing the results but we could uncover no such experiments. Many experiments involve comparing screens for their relative TSs. Thus, if 100 cards are put on each of five different screens (say  $S_1, S_2, \dots, S_5$ ) then assuming the number of defects entering the five screens are equal, the screens can be ranked by the number of defects discovered by the screens and the screens with the largest fallout will be best. However, these sorts of results do not permit actual computation of  $TS_i, i = 1, \dots, 5$  but only permit computation of the ratios

$$\frac{TS_j}{TS_k} \cdot j = 1, \dots, 5 \quad k = 1, \dots, 5, j \neq k.$$

Two approaches to computing TS are available and both have been used in this study. First, from the results of the questionnaire, several respondents felt they had enough experience with incoming defect rates so that we could compute TS for these responses. This occurred about fifteen times. The second approach is to build mathematical functions, derived from the physics of the screening environment, which yield TS (or SS). This has been done by the Hughes Aircraft Company in their Cost Reduction by Early Decision Information Techniques (Ref 42) program (Report No. TIC 20-42-732-R (P73-218)). In that report, equations are given for SS for vibration, constant temperature and temperature cycling screens. These equations were used to compute the SS used in the cost models. Before giving these equations we note that such equations are not available for power applied screens and this power-applied case is treated first.

Probability of detection,  $P_d$ , presents a different problem: it is highly dependent on the individual test setups. We have not found nor have we developed quantitative models for computing  $P_d$ . However, the factors to be included in  $P_d$  are: possible failure modes, functions performed, functions tested, test equipment available, test equipment quality and calibration, instrumentation setup and data recording.

### 3.4.1 TS for Power Continuously Applied Screens

Seven respondents to the questionnaires had conducted continuously applied (rated) power tests with all other conditions at ambient. We were able also to compute the TS's for these seven tests and of course the seven test times,  $t_i$ , were also available. The usual procedure is to fit (to the data) various linear or linearizable functions by least squares and select the one which is best fit. In the present case, TS must satisfy  $0 \leq TS \leq 1$  so we first selected a function of  $t$  (time of continuously applied rated power) such that  $TS = 0$  at  $t = 0$  and  $TS = 1$  at  $t = \infty$ . Such a function is

$$TS = g(t) = \begin{cases} \frac{bt}{a+bt}, & a, b, t > 0, \\ 0 & \text{elsewhere.} \end{cases}$$

Obviously,

$$TS = g(t) = \frac{t}{\frac{a}{b} + t}$$

so that TS depends only on the ratio  $a/b = c$ .

$$TS = g(t) = \frac{t}{t+c} \quad (1)$$

A description of the results for obtaining the constant  $c$  is contained in Section 4.1. No data was available for the case when the power applied was other than rated.

### 3.4.2 TS for Cycled (ON/OFF) Rated Power

The continuously applied rated power case is just a special case of this present case when the number of cycles, say  $N_c$ , is one. Now suppose  $N_c > 1$ , then TS, the probability of detection is just one minus the probability of non-detection in  $N_c$  cycles. That is,

$$TS = 1 - \left(1 - \frac{t}{t+c}\right)^{N_c} \quad (2)$$

where

$N_c$  = number of cycles

$t$  = length of a cycle in hours

This reduces to equation (1) when  $N_c = 1$ .

At first glance it appears that equation (2) "neglects" the screening effect due solely to the act of "turning" the power off and on. This is not quite true as can be seen by an example. Suppose  $N_c = 18$  and  $t = 4$ . Then from equation (2), using  $c = 886.62$ ,

$$TS = 1 - \left(\frac{886.62}{890.62}\right)^{18} = 0.078.$$

If however we choose to neglect the cycling and just assume  $4 \times 18$  total hours of screening and apply equation (1)

$$TS = \frac{72}{72 + 886.62} = \frac{72}{958.62} = 0.075$$

which is less (test strength) than 0.078. It is trivial to show that equation (2) always gives greater TS than equation (1) with  $t' = N_c t$  where  $t$  is the length of a cycle.

Thus, the effect of the on/off portion is included. Equation (2) may be generalized to the case where each of the  $N_c$  cycles has a differing time  $t_i$ ,  $i = 1, \dots, N_c$ :

$$TS = 1 - \prod_{i=1}^{N_c} \left( \frac{c}{c + t_i} \right)$$

### 3.5 Determination of Initial ( $D_I$ ) and Final ( $D_F$ ) Defects

To use the SDO model for minimizing cost the user must enter, among other parametric values, the initial ( $D_I$ ) and final ( $D_F$ ) number of defects desired.

$D_I$  should be based on the total parts count for the system, the number of systems, and the expected part fraction defective. This latter number will ordinarily be based on the quality level of parts purchased for each of the major part types. For

example suppose a system is composed of mil std parts (B and C level quality grades) as well as non-standard parts, then  $D_I$  is calculated as follows:

<u>Part Type</u>	<u>Quantity</u>	<u>Quality Grade</u>	<u>% Defective</u>
Resistors	1,000	C level	3
ICs	10,000	B level	1
Non-standard (various)	3,000	non-mil std	5

$$D_I = 1,000(0.03) + 10,000(0.01) + 3,000(0.05) = 280$$

The computer program defaults to 1% in the absence of user specification.

$D_F$  is determined by using the mature MTBF  $\theta_M$  (e.g., obtained from a MIL-HDBK-217B prediction),  $\theta_F$  the required MTBF before field delivery, and a relatively long period of field operation, say  $t$ , in the following equation

$$D_F = \left[ \frac{t}{\theta_F} - \frac{t}{\theta_M} \right] (TS)^{-1}$$

The left hand term in the bracket is the expected number of failures in  $t$  (field) hours when the MTBF is  $\theta_F$ , the at-delivery MTBF. The right hand term is the expected number of random failures in  $t$  (field) hours. Thus the difference is the expected number of engineering errors, manufacturing quality and unreliability failures removed during  $t$  (field) hours. When this difference is divided by the test strength for  $t$  (field) hours,  $D_F$  is obtained. That is, the equation

$$D_F \times TS = \left[ \frac{t}{\theta_F} - \frac{t}{\theta_M} \right] \text{ is solved for } D_F.$$

The SDO model has this equation in it with  $t = 26.280$  hrs (3 years operation) and the computation of TS is performed by the program.

#### Example

Suppose  $\theta_F = 300$  hours,  $\theta_M = 420$  hours, and  $t = 10,000$  hours with no temperature or vibration excursion expected during field operations. Then, (see Section 4.1.2).

$$TS = \frac{t}{t+c} = \frac{10,000}{10,000 + 4,066} = 0.71$$

$$t/\theta_F = \frac{10,000}{300} = 33.3; t/\theta_M = \frac{10,000}{420} = 23.8$$

and

$$D_F = \frac{[33.3 - 23.8]}{0.71} = 13.4 \approx 13.$$

## SECTION 4.0 -- DATA ANALYSIS FOR MODEL PARAMETERS

### 4.1 Computation of Test Strength for Power Applied

#### 4.1.1 Test Strength (TS) for Card/Unit Level:

The BMD07R, non-linear least squares computer program (described in the Appendix) was used to find the parameter,  $c$ , in the functional equation:  $TS = t/(t + c)$ . Using the seven data points shown in Table 4.1 (which were obtained from the questionnaire), an initial estimate of  $c = 1000.0$  allowed the iteration scheme to converge in a few steps to  $c = 886.62$ .

These survey points were also used to find  $a$  and  $c$  in  $TS = (t/(t + c))^a$  and  $a$  and  $b$  in  $TS = 1.0 - \exp[-(t/a)^b]$ . However, using these functional equations did not improve upon the performance of  $TS = t/(t+c)$  with respect to low estimating error.

The QK PLOT graphics subroutine (described in the appendix) was then used to plot Test Strength (TS) versus time (t) using the equation  $TS = 1.0 - [886.62/(t + 886.62)]^{N_c}$ . Three curves were produced where  $N_c = 1$ ,  $N_c = 5$ , and  $N_c = 10$  (see Figure 4.1) as a means of illustrating the improvement in TS as  $N_c$ , the number of cycles gets large.

Another illustration worth looking at is, for fixed total time T, the improvement in cycled power versus continuous power applied.

Suppose the total time of power applied  $T = 200$  hours. We consider

Case 1      200 hours of continuously applied power

$$TS = \frac{t}{t + c} = \frac{200}{200 + 886.62} = 0.184$$

Case 2      2 cycles of  $t = 100$  hours each

$$TS = 1 - \left[ \frac{886.62}{886.62 + 100} \right]^2 = 0.192$$

Case 3      4 cycles of  $t = 50$  hours each

$$TS = 1 - \left[ \frac{886.62}{886.62 + 50} \right]^4 = 0.198$$

Case 4      8 cycles of  $t = 25$  hours each

$$TS = 1 - \left[ \frac{886.62}{886.62 + 25} \right]^8 = 0.200$$

TABLE 4.1. Questionnaire Data Points

	Time (t)	Test Strength (TS)
1	168	.09257
2	168	.02900
3	100	.05696
4	96	.05479
5	96	.50000
6	48	.02805
7	17.07	.01013

Case 5     16 cycles of  $t = 12.5$  hours each

$$TS = 1 - \left[ \frac{886.62}{886.62 + 12.5} \right]^{16} = 0.201$$

While obviously more costly, the large cycles have slightly better TS. Clearly the upper limit of TS obtainable by infinitesimal lengths  $t$  of an infinite number of cycles is, for power applied for total time  $T$ ,

$$1 - e^{-T/886.62}$$

Thus, since the worst case is  $N_c = 1$  which has

$$TS = \frac{T}{T + 886.62},$$

then the difference

$$(1 - e^{-T/886.62}) - \left( \frac{T}{T + 886.62} \right)$$

represents the total possible improvement in TS for fixed total time  $T$ .

In the preceding example this difference is on the order of  $0.201 - 0.184 = 0.017$  which hardly seems worth the trouble of using  $N_c > 1$ .

#### 4.1.2 Test Strength (TS) for Equipment/System Level:

The BMD07R computer program (described in the appendix) was used again to find the parameter,  $c$ , in the functional equation  $TS = t/(t+c)$ . TS (test strength) was computed from the data points using the formula  $TS = 1.0 - (\theta_{\text{Initial}}/\theta_{\text{Final}})$ .

Eighteen, and finally, sixteen values of time ( $t$ ), MTBF initial and final ( $\theta_I$ ,  $\theta_F$ ) from the internal data base (see Table 4.2) were used, along with an initial estimate of  $c = 5000.0$ . This allowed the iteration scheme to converge in eight steps to  $c = 4084.8$  and  $c = 4066.5$ , respectively. The SDO model uses  $c' = 3049.875 = (0.75)(4066.5)$  in order to "factor out" the probability of detection,  $P_d$ . That is, the SDO model uses  $TS = t/(t + c'/P_d)$  because it separates SS and  $P_d$ .

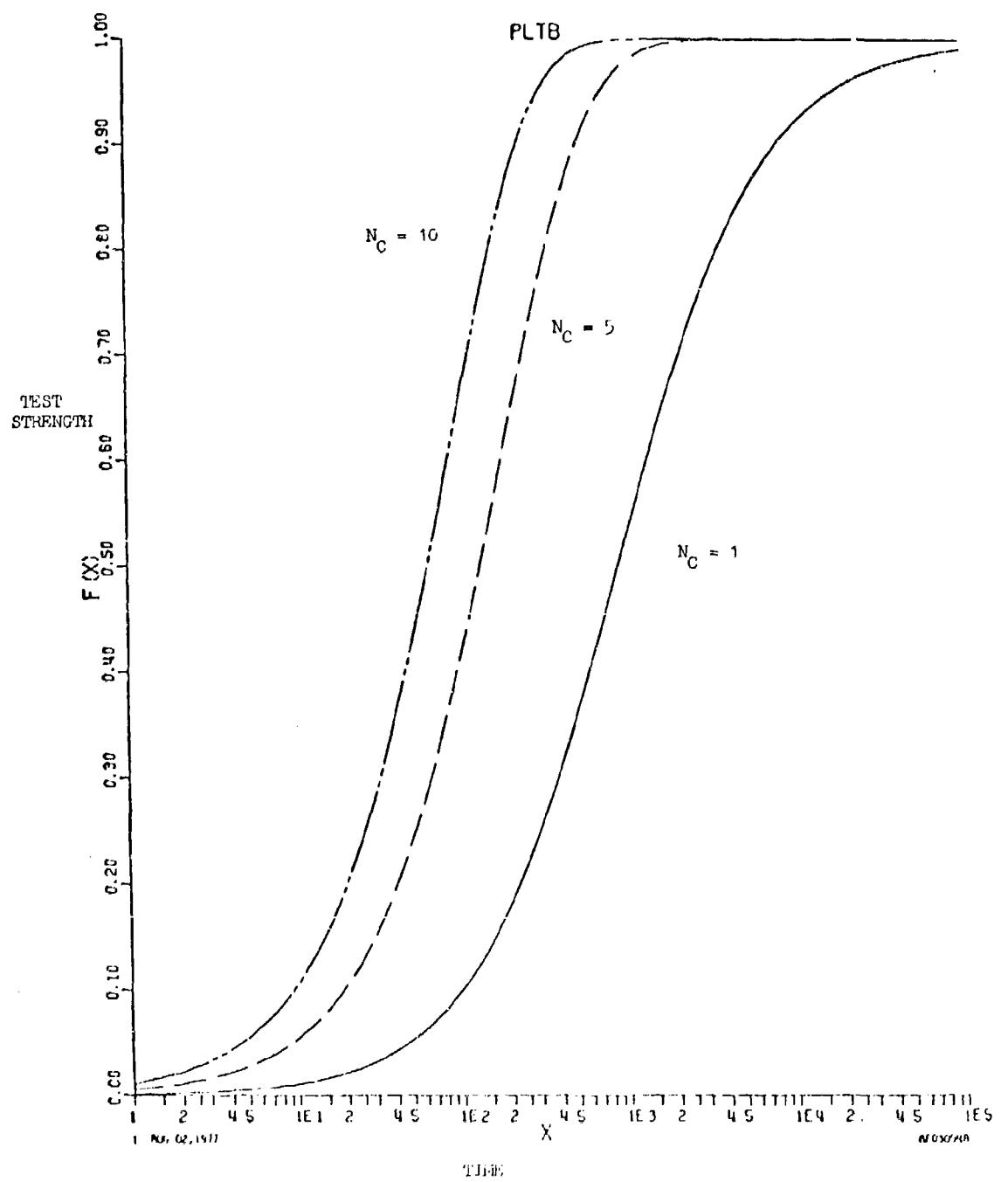


FIGURE 4.1 POWER TEST STRENGTH

TABLE 4.2. Internal Data Points

	Time	$\theta_{\text{Initial}}$	$\theta_{\text{Final}}$	Test Strength
1	4000	150	300	.5000
2	5000	50	100	.5000
3	1689	42	98	.5714
4	1690	92	216	.5741
5	4200	14	148	.9054
6	15000	10	29	.6552
7	4502	.6	17	.9647
8	16491	83	192	.5417
9	5986	134	456	.7061
10	14006	283	581	.5129
11	10339	182	233	.2189
12	4312	17	54	.6852
13	15519	231	251	.0797
14	5537	8	17	.5294
15	12100	6	22	.7273
16	6100	71	126	.4365

The Reliability Growth Study (ref. 47) provided eighty-one triples of time  $t$ ,  $P_1$  (Duane logarithmic growth rate) and  $P_2$  (Duane intercept parameter) for in-house ground (thirty-one points) and in-house airborne (fifty points) systems. Sixty-three of these points were used in a subsequent analysis (see Section 4.3) with the IBM model and are shown in Table 4.3. Using this data, the test strengths were computed from the above formula and  $\theta_{\text{Initial}}$  and  $\theta_{\text{Final}}$  were computed using:

$$\theta_I = P_2 * I^{\frac{1}{P_1}} [(1.0/P_1) + 1.0]$$

and

$$\theta_F = \left[ P_2^{P_1} * t^{(1.0 - P_1)} \right] / P_1$$

Using initial estimates of the parameter  $c$ , the iteration converged to  $c = 260.06$  for the airborne data;  $c = 916.31$  for the ground data, and  $c = 337.10$  for the ground and airborne data combined.

In order to see if test strength could be written as a function of several variables, the Multiple Linear Regression (MLRG) subroutine (described in the appendix) was used. With test strength (TS) as the dependent variable, various combinations of time, parts count, and environmental conditions were tried as independent variables. In all cases  $R^2$ , the coefficient of multiple determination, was very small (see Table 4.4).

The value  $c = 4066.5$  was used because the internal data base provided the best fit.

TABLE 4.3. Reliability Growth Data

	Time	$\theta_{\text{Initial}}$	$\theta_{\text{Final}}$	TS	Y'
1	3038	330.63	1781.5	.81441	.77819x10 <sup>2</sup>
2	3822	452.02	2030.5	.77739	.22781x10 <sup>-9</sup>
3	6369	44.516	362.86	.87732	.28907x10 <sup>1</sup>
4	3822	131.49	254.52	.48339	.64664x10 <sup>-34</sup>
5	1122	30.115	35.344	.14794	.45780x10 <sup>-4</sup>
6	2700	.099636	62.614	.99841	.68262x10 <sup>-1</sup>
7	3200	.41363	67.855	.99390	.37012x10 <sup>-2</sup>
8	3700	3.6481	142.88	.97447	.88842x10 <sup>-5</sup>
9	3700	1.0919	308.20	.99646	.41420x10 <sup>-5</sup>
10	2500	77.812	204.56	.61961	.27362x10 <sup>-1</sup>
11	2193	23.643	228.75	.89664	.58012x10 <sup>-6</sup>
12	2248	43.723	81.398	.46285	.22290x10 <sup>1</sup>
13	3415	.60827	13.321	.95434	.12416x10 <sup>3</sup>
14	4659	430.02	1150.4	.62619	.27249x10 <sup>-7</sup>
15	6144	196.60	337.75	.41793	.15031x10 <sup>2</sup>
16	4467	8.0679	172.94	.95335	.22948x10 <sup>-5</sup>
17	2043	29.593	336.10	.91195	.10984x10 <sup>-5</sup>
18	2792	234.62	360.44	.34908	.85282x10 <sup>-5</sup>
19	1540	14.205	39.810	.64317	.12821x10 <sup>-7</sup>
20	1726	3.5269	726.54	.99515	.83703x10 <sup>-6</sup>
21	2261	3.0704	215.75	.98577	.38257
22	3105	766.40	3595.0	.78681	.12800x10 <sup>1</sup>
23	3415	4.7866	74.381	.93565	.74311x10 <sup>1</sup>
24	4536	3.7041	123.07	.96990	.16717x10 <sup>1</sup>
25	4536	310.54	548.30	.43364	.34030x10 <sup>1</sup>
26	2076	69.100	202.04	.65799	.13173x10 <sup>-30</sup>
27	5085	29.501	128.74	.77085	.33722x10 <sup>2</sup>
28	1122	24.439	36.220	.32525	.30769x10 <sup>-15</sup>
29	1406	22.256	61.352	.63724	.18048x10 <sup>1</sup>
30	400	22.892	75.951	.69859	.58302x10 <sup>1</sup>
31	1200	7.3325	106.40	.93108	.18408x10 <sup>-2</sup>
32	4988	4.2615	44.205	.90360	.48216x10 <sup>3</sup>

TABLE 4.3. Reliability Growth Data (Continued)

	Time	$\theta_{\text{Initial}}$	$\theta_{\text{Final}}$	TS	Y'
33	1000	.92606	47.765	.98061	.13708x10 <sup>-1</sup>
34	2176	.027881	11.465	.99757	.11321
35	400	2.4662	24.715	.90022	.15516x10 <sup>1</sup>
36	1300	1.0681	65.495	.98369	.92072x10 <sup>-2</sup>
37	4996	36.735	95.991	.61730	.21104x10 <sup>1</sup>
38	1200	41.069	167.65	.75504	.12586
39	536	96.734	116.72	.17123	.86898x10 <sup>-1</sup>
40	500	21.399	88.647	.75860	.15739
41	760	8.8171	183.88	.95205	.14099
42	1400	31.520	61.006	.48333	.11316
43	800	29.775	74.712	.60148	.12527
44	760	32.189	213.28	.84908	.12449
45	782	39.047	202.26	.80694	.17877x10 <sup>-6</sup>
46	767	17.226	125.43	.86267	.57691x10 <sup>-1</sup>
47	760	25.414	125.13	.79690	.22788x10 <sup>1</sup>
48	782	33.681	301.72	.88837	.50858
49	760	7.0105	82.785	.91532	.26110
50	782	9.4446	106.30	.91115	.22573x10 <sup>-2</sup>
51	767	5.4092	108.96	.95036	.25400
52	2500	1.5325	73.877	.97926	.48286x10 <sup>-15</sup>
53	798	.25120	14.433	.98260	.16778x10 <sup>-1</sup>
54	1097	.60820	14.040	.95668	.40146
55	399	.33733	11.201	.96988	.13968x10 <sup>2</sup>
56	1192	.12900	5.5837	.97690	.37987x10 <sup>-1</sup>
57	2500	.84153	11.323	.92568	.14334x10 <sup>-11</sup>
58	760	1.12714	62.248	.97958	.32929
59	500	6.5611	41.867	.84329	.28217x10 <sup>1</sup>
60	767	6.4641	85.624	.92451	.56011x10 <sup>-1</sup>
61	782	6.0595	89.320	.93216	.32853x10 <sup>-1</sup>
62	766	5.9670	40.514	.85272	.21182
63	549	6.9464	16.115	.56896	.16692

Note: Numbers 1-28 represent the ground data.

Numbers 29-63 represent the airborne data.

TABLE 4.4. Test Strength as a Function of Several Variables

Independent Variables	$R^2 = \text{Coefficient of Multiple Determination}$
Environmental Conditions, Parts Count, and Time	$1.72927 \times 10^{-1}$
Parts Count and Time	$9.75238 \times 10^{-2}$
Environmental Conditions and Time	$1.58463 \times 10^{-1}$
Time	$6.47496 \times 10^{-2}$

4.2 Test Strength for Constant Temperature, Temperature Cycling, Vibration, and Combined Screens

The equations utilized for these screens were developed using Arrhenius relations and were taken from the Hughes Aircraft Company CREDIT report (ref 42).

Test strength for  $k$  combined screens is defined as  $TS = 1 - \prod_{i=1}^k (1 - TS_i)$ . That is, total test strength is the probability the defect is detected on at least one of the screens which is one minus the probability it is not detected on any of the screens.

Five types of screens are provided in the model: the two power screens previously discussed and temperature cycling, constant temperature and vibration. If a particular screen is not used its' TS defaults to 0.

$$TS_1 \text{ (constant temp)} = \left[ 0.6 \times P_d \left[ \frac{1}{1 - e^{-N \times t_T \times 2.63 \times 10^{-5}} \times e^{0.0122(T_a + 273)}} \right] \right]$$

$$TS_2 \text{ (cycled temp)} = \left[ 0.8 \times P_d \left[ \frac{1}{1 - e^{-N \times \frac{dT_i}{dt} \times 11.835 \times 10^{-5}} \times e^{0.0122(T_{dt} + 273)}} \right] \right]$$

$$TS_3 \text{ (vibration)} = \left[ 0.2 \times P_d \left[ \frac{1}{1 - e^{-N \times g \times t_v \times 7.69 \times 10^{-5}} \times e^{0.0122(T_v + 273)}} \right] \right]$$

where  $N$  = number of cycles

$t_T$  = time of temperature exposure (hours)

$T_a$  = actual temperature ( $^{\circ}\text{C}$ )

$\frac{dT_i}{dt}$  = rate of temperature change ( $^{\circ}\text{C}/\text{min}$ )

$T_{dt} = (|\text{hi temp} - 25| + |\text{lo temp} - 25| + 50)/2$  ( $^{\circ}\text{C}$ )

where  $g$  = vibration (g's) (sinusoidal at nonresonant frequency)

$t_v$  = length of vibration (hours)

$$T_v = |\text{temp at vibration } -25| + 25 \text{ } ^\circ\text{C}$$

The constants used are derived parameter values for cards containing miscellaneous parts, with the exception of 0.75 which is the default of  $P_d$  used in the definition of test strength. No "model" was available, anywhere, for  $P_d$ . The value 0.75 is the best number available based on Hughes internal experience.

#### 4.3 Conversion of MTBF ( $\theta$ ) to Defects

Several methods were tried in order to convert MTBF to defects. The recommended method is described in Section 3.5. In this section we describe other methods which were not successful. The multiple linear regression (MLRG) subroutine (described in the appendix) was used to write the dependent variable  $Y = \text{OMR}/\text{Part Count}$  as a function of the independent variables, MTBF final and MTBF predicted. In each case  $R^2$ , the coefficient of multiple determination, was very small. Eighteen values of the internal data were used, producing the results with  $Y$  as defined above:

Independent Variable	$R^2$
$\theta$ Final, $\theta_F$	$5.98614 \times 10^{-2}$
$\theta$ Predicted, $\theta_P$	$7.18709 \times 10^{-2}$

CURFIT, a least squares curve fit program (described in the appendix), was implemented using  $Y = \text{OMR}/\text{Part Count}$  as the dependent variable versus the independent variables of  $\theta_{\text{Final}}$ , and  $\theta_{\text{Predicted}}$ . Also,  $Y = \text{OMR}$  was used as a dependent variable versus the same independent variables as above. The data points used came from the internal data values. The results in Table 4.5 show that the coefficient of multiple determination was very small for all equation types.

Data values from the "Reliability Growth Study" (ref. 47) IBM model, "In-house" ground and airborne systems were combined with corresponding ones from the Duane model (refer to Table 4.3) to produce sixty-three points that were used in the CURFIT program. The dependent variable  $Y' = K_1 e^{-K_2 t} = P_2 e^{-P_3 t}$  ( $P_2$  and  $P_3$  are the computer codes used in ref. 47) and the independent variable  $\theta_F$  was computed from the Duane model. The IBM model gives: Cumulative number of correctable failures at remaining time  $t = K_1 e^{-K_2 t}$ . Obviously (when  $t = 0$ )  $K_1$  is the initial number of defects present and  $K_2$  is a "removal rate." Thus at program end (i.e., at delivery time)  $t_F$ ,  $Y' = K_1 e^{-K_2 t_F}$  is  $D_F$ .

Again, the coefficient of multiple determination was very small when it could be computed, as Table 4.6 shows that there was no fit possible for four of the six curve types.

The poor results caused us to abandon this approach to converting  $\theta$  to defects. The approach adopted is given in Section 3.5.

TABLE 4.5. Conversion of MTBF to Defects Using Internal Data

Curve Type	Y = OMR versus $\theta_{\text{Final}}$		Y = OMR versus $\theta_{\text{Predicted}}$	
	Index of Determination A	B	Index of Determination A	B
$Y = A + BX$	0.102556	256.142	-0.457676	4.88187x10 <sup>-3</sup>
$Y = Ae^{BX}$	0.190088	160.936	-3.33675x10 <sup>-3</sup>	1.55167x10 <sup>-3</sup>
$Y = AX^B$	0.263598	1126.6	-0.54605	9.01423x10 <sup>-3</sup>
$Y = A + B/X$	0.284264	65.565	6214.9	.108037
$Y = 1/(A + BX)$	0.085478	1.25028x10 <sup>-2</sup>	4.7611x10 <sup>-5</sup>	1.29326x10 <sup>-2</sup>
$Y = X/(A + BX)$	0.140562	-0.437978	2.95385x10 <sup>-2</sup>	3.96961x10 <sup>-2</sup>
Y = OMR/Part Count versus $\theta_{\text{Final}}$				
Curve Type	Y = OMR/Part Count versus $\theta_{\text{Final}}$		Y = OMR/Part Count versus $\theta_{\text{Predicted}}$	
	Index of Determination A	B	Index of Determination A	B
$Y = A + BX$	5.98638x10 <sup>-2</sup>	3.97853x10 <sup>-2</sup>	-9.51293x10 <sup>-5</sup>	7.18726x10 <sup>-2</sup>
$Y = Ae^{BX}$	9.34212x10 <sup>-2</sup>	1.06872x10 <sup>-2</sup>	-2.83562x10 <sup>-3</sup>	2.67243x10 <sup>-2</sup>
$Y = AX^B$	0.167471	7.45869x10 <sup>-2</sup>	-0.527607	5.38234x10 <sup>-2</sup>
$Y = A + B/X$	0.129448	3.00551x10 <sup>-3</sup>	1.14097	8.52037x10 <sup>-2</sup>
$Y = 1/(A + BX)$	0.024747	257.687	.376589	2.28171x10 <sup>-2</sup>
$Y = X/(A + BX)$	0.127914	-6983.29	449.607	3.88004x10 <sup>-4</sup>
Y = OMR versus $\theta_{\text{Predicted}}$				

TABLE 4.6. Conversion of MTBF to Defects Using Reliability Growth Data

Curve Type	$Y' = K_1 e^{-K_2 t}$ versus $\theta_{\text{Final}}$		
	Index of Determination	A	B
$Y = A + BX$	*	*	*
$Y = Ae^{BX}$	$6.70122 \times 10^{-5}$	$6.82998 \times 10^{-4}$	$-2.19542 \times 10^{-4}$
$Y = AX^B$	$1.35703 \times 10^{-2}$	0.321833	-1.32986
$Y = A + B/X$	*	*	*
$Y = 1/(A + BX)$	*	*	*
$Y = X/(A + BX)$	*	*	*

\*No fit

#### 4.4 Cost Data and Cost Equations

Cost data, as was previously mentioned, was obtained from the Hughes Aircraft Company manufacturing departments of the Fullerton and El Segundo sites.

It was decided, due to the instability of the dollar, to use man-hours as the basic unit of the cost data. This data was obtained by engineering estimates of a "typical" card, unit, equipment, and system man-hour usage of the five screening techniques addressed in this report, i.e., constant temperature, cycled temperature, vibration and, constant and cycled power.

Rework cost data was also given by engineering estimates of the cost involved as well as an estimate of number of defects caught by each rework cycle at "typical" card/unit and equipment/system levels.

The basic cost equation is of a linear nature where total cost = fixed cost + (variable cost x duration of test). The default used for fixed cost in the SDO model is zero due to the assumption that test equipment, etc., are already available to the user. The two defaults of the variable cost are derived from the cost data obtained at card/unit and equipment/system levels respectively.

The cost equations derived for the following screens given for 1) card/unit levels, and 2) equip/system levels:\*

##### Constant Temperature:

1. Test Cost = Fixed Cost + B1 x Test Time (man-hours)
2. Test Cost = Fixed Cost + B2 x Test Time (man-hours)

##### Cycled Temperature:

1. Test Cost = Fixed Cost + B1 x Difference in Temp Extremes x Rate of Temp Change x Number of Cycles
2. Test Cost = Fixed Cost + B2 x Difference in Temp Extremes x Rate of Temp Change x Number of Cycles

##### Vibration:

1. Test Cost = Fixed Cost + B1 x Duration of Vibration x Number of Cycles
2. Test Cost = Fixed Cost + B2 x Duration of Vibration x Number of Cycles

##### Constant Power:

1. Test Cost = Fixed Cost + B1 x Duration of Power Applied
2. Test Cost = Fixed Cost + B2 x Duration of Power Applied

\*The constants B1 and B2 represent the average labor hours per hour of test in the default option for monitoring and data collection. The default values are 0.15 and 1.0 respectively and the fixed costs are zero.

Cycled Power:

1. Test Cost = Fixed Cost +  $B_1 \times$  Duration of Power Applied  $\times$  Number of Cycles
2. Test Cost = Fixed Cost +  $B_2 \times$  Duration of Power Applied  $\times$  Number of Cycles

Where time is given in hours and temperature is given in °C.

Rework costs are an integral part of the total cost considerations, rework's purpose being to correct those defects discovered by the screens. It is a cost that must be incurred as an alternative to discarding in that "new" inputs to a system may contain the same defects as a "rework" if not more. The following table represents the man-hours required to rework at card and higher assembly levels the defects discovered at card, unit, equipment, and system level as obtained from the survey and Hughes internal data.

TABLE 4.7. Man-hours Per Defect

Rework Location	Rework Man-Hours per Defect			
	Card	Unit	Equip	System
Card Level	.5	9.46	51.5	63.0
Higher Assy Level	.1	3.67	45.5	57.0

4.5 Determining Relationship of Number of Subcontractors and Percent Non-Standard Parts to Test Strength

To determine any relationship between Test Strength (TS) and number of subcontractors and percent non-standard parts, a requirement of the work statement, the computer program, CURFIT, described in the appendix, was implemented with data points from the internal data.  $TS = 1.0 - (\theta_I/\theta_F)$  was used as the dependent variable versus the independent variables of number of subcontractors and percent non-standard parts. The results shown in Table 4.8 indicate that the index of determination was not very large in any of the curve fitting cases.

The BMD07R computer program (described in the appendix) was used to find the parameter,  $c$ , in the functional equation  $TS = s/(s + c)$ , where  $s$  = the number of subcontractors. Sixteen data points from the internal data were used with an initial estimate of  $c = 10.0$  which allowed the iteration scheme to converge in eight steps to  $c = 3.8$ . This last analysis also showed no correlation between TS and subcontractors.

TABLE 4.8. Determining Relationship of Non-Standard Parts and Number of Subcontractors to Test Strength

Curve Type	TS versus Number of Subcontractors		TS versus Percent Non-Standard Parts	
	Index of Determination A	B	Index of Determination A	B
$Y = A + BX$	0.48996	0.467763	0.00878	$2.34507 \times 10^{-3}$
$Y = Ae^{BX}$	0.214789	0.421927	$1.57241 \times 10^{-2}$	$7.88787 \times 10^{-4}$
$Y = AX^B$	0.342686	0.274205	0.341502	$1.46446 \times 10^{-4}$
$Y = A + B/X$	0.602609	0.854768	-1.24422	$2.44974 \times 10^{-3}$
$Y = 1/(A + BX)$	$6.90378 \times 10^{-2}$	3.01311	$-4.16702 \times 10^{-2}$	$2.85672 \times 10^{-3}$
$Y = X/(A + BX)$	0.390392	12.6619	-0.373962	$7.9833 \times 10^{-5}$

## SECTION 5.0 - COST/STRENGTH MODELS

### 5.1 Manufacturing Process

From a product reliability point of view, the manufacturing-assembly-test process can be viewed as a "machine" for identifying and removing hardware defects that are induced through the use of defective parts (resistors, capacitors, ICs, etc.), poor designs and assembly errors. This machine can be effective by removing a large number of defects at a reasonable cost, or ineffective by removing only a small number of defects at a high cost and not meeting product reliability requirements imposed by the customer.

Figure 5.1 represents the various levels of assembly and test/rework stations at each level of a generalized manufacturing process. The process is cyclic in the sense that defects that are "caught" by a particular test may not be corrected (and removed from the process) but instead would go back into the process incurring additional test and rework costs. The symbols used to identify the process are defined below:

- $A_j$  =  $j^{\text{th}}$  assembly level
- $T_{jk}$  =  $k^{\text{th}}$  test of  $j^{\text{th}}$  assembly level
- $I_{jk}$  = Inspection/verification station for  $k^{\text{th}}$  test of  $j^{\text{th}}$  assembly level
- PDEF = Quantity of part defects initially present
- $ADEF_j$  = Quantity of assembly defects initially present at the  $j^{\text{th}}$  level
- $X_{ijk}$  = Number of defects present at the start of the  $k^{\text{th}}$  test,  $j^{\text{th}}$  assembly level, during the  $i^{\text{th}}$  test/rework cycle.
- $Q_{ijk}$  = Probability of passing a defect from the  $k^{\text{th}}$  test to the  $k + 1^{\text{st}}$  test,  $j^{\text{th}}$  assembly level during the  $i^{\text{th}}$  test/rework cycle.
- $P_{ijk}$  =  $1 - Q_{ijk}$  = joint probability of raising a defect to a detectable level and detecting the defect with the test equipment employed at the  $k^{\text{th}}$  test,  $j^{\text{th}}$  assembly level, during the  $i^{\text{th}}$  test/rework cycle.
- $F_{ijk}$  = Fraction of the time a defect results in test/rework at the "card" assembly level (i. e., at  $j=1$ ). The remaining portion of the time,  $1 - F_{ijk}$ , defects are tested/reworked at the  $j^{\text{th}}$  assembly level.
- $R_{1ijk}$  = Probability that a defect detected at the  $k^{\text{th}}$  test,  $j^{\text{th}}$  assembly level, during the  $i^{\text{th}}$  cycle is corrected (and removed from the process) during rework at the  $j^{\text{th}}$  card level.
- $R_{2ijk}$  = Probability that a defect detected at the  $k^{\text{th}}$  test,  $j^{\text{th}}$  assembly level, during the  $i^{\text{th}}$  cycle is corrected (and removed from the process) during rework at the  $j^{\text{th}}$  assembly level.

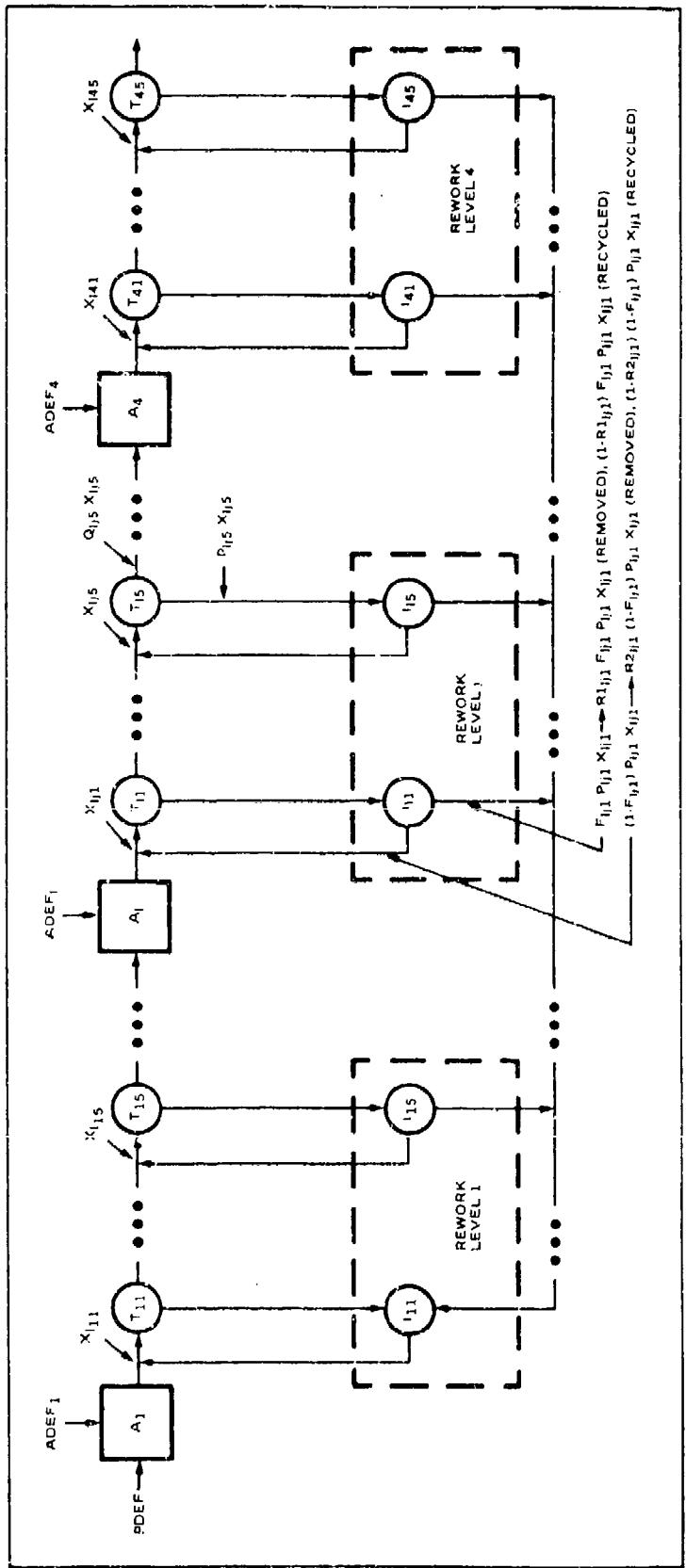


Figure 5.1. Generalized Manufacturing Assembly – Test Rework Process

Thus, at the first cycle ( $i=1$ ) of test station  $T_{jk}$ ,  $X_{1jk}$  defects (part and assembly defects that were not caught in the previous tests) are tested (stressed).  $Q_{1jk} X_{1jk}$  defects are not caught and are passed on to the next test and  $P_{1jk} X_{1jk}$  defects are detected and sent to the  $I_{jk}$  inspection station (for determination of defect type: card level or assembly level).  $F_{1jk} P_{1jk} X_{1jk}$  result in card level rework and  $(1 - F_{1jk}) P_{1jk} X_{1jk}$  result in assembly ( $j$ th) level rework. Of the number of defects resulting in card level rework,  $R_{1jk} F_{1jk} P_{1jk} X_{1jk}$  will actually be corrected and removed from the process and  $(1 - R_{1jk}) F_{1jk} P_{1jk} X_{1jk}$  will not be corrected. A similar breakdown of defects exists for assembly level rework. At the next cycle ( $i = 2$ ), the number of defects entering the  $T_{jk}$  test station (i. e.,  $X_{2jk}$ ) is given by:

$$X_{2jk} = Q_{2jk} X_{2jk-1} + (1 - R_{1jk})(1 - F_{1jk}) P_{1jk} X_{1jk} \quad (1)$$

The first term on the right side of (1) represents defects passed by the previous test on the second cycle and the second term represents the assembly ( $j$ th level) defects that were not removed from cycle 1.

The general equations representing the entire manufacturing process are given in Figure 5.2. The equations are recursive with initial conditions as shown. A computer routine was developed to solve these equations and is provided as a subroutine (SCREEN) in the Screening and Debugging Optimization (SDO) model. The solutions  $\{X_{njk}\}_{jk}$  provide the number defects present at the start of the  $n$ th cycle for test stations  $\{T_{jk}\}_{jk}$  so that for a given set of tests, the test strength TS is given by:

$$TS = 1 - \frac{D_F}{D_I} \quad (2)$$

where:

$$D_I = PDEF + \sum_{j=1}^M ADEF_j; \quad (\text{incoming defects})$$

$$D_F = \sum_{n=1}^{NCYC} Q_{nM} N(M) X_{nM N(M)} \quad (\text{outgoing defects - i. e., number remaining in the system})$$

PDEF = number of part defects

$ADEF_j$  = number of assembly defects introduced at the  $j$ th assembly level

M = number of assembly levels

NCYC = maximum number of process cycles

$Q_{nM} N(M)$  = probability of passing a defect at final test (N(M)) of final level of assembly (M) during the  $n$ th cycle.

$X_{nM N(M)}$  = number of defects present at the  $N(M)$ th test,  $M$ th assembly level during the  $n$ th cycle.

$$X_{i+1,j-1,N(j-1)} = \begin{cases} P_{i+1,j-1,N(j-1)} X_{i+1,j-1,N(j-1)} + (1-P_{ijk}) (1-F_{ijk}) (1-R2_{ijk}) X_{ijk}, & \text{for } K=1, j>1 \\ P_{i+1,jK-1} X_{i+1,jK-1} + (1-P_{ijk}) (1-F_{ijk}) (1-R2_{ijk}) X_{ijk}, & \text{for } K>1, j\geq 1 \end{cases}$$

$$X_{i+1,jK} = \sum_{n=1}^M \sum_{m=1}^{N(n)} (1-P_{inn}) F_{inn} (1-R1_{inn}) X_{inn}, \text{ for } K=1, j=1$$

Initial Conditions

$$X_{111} = ADEF_1 + PDEF$$

$$X_{1j1} = P_{1j-1,N(j-1)} X_{1j-1,N(j-1)} + ADEF_j \text{ for } j>1$$

$$X_{1jk} = P_{1jk-1} X_{1jk-1} \text{ for } K>1$$

Figure 5.2. General Manufacturing Process Equations

The corresponding cost function (TCOST) representing the total test and rework costs incurred during NCYC cycles is given by:

$$TCOST = \sum_{n=1}^{NCYC} \sum_{j=1}^M \sum_{k=1}^{N(j)} [CT_{jk} + CR_{njk} (1 - Q_{njk}) X_{njk}] \quad (3)$$

where:

$CT_{jk}$  = test cost for conducting test  $T_{jk}$

$CR_{njk}$  =  $CR1_{jk} F_{ijk} + CR2_{jk} (1 - F_{ijk})$

$CR1_{jk}$  = rework cost per cycle for defects identified for card level rework at the  $I_{jk}$  inspection station.

$CR2_{jk}$  = rework cost per cycle for defects identified for assembly level rework at the  $I_{jk}$  inspection station.

Other terms are as defined previously.

The manufacturing process represented by Figure 5.1 identifies the test stations ( $T_{jk}$ ) at each level as separate events. The purpose of this separation is to compare test strengths and associated costs of individual screens as required for screen optimization (described in the next section). However, in actual practice, most testing at the same level of assembly is conducted in parallel. Thus, temperature cycling, vibration and power cycling may be used in a combined test (e.g., Test Level E of MIL-STD-781B).

It is generally felt that a combined test applies more stress on a unit than when the same tests are conducted separately because of the stress interactions. However, no information from the literature search or from internal data was found to support this contention, and therefore, the test strengths used do not reflect any stress interaction. The resulting effect of not including this additional stress of a combined test is a more conservative solution.

In addition, the manufacturing process assumes that defects which are not removed during a given rework cycle are (1) introduced back into the same test if classified to be an assembly level defect, or (2) put back into the first test if classified to be a card level defect. This is the baseline manufacturing policy chosen for the study. An alternate policy commonly used is to pass defects to the next test in sequence. In this way, defects that are caught at, say, test station  $T_{jk}$  and which are not removed at assembly level rework would not return to test station  $T_{jk}$  but, instead, to test station  $T_{jk+1}$ . Only minor changes in the process equations (Figure 5.2) are needed to represent this type of testing policy. Accordingly, Figure 5.3 gives the corresponding equations with the necessary changes.

## 5.2 Optimization Algorithm

An optimization algorithm has been developed to solve two related problems in the use of screening/debugging tests to reduce the number of part and manufacturing assembly defects. Briefly stated, these two problems are: (1) how

$$\left\{ \begin{array}{l} P_{i+1} j-1 N(j-1) X_{i+1 j-1 N(j-1)} + (1-P_{ij-1 N(j-1)}) (1-F_{ij-1 N(j-1)}) (1-R^2_{ij-1 N(j-1)}) X_{ij-1 N(j-1)} \\ \text{for } k=1, \quad j>1 \end{array} \right.$$

$$X_{i+1jk} = \left\{ \begin{array}{l} P_{i+1 jk-1} X_{i+1 jk-1} + (1-P_{ijk-1}) (1-F_{ijk-1}) (1-R^2_{ijk-1}) X_{ijk-1}, \quad \text{for } k>1, \quad j \geq 1 \end{array} \right.$$

$$\sum_{n=1}^M \sum_{m=1}^{N(n)} (1-P_{inm}) F_{inm} (1-R^1_{inm}) X_{inm}, \quad \text{for } k=1, \quad j=1$$

Initial Condition

$$X_{111} = ADEF_1 + PDEF$$

$$X_{ij1} = P_{ij-1 N(j-1)} X_{1j-1 N(j-1)} + ADEF_j \quad \text{for } j>1$$

$$X_{ijk} = P_{ijk-1} X_{1jk-1} \quad \text{for } k>1$$

Figure 5.3. General Manufacturing Process Equations (Alternate Retest Policy)

to remove a given number of hardware defects (bad parts, assembly errors, design errors, etc.) in the manufacture of a system at a minimum total cost, and (2) given a fixed "not-to-exceed" dollar budget, what screening tests should be conducted to minimize the number of defects getting into the final system.

Unless a manufacturer's resources are severely constrained with respect to testing facilities or the product being manufactured is of a simple nature (e.g., a single level of assembly), the solution to (1) or (2) is not an easy one. The manufacturer not only has a choice of various types of tests (power conditioning, temperature cycling, vibration, etc.) and severities (duration of test, temperature extremes, vibration amplitude, etc.) but must choose where to place his test selections in the manufacturing assembly process. For example, in a situation where there are three types of tests, each one of which has three test parameters and a selection of five values for each parameter, and any combination of the three tests can be conducted at four different levels of assembly

(e.g., card, unit, equipment and system), then there would be a total  $5^{36} \approx 1.4 \times 10^{25}$  possible test sequences, one of which would be optimal.

A given test sequence is considered to be better (more optimal) than another if it provides the same or higher screening strength at a lower cost, or for the same or lower cost it provides a higher screening strength. The measures used in this algorithm to determine optimality are:

$$TS \equiv (\text{total defects removed}) / (\text{total defects introduced})$$

$$TC \equiv (\text{average cost per removed defect})$$

Figure 5.4 outlines the computation procedure for the algorithm. The SDO model provides a selection of five test types (constant temperature, temperature cycling, vibration, power continuous and power cycled) and four assembly levels (card, unit, equipment, system). The number of test parameters are 2, 4, 4, 1, 2 for tests 1 through 5, respectively, and the maximum number of steps (values) for each parameter have been set at 4 (including the parameter values which eliminate the test). Thus, the total number of test sequence combinations possible for the maximum case is:

$$(4^2 \cdot 4^4 \cdot 4^4 \cdot 4 \cdot 4^2)^4 = 4^{52} \approx 2 \times 10^{31}.$$

At the first step, test  $T_{11}$  is combined with test  $T_{12}$  to form the sequence  $\{u_\ell, v_\ell\}_{\ell=1, \dots, 4^6}$  defined as follows:

$$u_\ell = TS(\ell, j) [1 - TS(\ell-1, i)] + TS(\ell-1, i) \quad (1)$$

$$v_\ell = TC(\ell, j) TS(\ell, j) [1 - TS(\ell-1, i)] + TC(\ell-1, i)/\ell \quad (2)$$

for  $i = 1, \dots, 4^2$  (combinations of  $T_{11}$ ) and  $j = 1, \dots, 4^4$  (combinations of  $T_{12}$ ). This sequence is ranked from lowest to highest cost ( $v_\ell$ ) and a dominant sequence is formed by removing combinations in which the test strength ( $u_\ell$ ) is lower than the preceding combination (i.e.,  $u_\ell < u_{\ell-1}$ ). The dominant sequence

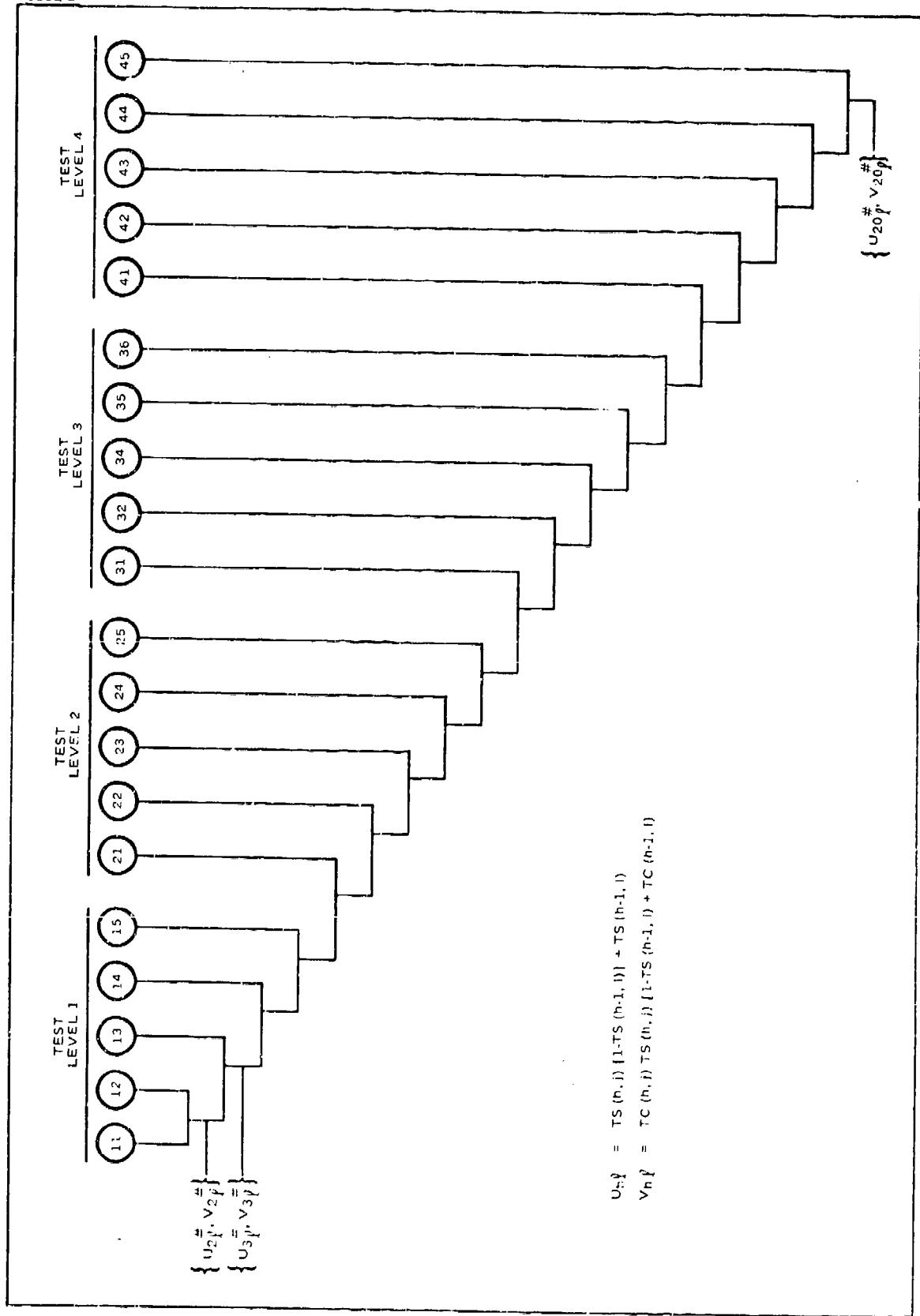


Figure 5.4. Computation Procedure for the Optimization Algorithm

formed in this way (say  $\{u_{\ell}^*, v_{\ell}^*\}$ ) provides the input necessary for the next step, i. e.,

$$TS(\ell-1, i) = u_{\ell}^*$$

and

$$TC(\ell-1, i) = v_{\ell}^*$$

This dominant sequence is then combined with  $T_{13}$  according to equations (1) and (2) above to form the next sequence. The procedure is continued until all test combinations have been exhausted. The final dominant sequence, say  $\{u_{\ell}^*, v_{\ell}^*\}$ , is therefore optimal in the following sense:

1. If  $m > n$  then  $u_m^* \geq u_n^*$  and  $v_m^* \geq v_n^*$  (i. e., the sequence is monotone never decreasing in both  $u_{\ell}^*$  and  $v_{\ell}^*$ )
2. If  $(u', v')$  represent the test strength and cost, respectively, of any other sequence of tests which does not belong to  $\{u_{\ell}^*, v_{\ell}^*\}$ , then there is a test sequence that does belong to  $\{u_{\ell}^*, v_{\ell}^*\}$  which dominates  $(u', v')$ .

The successively larger number of test sequence combinations produced at each step can also be reduced by eliminating terms from the dominant sequences that are too close to matter. For example, costs and test strengths that differ by less than one percent could be removed and would not appear in a dominant sequence. This would not produce a "pure optimum" solution but would produce a practical "near optimum" solution.

Figure 5.5 gives a flow diagram of the optimization procedure. The procedure has been computerized (written in FORTRAN IV) for processing on an IBM 360/370 computer and consists of a MAIN calling routine and six subroutines which are defined as follows:

DATA — This subroutine reads and writes all input data with the exception of individual test cost parameter values which are read in for each test in the MAIN. Default values for most parameters are also defined in the event no user data is available.

SSPROB — This subroutine: (1) calculates the screening strength probabilities of each of five tests as a function of test parameter values, (2) calculates single cycle test strengths ( $P_{ijk}$  values of the previous topic) based on test equipment detection probabilities and the screening strength, and (3) calculates test cost based on the duration of the test.

SCREEN — This routine models the manufacturing process and calculates the total test strength and cost of a specified test sequence.

RANK — This routine ranks a given set of test combinations by cost from lowest to highest.

SEARCH — This routine searches through the optimal sequence for the combination that satisfies the specified requirement (i. e., MTBF requirement or fixed cost requirement).

REPORT — This routine decodes the selected test sequence into the original test parameter values and writes an output report which (1) identifies each test and test parameter value, and (2) summarizes the cost for each test and level of assembly.

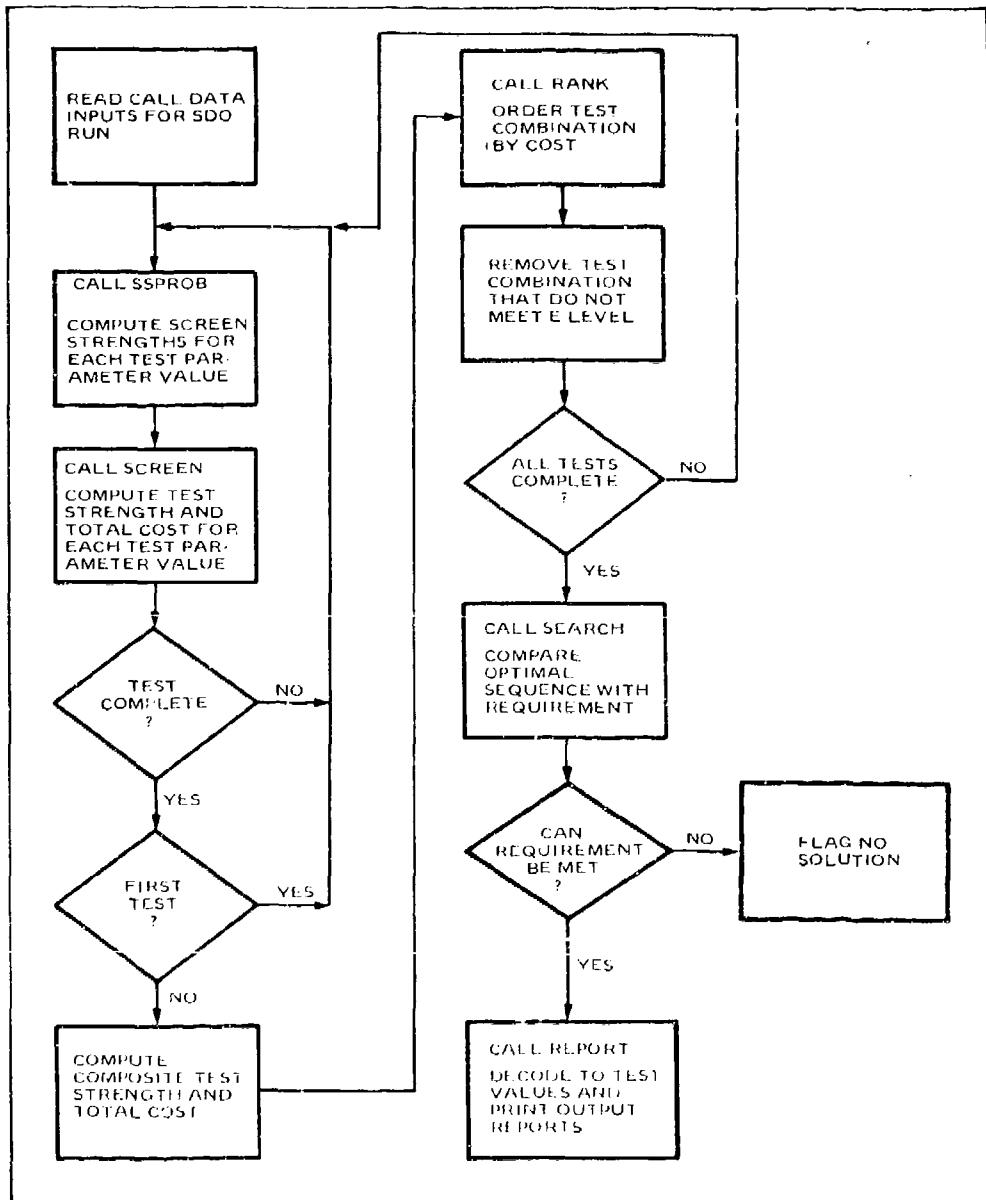


Figure 5.5. Flow Diagram of Optimization Procedure

## SECTION 6.0 - PROCEDURES AND EXAMPLES

### 6.1 Procedure for Using the SDO Model

#### Data Input Requirements and Fall-back Options

In order to operate the SDO model, certain input data requirements must be provided. Some data must be provided by the user such as an estimate of the system failure rate (e.g., based on a handbook prediction), the customer required failure rate and system complexity (part count). Other data is optionally provided by the user with a default to SDO model supplied values based on study findings. Table 6.1 summarizes the data requirements, parameter symbol, computer input format, units, source and fall-back option. The following paragraphs describe the data requirements and parameters in more detail.

#### Test Characteristics

The test cost function has the form:  $CT = A_1 + (B_1) (\text{HOURS})$  where  $A_1$  and  $B_1$  are the fixed and variable costs, respectively, and HOURS represents the duration of the test.  $CT$  represents the total cost of conducting the test and is expressed in dollars. The option is exercised whenever  $B_1$  is zero. In this case,  $CT$  is expressed in hours. The option on fixed test cost ( $A_1 = 0.0$ ) assumes that no large capital investment in test equipment is needed to implement any of the test sequences being considered. The option on variable test costs is based on Hughes experience in testing at various levels of assembly.

Screening strength (SS) is computed automatically using subroutine SSPROB by evaluating the CREDIT (ref. 42) equations for screening strength at selected points of the test parameters. The probability of detecting a defect ( $P_d$ ) with test equipment is dependent on the screen used, the level of assembly and the number of test rework cycles.  $P_d$  is an array defined by:  $P_d = P_d(I, I_1, I_2)$  where  $1 \leq I \leq N_{CYC}$  cycles,  $1 \leq I_1 \leq M$  assembly levels and  $1 \leq I_2 \leq 5$  tests. If any value of  $P_d$  is zero, the default to  $P_d = 0.75$  is exercised.

The parameters of each test are sequenced through values in the optimization process described in the previous section. The minimum and maximum limits for each test parameter are optionally specified by the user. These limits are denoted by  $A_{\text{MIN}}{}_{jk}$  and  $A_{\text{MAX}}{}_{jk}$  representing the minimum and maximum limits of the  $k$ th parameter of the  $j$ th test. The standard values of these limits are provided by the SDO model as default values in the following table.

Test Parameter	Type of Test				
	Constant Temperature	Cycled Temperature	Vibration	Constant Power	Cycled Power
Parameter No. 1	Temp Extreme	Upper Temp	G-Level	Time	Time
max	70°C	70°C	6	168	8 hours
min	-55°C	25°C	1	0	0 hours
Parameter No. 2	Test Time	Lower Temp	Time	-	Cycles
max	170 hours	-50°C	2 hours	-	168
min	0 hours	-25°C	0 hours	-	3
Parameter No. 3	-	Temp Rise	Temp	-	-
max	-	10°C/min	25°C	-	-
min	-	0°C/min	25°C	-	-
Parameter No. 4	-	Cycles	Cycles	-	-
max	-	64	64	-	-
min	-	10	10	-	-

Manufacturing Process - Fixed rework cost is specified as a factor of the variable rework cost which is furnished by the user. If the variable rework cost (B2 for card level and B3 for higher levels) is zero, the SDO model default is exercised as follows:

Location of Repair	Manufacturing Assembly Level			
	1	2	3	4
Card level (B2)	0.5	9.46	51.5	63.0
Higher Assembly Level (B3)	0.1	3.67	45.5	57.0

The above quantities are expressed in man-hours and are averages based on a wide variety of card and assembly types (see page 42).

Fraction defective sent to card level repair ( $F$ ) is the expected fraction of defects occurring at higher assembly levels which are sent to the card level

TABLE 6.1. Input Data Requirements/Options

Data Requirements	Prog Symbol	Input Format	Units	Source	Fall-back Option
<u>Test Characteristics</u>					
Fixed Test Cost	A1	Real	Dollars	User	Assumed zero
Variable Test Cost	B1	Real	Dollars/Hour	User	Provided as a function of test duration
Screening Strength	SS	—	—	SDO Model	None
Detection Probability	P	—	—	User	Default to 0.75
Test Parameter Limits	A <sub>MINjk</sub> -A <sub>MAXjk</sub>	Real	(See test description)	User	Standard ranges provided by SDO model
<u>Mfg Process</u>					
Fixed Rework Cost	—	—	—	Function of variable cost	None
Variable Rework Cost	B2, B3	Real	Hours	User	Provided as a function of the rework level
Fraction Defectives to Card Rework	F	Real	—	User	Provided as a function of the rework level
Fraction Defectives Removed at Card Rework	R1	Real	—	User	Default to 0.5
Fraction Defectives Removed at Assy Rework	R2	Real	—	User	Default to 0.5
Number of Assy Levels	M	Integer	—	User	4-levels maximum
Maximum Number of Test-Rework Cycles	NCYC	Integer	—	User	10-cycles maximum
Assy Defects	ADEF	Real	Fraction of total parts	User	Default to 2.3% of total parts for each level
Part Quality Defects	PDEF	Real	Fraction of total parts	User	Default to 1.0% of total parts

TABLE 6.1. Input Data Requirements/Options (Continued)

Data Requirements	Prog Symbol	Input Format	Units	Source	Fall-back Option
<u>System Description</u>					
Complexity (total parts)	NPARTS	Integer	-	User	None
Predicted Failure Rate*	FRM	Real	per million hours	User	None
Field Stress (3-year operation)**:					
Temperature Rise Time	PMAX21	Real	Deg C/min	User	
Duration of Vibration	PMAX32	Real	Hours	User	
Duration of on-off cycle	PMAX51	Real	Hours	User	
Number of Cycles	CYCMAXjk	Real	-	User	
<u>Program</u>					
Option A:					
Required Failure Rate	FRF	Real	per million hours	User	None
Option B:					
Cost Budget	CREQD	Real	Dollars	User	If cost defaults are used, cost budget must be in hours
Number Test Values	ITV	Integer	-	User	Recommend 4 maximum
Tolerance	E	Real	-	User	Recommend 0.01 minimum

\*Required for program option A only

\*\*Three types of field stress are provided : (1) temperature cycling, (2) vibration and (3) power on-off cycling.

for repair. All other defects are repaired at the assembly level in which detection took place. F is a function of the screen used, the level of assembly and number of test rework cycles. If zero values are provided for by the user, the SDO model default is exercised as follows:

Manufacturing Assembly Level				
	1	2	3	4
F	1.0	0.23	0.43	0.65

The above values are based on Hughes' manufacturing experience at the various assembly levels.

Fraction defectives corrected (R1 for the card level and R2 for the higher assembly level) are functions of the screen used, the level of assembly and the number of test-rework cycles. If zero values are provided by the user for R1 or R2 the SDO model default is exercised (i.e., R1 = 0.5 and R2 = 0.5).

The number of assembly levels (M) is based on a card-unit-equipment-system assembly structure. A value of M equal to 4 will use the complete structure and smaller values of M will use a limited structure. For example, a user may only build to the equipment level in which case he would set M = 3.

The number of defects entering the process is based on fraction defectives for parts (PDEF) and assembly errors (ADEF), the total number of parts used in the system and the number of systems being produced. PDEF is input as a fraction of the total parts (NPARTS) and defaults to 0.01 (i.e., 1% of the total number of parts used in system) when user data is not available. Similarly, ADEF defaults to 0.023 (2.3% of the total parts) for all assembly levels when user data is not available. The maximum number of test-rework cycles (NCYC) a single defect would see is dependent upon the complexity and testability of the hardware. Since the number of defects is based on the total parts required for the system, NCYC should be sufficiently large to exhaust the process (i.e., no defects remaining in rework), otherwise the systems being assembled are not complete.

#### System Description/Program Data

System description data is required whenever the test sequence selection is driven by a product reliability requirement (Option A). In this case the model requires the predicted failure rate (FRM) for the "mature" system (e.g., in accordance with MIL-HDBK-217), the required (by a customer specification) system failure rate (FRF) and the total number of parts (NPARTS) used in all systems being produced. If cost is the driving factor in the test sequence selection (Option B), NPARTS and the total cost budget (CREQD) are required. CREQD includes the budget for the total test cost plus the total rework cost necessary to assemble all systems.

Field stress is characterized by three "tests": temperature cycling, vibration, and power cycling. Thus, defects that are present in the system at delivery will continue to show up in the field (together with random failures) until the system reaches "Maturity" (i. e., no more defects). At this point in time, the system still fails but only due to "random" failures. All latent defects have been removed by the field stress.

The SDO model defaults to a field stress (i. e., a "test" strength) of 1.0 and three operating years to maturity (26,280 hours). The user can optionally provide the estimated amount of field stress the system will experience until maturity is reached by specifying  $P_{MAX_{jk}}$  for the appropriate tests and  $CYCMAX_{jk}$ .

The number of test values (ITV) represents the number of values each test parameter takes on in computing screening strengths for each test. The value of ITV has a significant effect of the computer running time and core requirements. It is, therefore, recommended that the value of ITV not exceed 4. The tolerance (E) has a similar effect on computer running time and core requirements and may be varied with ITV to get better usage of the model. It is recommended, however, that E does not go below 0.01. This value corresponds to eliminating from further consideration those test sequences that are closer than 1% in cost or test strength at each step of the optimization process.

#### Procedure and Examples

The examples given in this section detail the step-by-step procedure for using the SDO model. Example 1 describes the procedure for determining the optimal screen under a fixed cost constraint and Example 2 describes the procedure for determining the least cost screen for meeting a reliability requirement. The following general operational description applies to all procedures for processing input data and execution of the SDO model on an IBM 360/370 computer.

Job Control Language (JCL) - The following statements are required for allocating storage and assignment of input data files:

```
//TF19556A JOB (2,                                     GENERATED JOB STATEMENT
// 606,T09520,00,42,SNUMB),'K218,JAMES, L E      ',CLASS=B,REGION=500K,
//           NOTIFY=TF19556,
//           MSGLEVEL=(1,1)
/*MAIN  ORG=RM029
// EXEC FORTGO,GOPGM=TEMPNAME,TIME=10
//STEPLIB DD DISP=SHR,DSN=TF19556, SD2.MAIN.LOAD
//GO,FT04F001 DD DISP=SHR,DSN=TF19556.PROGRM.DATA
//GO,FT11F001 DD DISP=SHR,DSN=TF19556.PD.DATA
//GO,FT08F001 DD DISP=SHR,DSN=TF19556.F.DATA
//GO,FT09F001 DD DISP=SHR,DSN=TF19556.R.DATA
```

```
//GO, FT10F001 DD DISP=SHR, DSN=TF19556.AB.DATA  
//GO, FT12F001 DD DISP=SHR, DSN=TF19556.LIMITS.DATA  
//GO, FT13F001 DD DISP=SHR, DSN=TF19556.OPS.DATA
```

Definition of Input Data Sets - The above JCL creates the data sets noted below. The asterisks denote that the associated parameter has a default value and the numbers in the parentheses give the maximum dimension for arrays.

**File 04 PROGRM** - This data set contains data for the parameters NCYC, M, NPARTS, CREQD, E, ITV, FRF, FRM, PDEF\*, ADEF(5)\*, N(5), B2, B3.

**File 08 F-** This data set contains data for the array F(10, 4, 5)\*

**File 09 R** - This data set contains data for the arrays R1 (10, 4, 5)\* and R2 (10, 4, 5)\*

**File 10 AB** - This data set contains data for the parameters A1\*, B1\*

**File 11 PD** - This data set contains data for the array P (10, 4, 5)\*

**File 12 LIMITS** - This data set contains data for upper and lower limits on each test parameter

**File 13 OPS** - This data set contains test parameter data for simulating field stress

All files must be filled. If the default is to be exercised, zero values for all parameters and array elements must be used. The input form for each data set is "unformatted" (i.e., the values for each record are simply separated by commas - this is illustrated in the examples given below).

Diagnostics - If the number of combination of test sequences becomes too large for a particular choice of number of test parameter values (ITV) and tolerance (E), a subscript error check will occur. The error condition can be removed by: (1) increasing the dimension on the appropriate arrays, (2) decreasing the value of ITV, and/or increasing the value of E.

Output Reports - All input data used in the SDO model is printed out in table form. If defaults are exercised the default values are printed in the tables. The optimal test sequence is also printed out in table format and the following table gives a cross-reference index between test parameter number and definition for each test:

**Test Parameter Cross Reference**

	Test Parameter			
	No. 1	No. 2	No. 3	No. 4
Constant Temperature (CT)	Temp Extreme (TA) °C	Test Time (TT) hours	—	—
Cycled Temperature (CYT)	Upper Temp (TU) °C	Lower Temp (TL) °C	Temp Rise (TR) °C per minute	Number cycles (CY)
Vibration (VIB)	Vibration G-level (V)	Time (TM) hours	Temp (TV) °C	Number cycles (CY)
Constant Power (CP)	Time (TM) hours	—	—	—
Cycles Power (CYCP)	Time (TM) hours	Number cycles (CY)	—	—

#### 6.2 Fixed Cost-Optimum Screen

A ground display equipment manufacturer has a limited budget for funding a test-conditioning rework effort during production and, therefore, desires to minimize the number of defects getting into the final production systems and still remain within budget.

- Step 1** - Assembly Required Data for Coding.
- NCYC - Number of test-rework cycles is set at 10 as adequate to exhaust the process.
- M - The number of manufacturing assembly levels is three: card assembly, unit assembly and equipment assembly.
- NPARTS - The display contains a total of 10,000 parts per system and 60 systems are planned for production (i.e., NPARTS = 600000).
- CREQD - The total cost budget for testing and rework is \$500,000.
- E - Test screens that are closer than 1% in cost are considered to be equal (i.e., E = 0.01).

**ITV** - Three values of a test parameter are felt to be adequately sensitive (i.e.,  $ITV = 3$ ).

**FRF, FRM** - These parameters are set to zero since the cost option is being used (i.e.,  $FRF = 0.0$ ,  $FRM = 0.0$ ,  $P_{MAX_{jk}} = 0.0$ , and  $CYCMAX_{jk} = 0.0$ )

**A<sub>MAX</sub><sub>jk</sub>, A<sub>MIN</sub><sub>jk</sub>** The SDO model defaults are used for these parameters.

**PDEF** The average quality level of parts used in the display is 1.5% defective based on the manufacturer's field usage history.

**ADEF** - The manufacturer's production records show that he can expect 0.2%, 0.3%, and 0.3% defectives due to assembly errors, wiring errors and generally poor workmanship at the card, unit, and equipment levels, respectively.

**P** - The detection probabilities of the test equipment used by test department is assumed to be the same as the SDO model default value (i.e.,  $P = 0.75$ ).

**F** - The manufacturer has found that 80% of unit and equipment failures isolate to a card failure. All others are unit and equipment level assembly/wiring errors.

**R1** - Card level repair records show that 80% of the defects are removed at each rework cycle.

**R2** - Assembly level repair records show that 50% of the defects are removed on each rework cycle.

**B2, B3** - The average repair costs per defect are given (in dollars) as follows:\*

Type of Repair \ Level	Card Test	Unit Test	Equipment Test
Card (B2)	5.	60.	250.
Assembly (B3)	5.	25.	60.

**A1, B1** - The average test cost for level are given below in dollars for A1 and dollars per hour for B1 as follows:

---

\*The default option could also be used in which case the results would be expressed in hours rather than dollars (see page 54).

Test Cost	CT			CYCT			VIB			CP			CYCP		
	C	U	E	C	U	E	C	U	E	C	U	E	C	U	E
Fixed Cost (A1)	225.	250.	250.	225.	250.	250.	225.	250.	250.	100.	250.	100.	225.	225.	225.
Var Cost (B1)	10.	10.	10.	25.	25.	25.	25.	25.	25.	10.	10.	10.	25.	25.	25.

C: card level testing, U: unit level testing, E: equipment level testing

**Step 2** - Coding Input Data. Based on a standard 80-column IBM card, the input for each data set in this example is specified below. The order is the same as in Step 1 and each line represents a single record.

File 04 (PROGRM)

Col. No. 1

10,3,600000,500000,.01,3,0.,0.

.015  
.002  
.003  
.003

5.,5.  
5.,5.  
5.,5.  
5.,5.  
5.,5.  
60.,25.  
60.,25.  
60.,25.  
60.,25.  
60.,25.  
250.,60.  
250.,60.  
250.,60.  
250.,60.  
250.,60.

File 11 (PD)

Col. No. 1

0.  
0.  
:  
fifteen entries  
:  
:  
0.

File 08 (F)

Col. No. 1

.4  
.4  
:  
:  
:  
.4  
  
fifteen entries

File 09 (R)

Col. No. 1  
↓  
.8,.5  
.8,.5  
... } fifteen entries  
...  
.8,.5

File 10 (AB)

Col. No. 1  
↓  
225.,10.  
225.,25.  
225.,25.  
100.,10.  
225.,25.  
250.,10.  
250.,25.  
250.,25.  
250.,10.  
225.,25.  
250.,10.  
250.,25.  
100.,10.  
225.,25.

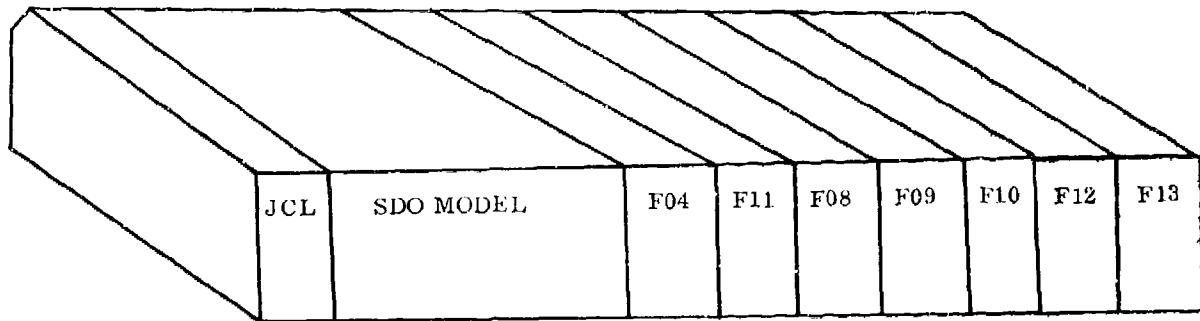
File 12 (LIMITS)

Col. No. 1  
↓  
0.,0.  
⋮ 13 entries  
0.,0.

File 13 (OPS)

Col. No. 1  
↓  
0.,0.  
0.,0.  
0.,0.

**Step 3** - Prepare Card Deck for Processing. The card deck must be put  
in the following order for processing:



- Step 4** - Output Reports. The output reports for this example are given in Tables 6.2 through 6.9. Table 6.10 gives the optimal (highest test strength) screening sequence for the budget specified. Therefore, the best the manufacturer can do for the given budget is a test strength of .7727 which indicates that he will eliminate 77.3% of the defects per equipment at a total cost of \$494,240 or \$8,237 per system.

### 6.3 Fixed Reliability - Minimum Cost

A radar manufacturer has a customer requirement for a 300-hour system MTBF. Based on MIL-HDBK-217 he estimates his radar system has a mature MTBF of 500 hours. The manufacturer would like to determine the minimum amount of screening (i.e., least cost) which would allow him to meet his customer's requirement. In addition to choosing screening tests, the manufacturer also has a choice of using higher quality parts in his system which would increase the mature system MTBF to 1000 hours but would increase the cost per system. However, this action would also reduce the number of part defects entering the manufacturing process thereby reducing test rework costs. The manufacturer's problem, therefore, is a tradeoff between part quality and amount of test screening to determine (1) whether he should buy the more expensive parts to use in his system, and (2) what screens should he implement to meet the required 300-hour MTBF.

- Step 1** - Assemble Required Data for Coding.
- NCYC - Number of test rework cycles is set at 10.
- M - Four manufacturing levels are used: card, unit, equipment and system.
- NPARTS - The radar contains a total of 20,000 parts per system and 40 systems are planned for production (i.e., NPARTS = 800000)
- CREQD - Not required for the option (i.e., CREQD = 0.0).
- E - Test screens that are closer than 2% in cost are considered equal (i.e., E = .02)
- ITV - Three values of the test parameters are needed (i.e., ITV = 3) to provide the necessary sensitivity.
- FRF (40 systems) -  $133320./10^6$  hours (300-hour MTBF per system) customer requirement for 40 systems.
- FRM (40 systems) -  $80000./10^6$  hours (500 hours MTBF for a mature system using medium quality parts)
- FRM (40 systems) -  $40000./10^6$  hours (1000-hour MTBF for a mature system using high quality parts)

TABLE 6.2  
TABLE A - PROGRAM DATA

NCYC	M	(PDEF,NPARTS)	CREQD	E	ITV	FRF	FRM
10	3	9000.	500000.00	0.010000	3	0.0	0.0

TABLE 6.3

ASSEMBLY LEVEL		NUMBER OF SCREENS	EXPECTED NUMBER OF ASSEMBLY DEFECTS
1		5	1200.
2		5	1800.
3		5	1800.

TABLE 6.4

TABLE C - DETECTION PROBABILITIES FOR TEST EQUIPMENT

.750000000	.750000000	.750000000	.750000000
.750000000	.750000000	.750000000	.750000000
.750000000			

TABLE 6.5

TABLE D - FRACTION OF DEFECTS REQUIRING PRIMARY LEVEL (CARD ASSY) REWORK

.399999976	.399999976	.399999976	.399999976
.399999976	.399999976	.399999976	.399999976
.399999976			

TABLE 6.6

TABLE E - FRACTION OF DEFECTS REMOVED AT THE PRIMARY LEVEL PER REWORK/RETEST CYCLE

.800G00012	.800000012	.800000012	.800000012	.800000012
.800000012	.800000012	.800000012	.800000012	.800000012
.800G00012				

TABLE 6.7

TABLE F - FRACTION OF DEFECTS REMOVED AT THE HIGHER ASSEMBLY LEVEL PER REWORK/RETEST CYCLE

.500000000	.500000000	.500000000	.500000000	.500000000
.500000000	.500000000	.500000000	.500000000	.500000000
.500000000				
.500000000				

TABLE 6.8

TABLE G - PRIMARY LEVEL REWORK COST PER CYCLE

5.00000000	5.00000000	5.00000000	5.00000000	60.0000000
60.0000000	60.0000000	60.0000000	250.000000	250.000000
250.000000				

TABLE 6.9

TABLE H - HIGHER ASSEMBLY LEVEL REWORK COST PER CYCLE

5.00000000	5.00000000	5.00000000	5.00000000	25.0000000
25.0000000	25.0000000	60.0000000	60.0000000	60.0000000
60.0000000				

TABLE 6.10

TEST SEQUENCE LEVEL NO.	TEST NO.	TEST DESCRIPTION PARAMETER VALUE				TOTAL COST
		TYPE NO. 1	NO. 2	NO. 3	NO. 4	
TEST NO. 1	1	CT	70.00	0.0	0.0	329244.50
	2	CYT	47.50	-50.00	10.00	27561.11
	3	VIB	6.00	1.00	25.00	137318.06
	4	CP	84.00	0.0	0.0	28512.21
	5	CYP	8.00	85.50	0.0	20249.07
TEST NO. 2	1	CT	0.0	0.0	0.0	115604.12
	2	CYT	25.00	-25.00	10.00	164995.06
	3	VIB	3.50	2.00	25.00	0.0
	4	CP	0.0	0.0	0.0	0.0
	5	CYP	0.0	0.0	0.0	0.0
TEST NO. 3	1	CT	0.0	0.0	0.0	141504.56
	2	CYT	0.0	0.0	0.0	23490.55
	3	VIB	0.0	0.0	0.0	0.0
	4	CP	0.0	0.0	0.0	0.0
	5	CYP	0.0	0.0	0.0	0.0
TOTAL COST						<u>494239.56</u>
TOTAL SEQUENCE TEST STRENGTH = 0.951277						0.772650

- PMAX<sub>jk</sub>** - CYCMAX<sub>jk</sub> Over a 3-year field usage period the radar is expected to experience temperature cycling and power on-off cycling characterized by approximately 1095 cycles of power (on:2.5 hours and off:2.5 hours, i.e. PMAX<sub>51</sub> = 5 hours and CYCMAX<sub>52</sub>=1095) and 1095 cycles of temperature (at 2°C/min. between extremes, i.e., PMAX<sub>23</sub> = 2 and CYCMAX<sub>24</sub> = 1095).
- PDEF** - The average part quality is estimated at 1% defects for existing parts and 0.5% for high quality parts.
- ADEF** - The average percentage of assembly type defects from the manufacturing process is estimated at 1.2% of the total number of parts per system (i.e., 0.3% at each of four assembly levels). 1
- R1** - Card level repair records show that 80% of the defects are removed on each rework cycle.
- R2** - Assembly level repair records show that 50% of the defects are removed on each rework cycle.
- P, F,** - The manufacturer decided to use SDO model defaults for these parameters in this tradeoff.  
**B2, B3,**  
**A1, B1**
- AMAX<sub>jk</sub>** - **AMIN<sub>jk</sub>** The temperature extremes used in the SDO model were not adequate for the radar system so these were extended to -55°C and 125°C. All other default test values were considered adequate.

**[Step 2] - Coding Input Data**

**Case A: Existing quality parts**

**File 04 (PROGRAM)**

**Col. No. 1**

10, 4, 800000, 0, .01, 3, 133320., 80000.

.010  
 .003  
 .003  
 .003  
 .003  
 0., 0.  
 0., 0.  
 0., 0.  
 . } 20 entries  
 .  
 .  
 0., 0.

**File 11 (PD)**

(All 20 values set to zero)

**File 08 (F)**

(All 20 values set to zero)

**File 09 (R)**

Col. No. 1

.8,.5  
.8,.5  
·  
·  
.8,.5

} 20 entries

**File 10 (AB)**

(All 20 values set to zero)

**File 12 (LIMITS)**

Col. No. 1

125.,-55.  
0.,0  
125.,25.  
-55.,-25  
0.,0.  
·  
·  
0.,0

} 9 entries

**File 13 (OPS)**

Col. No. 1

2.,1095.  
0.,0  
5.,1095.

**Step 3** - Prepare Card Deck for Processing

(Same as previous example)

**Step 4** - Output Reports. The output reports for Case A are given in Tables 6.11 through 6.18. Table 6.19 gives the optimal screening sequence which will meet customer MTBF requirements at minimum cost using medium quality parts. The test strength of the total sequence will remove an estimated 95% of the total defects entering the manufacturing process. Cost is in terms of total labor hours for test and rework effort for the total program (i.e., manufacturer of 40 radar systems). The average cost per system is 4364 hours in test/rework labor.

Case B: High quality parts

The only change to the input data is in File 04 (PROGRM) which is shown below:

File 04 (PROGRM)

Col. No. 1

↓  
10, 4, 800000., 0., 02, 3, 133320., 40000.  
.005  
.003  
.003  
.003  
.003  
0., 0.  
0., 0.  
·  
·  
0., 0. } 20 entries

The output reports for Case B are given in Table 6.20 through 6.27. Table 6.28 gives the optimal sequence for this case.

TABLE 6.11

		TABLE A - PROGRAM DATA						
N CYC	M	(PDEF, NPARTS)	CREQD	E	ITV	FRF	FRM	
10	4	8000.	0.0	0.020000	3	133320.00	80000.00	

TABLE 6.12

		TABLE B - TEST AND ASSEMBLY DATA						
ASSEMBLY LEVEL		NUMBER OF SCREENS	EXPECTED NUMBER OF ASSEMBLY DEFECTS					
1		5					2400.	
2		5					2400.	
3		5					2400.	
4		5					2400.	

TABLE 6.13

		TABLE C - DETECTION PROBABILITIES FOR TEST EQUIPMENT						
ASSEMBLY LEVEL		1	2	3	4	5	6	7
1	.75000000	.75000000	.75000000	.75000000	.75000000	.75000000	.75000000	.75000000
2	.75000000	.75000000	.75000000	.75000000	.75000000	.75000000	.75000000	.75000000
3	.75000000	.75000000	.75000000	.75000000	.75000000	.75000000	.75000000	.75000000
4	.75000000	.75000000	.75000000	.75000000	.75000000	.75000000	.75000000	.75000000

TABLE 6.14

		TABLE D - FRACTION OF DEFECTS REQUIRING PRIMARY LEVEL (CARD ASSY) REWORK						
ASSEMBLY LEVEL		1	2	3	4	5	6	7
1	1.0000000	1.0000000	1.0000000	1.0000000	1.0000000	1.0000000	.230000019	.230000019
2	.230000019	.230000019	.230000019	.430000007	.430000007	.430000007	.430000007	.430000007
3	.430000007	.649999976	.649999976	.649999976	.649999976	.649999976	.649999976	.649999976

TABLE 6.15

TABLE E-FRACTION OF DEFECTS REMOVED AT THE PRIMARY LEVEL PER REWORK/RETEST CYCLE

.600000012	.800000012	.800000012	.800000012	.800000012	.800000012
.800000012	.800000012	.800000012	.800000012	.800000012	.800000012
.800000012	.800000012	.800000012	.800000012	.800000012	.800000012
.800000012	.800000012	.800000012	.800000012	.800000012	.800000012

TABLE 6.16

TABLE F-FRACTION OF DEFECTS REMOVED AT THE HIGHER ASSEMBLY LEVEL PER REWORK/RETEST CYCLE

.500000000	.500000000	.500000000	.500000000	.500000000	.500000000
.500000000	.500000000	.500000000	.500000000	.500000000	.500000000
.500000000	.500000000	.500000000	.500000000	.500000000	.500000000

TABLE 6.17

TABLE G-PRIMARY LEVEL REWORK COST PER CYCLE

.500000000	.500000000	.500000000	.500000000	.500000000	.500000000
9.46000004	9.46000004	9.46000004	51.500000	51.500000	51.500000
51.500000	63.000000	63.000000	63.000000	63.000000	63.000000

TABLE 6.18

TABLE H-HIGHER ASSEMBLY LEVEL REWORK COST PER CYCLE

.100000024	.100000024	.100000024	.100000024	.100000024	.100000024
3.67000008	3.67000008	3.67000008	45.500000	45.500000	45.500000
45.500000	57.000000	57.000000	57.000000	57.000000	57.000000

TABLE 6.19

TEST SEQUENCE LEVEL NO. 1	TEST NO.	TEST DESCRIPTION PARAMETER VALUE			NO. 4	TOTAL COST
		TYPE	NO. 1	NO. 2	NO. 3	
TEST NO. 2	1	CT	0.0	0.0	0.0	0.0
	2	CYT	25.00	-40.00	10.00	64.00
	3	VIB	6.00	2.00	25.00	64.00
	4	CP	84.00	0.0	0.0	0.0
	5	CYP	8.00	168.00	0.0	27466.40
TEST NO. 3	1	CT	125.00	170.00	0.0	0.0
	2	CYT	125.00	-25.00	5.00	64.00
	3	VIB	6.00	2.00	25.00	37.00
	4	CP	0.0	0.0	0.0	0.0
	5	CYP	8.00	168.00	0.0	0.0
TEST NO. 4	1	CT	0.0	0.0	0.0	0.0
	2	CYT	75.00	-40.00	5.00	64.00
	3	VIB	0.0	0.0	0.0	0.0
	4	CP	0.0	0.0	0.0	0.0
	5	CYP	0.0	0.0	0.0	0.0
TEST NO. 5	1	CT	0.0	0.0	0.0	0.0
	2	CYT	25.00	-25.00	10.00	37.00
	3	VIB	0.0	0.0	0.0	0.0
	4	CP	0.0	0.0	0.0	0.0
	5	CYP	0.0	0.0	0.0	0.0
TOTAL COST						174546.37
TOTAL SEQUENCE TEST STRENGTH = 0.951277						

TABLE 6.20

TABLE A - PROGRAM DATA

NCYC	M	(PDEF, NPARTS)	CREQD	E	ITV	FRF	FRM
10	4	4000.	3.0	0.020000	3	133320.00	40000.00

TABLE 6.21

TABLE B - TEST AND ASSEMBLY DATA

ASSEMBLY LEVEL	NUMBER OF SCREENS	EXPECTED NUMBER OF ASSEMBLY DEFECTS
1	5	2400.
2	5	2400.
3	5	2400.
4	5	2400.

TABLE 6.22

TABLE C - DETECTION PROBABILITIES FOR TEST EQUIPMENT

.750000000	.750000000	.750000000	.750000000	.750000000	.750000000
.750000000	.750000000	.750000000	.750000000	.750000000	.750000000
.750000000	.750000000	.750000000	.750000000	.750000000	.750000000

TABLE 6.23

TABLE D - FRACTION OF DEFECTS REQUIRING PRIMARY LEVEL (CARD ASSY) REWORK

1.00000000	1.00000000	1.00000000	1.00000000	1.00000000	.230000019	.230000019
.230000019	.230000019	.230000019	.430000007	.430000007	.430000007	.430000007
.430000007	.649999976	.649999976	.649999976	.649999976	.649999976	.649999976

TABLE 6.24

TABLE E-FRACTION OF DEFECTS REMOVED AT THE PRIMARY LEVEL PER REWORK/RETEST CYCLE			
.800000012	.800000012	.800000012	.800000012
.800000012	.800000012	.800000012	.800000012
.800000012	.800000012	.800000012	.800000012
.860000012	.800000012	.800000012	.800000012

TABLE 6.25

TABLE F-FRACTION OF DEFECTS REMOVED AT THE HIGHER ASSEMBLY LEVEL PER REWORK/RETEST CYCLE			
.500000000	.500000000	.500000000	.500000000
.500000000	.500000000	.500000000	.500000000
.500000000	.500000000	.500000000	.500000000

TABLE 6.26

TABLE G-PRIMARY LEVEL REWORK COST PER CYCLE			
.500000000	.500000000	.500000000	.500000000
.9.4600004	9.4600004	51.5000000	51.5000000
51.5000000	63.0000000	63.0000000	63.0000000

TABLE 6.26

TABLE H-HIGHER ASSEMBLY LEVEL REWORK COST PER CYCLE			
.106000024	.100000024	.100000024	.100000024
3.6700008	3.6700008	45.5000006	45.5000006
45.5000000	57.0000000	57.0000000	57.0000000

TABLE 6.27

TABLE H-HIGHER ASSEMBLY LEVEL REWORK COST PER CYCLE			
.106000024	.100000024	.100000024	.100000024
3.6700008	3.6700008	45.5000006	45.5000006
45.5000000	57.0000000	57.0000000	57.0000000

TABLE 6.28

TEST SEQUENCE LEVEL NO. 1	TEST NO.	TYPE	TEST DESCRIPTION	PARAMETER NO. 1	PARAMETER NO. 2	PARAMETER NO. 3	PARAMETER NO. 4	TOTAL COST
TEST NO. 1	1	CT	0.0	0.0	0.0	0.0	0.0	45346.00
	2	CYT	125.00	-55.00	10.00	37.00	0.0	0.0
	3	VIB	6.00	2.00	25.00	64.00	18346.68	18346.68
	4	CP	84.00	0.0	0.0	0.0	4414.02	4414.02
	5	CYP	8.00	168.00	0.0	0.0	2589.91	2589.91
TEST NO. 2	1	CT	125.00	170.00	0.0	0.0	0.0	19995.40
	2	CYT	75.00	-25.00	5.00	64.00	64.00	51513.27
	3	VIB	6.00	1.00	25.00	64.00	64.00	6053.69
	4	CP	168.00	0.0	0.0	0.0	0.0	17159.95
	5	CYP	8.00	168.00	0.0	0.0	0.0	3646.78
TEST NO. 3	1	CT	125.00	170.00	0.0	0.0	0.0	4657.46
	2	CYT	25.00	-25.00	10.00	64.00	64.00	19995.39
	3	VIB	0.0	0.0	0.0	0.0	0.0	24152.43
	4	CP	0.0	0.0	0.0	0.0	0.0	6053.68
	5	CYP	0.0	0.0	0.0	0.0	0.0	18038.75
TEST NO. 4	1	CT	0.0	0.0	0.0	0.0	0.0	0.0
	2	CYT	0.0	0.0	0.0	0.0	0.0	0.0
	3	VIB	0.0	0.0	0.0	0.0	0.0	0.0
	4	CP	0.0	0.0	0.0	0.0	0.0	0.0
	5	CYP	0.0	0.0	0.0	0.0	0.0	0.0
<b>TOTAL COST</b>								<b>121011.56</b>
<b>TOTAL SEQUENCE TEST STRENGTH = 0.892764</b>								

Comparing the results of Case A and Case B shows that a total savings of 53,535 hours can be realized in test and rework costs of using the higher quality parts. This amounts to a savings per system of 1338 hours. If a labor rate of \$20 per hour is used, \$26,760 per system is saved. This amount must of course be offset by the increased cost per system using the higher quality parts. If there is still a significant savings, the decision would be to use the higher quality parts and screen according to Table 6.28. If there is no savings, the decision would be to use the medium quality parts and screen according to Table 6.19.

It should be noted (also see p. 48) that the tests given in the tables for a given level of assembly can, of course, be conducted at the same time which would further reduce associated testing costs.

## SECTION 7.0 - RECOMMENDATIONS AND CONCLUSIONS

In any development of an SDO model the key variables are test costs, (fixed and variable), screening strength (SS) and probability of detection ( $P_d$ ). In this present study we were able to obtain quantitative information on all three of these parameters.

However, the issue is far from closed on the variables screening strength and probability of detection. Good estimates of these two parameters are required for any tradeoff studies regarding the cost-effectiveness of screening and debugging tests. Unfortunately, beyond this present study, little has been done regarding quantitative functions for SS and  $P_d$ .  $P_d$ , of course, is an "equal" component of test strength (TS). It is entirely conceivable that analytic, or at least quantitative, models for  $P_d$  can be developed. The development would proceed much along the lines of a failure modes and effects analysis. Types of failures could be identified, the type of test equipment needed to detect them and so on. The determination of SS might not be as "easy" as  $P_d$  but it is clearly worth further study. In view of the difficulty in measuring SS, fairly carefully designed experiments would be required to randomize out the superfluous effects such as the (unknown) number of defects entering the screen. Also, care should be taken to control or design out  $P_d$  for the determination of SS.

It is recommended that, as part of applying the model results, the techniques of monitoring and controlling the screening/debugging process be further studied and developed and that a data base be built for verification and refinement of the model inputs.

## SECTION 8.0 - BIBLIOGRAPHY AND REFERENCES

1. AIAA Systems Effectiveness and Safety Technical Committee. The Role of Testing in Achieving Aerospace Systems Effectiveness. American Institute of Aeronautics and Astronautics, New York, New York, January, 1973.
2. Barlow, Richard E., and Campo, Rafael A. Total Time on Test Processes and Applications to Failure Data Analysis. University of California, Berkeley, CA, June 1975.
3. Bear, J. C. "Approach to Reliability for the SM-2 Missile," Proceedings of the 1973 Annual Reliability and Maintainability Symposium, IEEE, New York, New York, 1973, p. 79.
4. Berger, Paul D., and Gerstenfeld, Arthur. "Cost Effective Test Sequencing." 1972 NATO Conference Proceeding: Reliability Test and Reliability Evaluation. California State University, Northridge, September 1972, p. 11-A-1.
5. Biran, David. "Reliability Problems in Electronic Military Equipments" The 8th Convention of Electrical and Electronics Engineers in Israel, Tel Aviv, 1973, p. 11/v.
6. Blanks, H. S. "Accelerated Vibration Fatigue Life Testing of Leads and Soldered Points." Microelectronics and Reliability, Vol. 15, #3, Pergamon Press, Elmsford, New York, 1976, p. 213.
7. Borgars, S. J. "Components Subjected to Continuous Thermal Cycling." IEE Conference: Components and Materials used in Electrical Engineering, Vol. 12, London, England, 1965, p. 26.
8. Burrows, R. W. "Long Life Assurance Study for Manned Spacecraft Long Long Life Hardware," Vols. 1-5, Martin Marietta Corporation, Denver, Colorado, December, 1972.
9. Bussolini, Jacob J. "The Application of Overstress Testing to Failure to Airborne Electronics - A Status Report," Aerospace and Electronic Systems, Vol. AES-4 #2, IEEE; New York, New York, March 1968, p. 142.
10. Colbourne, E. Denis; Coverley, G. D.; and Beherq, S. K. "Reliability of MOS LSI Circuits," Proceedings of the IEEE, Vol. 62 #2, New York, New York, February 1974, p. 244.
11. Cottrell, R. G. "The Simulation of Production Test Economics," Proceedings of the 1974 Annual Reliability and Maintainability Symposium, IEEE; New York, 1974, p. 91.
12. Crow, L. H. "Tracking Reliability Growth," Interim Note No. R-30, U.S. Army Materiel Systems Analysis Agency, Aberdeen Proving Grounds, Maryland, April 1974.
13. Curtis, A. J.; Tinling, N. G., and Abstein, H. T. Jr. "Selection and Performance of Vibration Tests," Hughes Aircraft Company, Culver City, California, January 1970.

14. Doyle and Kapfer, "Failure Analysis: Its Role in Screening Decisions," Proceedings of the 1969 Annual Reliability and Maintainability Symposium, IEEE; New York, New York, 1969, p. 211.
15. Duane, J. T., "Learning Curve Approach to Reliability Monitoring," IEEE Transactions, Aerospace, Vol. 2, 1964.
16. "Environmental Stress Screening Studies (Phase I)," Hughes Aircraft Company, Report No. TIC 5150.77/1, May 1977.
17. Evans, Dr. Ralph A., "Literature Review Study on Accelerated Testing of Electronic Parts," Jet Propulsion Laboratory, California Institute of Technology, Pasadena, Final Report NASA-CR-97207, April 1968, p. 114.
18. Foster, Robert C., "How to Avoid Getting Burned with Burn-in," Circuits Manufacturing, Vol. 16, #8, Benwill Publishing Corporation, Brookline, Massachusetts, August 1976, p. 56.
19. Gironi, G., and Malberti, P., "A Burn-in Program for Wearout Unaffected Equipments," Microelectronics and Reliability, Vol. 15, #3, Pergamon Press, Elmsford, New York, 1976, p. 227.
20. Goldshine and Martin, "Component Defects and System Reliability," Proceedings of the 1973 Annual Reliability and Maintainability Symposium, IEEE; New York, New York, 1973, p. 214.
21. Green and Bailey, "Subcontracting for Parts Screening," Proceedings of the 1970 Annual Reliability and Maintainability Symposium, IEEE; New York, New York, 1970, p. 260.
22. Green, J. E., and Mead, P. H., "Experience Gained from Reliability Trials on an Airborne Radar," IEEE Reliability in Electronics, Vol. 68, London, England, December 1969, p. 52.
23. Haythornthwaite, Raymond F.; Molozzi, A. R., and Sulway, D. V., "Reliability Assurance of Individual Semiconductor Components," Proceedings of the IEEE, Vol. 62, #2, New York, New York, February 1974, p. 260.
24. Hirschberger, George and Dantowitz, Allan, "Evaluation of Environmental Profiles for Reliability Demonstration," Grumman Aerospace Corporation, Final Report, RAD-C-TR-75-242, September, 1975, B007946.
25. Ingram-Cotton; Sulway; and Leduc, "Is there a Reliable Screen for VHF Power Transistors?", Proceedings of the 1972 Annual Reliability and Maintainability Symposium, IEEE; New York, New York, 1972, p. 533.
26. Isken and Sabre, "Reliability Improvement through Effective Non-destructive Screening," Proceedings of the 1970 Annual Reliability and Maintainability Symposium, IEEE; New York, New York, 1970, p. 326.
27. Klass, Philip J., "Heating Speeds Microcircuit Screening," Aviation Week and Space Technology, McGraw-Hill; New York, New York, October 4, 1976, p. 57.

28. Krause, B. D., and Walters, N. A., "Accelerated Life Testing of Thick Film Resistors," 1972 NATO Conference Proceedings: Reliability Testing and Reliability Evaluation, California State University, Northridge, September 1972, p. VII-A.
29. Mead, P. H.; Gordon, J. R.; and Boreham, J. D., "Component Reliability in Simulated Aircraft Conditions," IEEE Conference #12, 1965, Components and Materials Used in Electronic Engineering, London, England, 1965, p. 27-1.
30. Michaelis, L. P., "Reliability Cost Effectiveness Through Parts Control and Standardization," IEEE Transactions on Parts, Materials and Packaging, Vol. PMP1, #1, New York, New York, June 1965, p. S327.
31. Miller, Lewis E., "Reliability of Semiconductor Devices for Submarine-Cable Systems," Proceedings of the IEEE, Vol. 62, #2, New York, New York, February 1974, p. 230.
32. Minner, E. S. and Romero, H. A., "Reliability Testing of F-111A Avionics Systems," Proceedings of the 1968 Annual Reliability and Maintainability Symposium, IEEE; New York, New York, 1968, p. 567.
33. Nalos, E. J., and Schulz, R. B., "Reliability and Cost of Avionics," IEEE Transactions on Reliability, Vol. R-14, #2, New York, New York, October 1965, p. 120.
34. Nowaks, T. J., "Reliability of Integrated Circuits by Screening," Proceedings of the 1967 Annual Reliability and Maintainability Symposium, IEEE; New York, New York, 1967, p. 365.
35. Parker and Lawson, "Comparison of DPA Results on Electronic Components," Proceedings of the 1976 Annual Reliability and Maintainability Symposium, IEEE; New York, New York, 1976, p. 456.
36. Quart, I., "Stress Screening Design Using the RAF Technique," Hughes Aircraft Company, Culver City, CA, April 1976.
37. Reich, H., "Components Behavior at Low Operating Stress Levels," IEEE Conference: Components and Materials Used in Electrical Engineering, London, England, 1965, p. 39.
38. Reynolds, F. H.; Parrott, R. W.; and Braithwaite, D., "Use of Tests at Elevated Temperatures to Accelerate the Life of a MOS Integrated Circuit," Proceedings of the IEEE, Vol. 118, #3, 4, London, England, March/April 1971, p. 475.
39. Reynolds, Frederik H., "Thermally Accelerated Aging of Semiconductor Components," Proceedings of the IEEE, Vol. 62, #2, New York, New York, February 1974, p. 212.
40. Rosner, Nathan, "System Analysis - Non-Linear Estimation Techniques," IBM Corporation, New York, 1965.

41. Rue, Herman D., "System Burn-in for Reliability Enhancement," Proceedings of the 1976 Annual Reliability and Maintainability Symposium, IEEE; New York, New York, 1976, p. 336.
42. Ryerson, C. M., Reliability CREDIT (Cost Reduction Early Decision Information Techniques), Hughes Aircraft Company, Culver City, Report No. TIC 20-42-732-R (P73-218), September 1973.
43. Ryerson, C. M., "Modern Basic Concepts in Component Part Reliability," Microelectronics and Reliability, Vol. 5, Pergamon Press, Great Britain, 1966, pp. 239-250.
44. Ryerson, C. M., "Relating Factory Test Failure Results to Field Reliability, Required Field Maintenance, and to Total Life Cycle Cost," Hughes Aircraft Company, Culver City, CA, June 1972.
45. Ryerson, C. M., "Relative Costs of Different Reliability Screening Techniques," Proceedings of the 1967 Annual Reliability and Maintainability Symposium, IEEE; New York, New York, 1967, p. 408.
46. Ryerson, C. M., "Reliability Testing and Screening - A General Review," Hughes Aircraft Company, Culver City, October 1975.
47. Schafer, R. E.; Sallee, R. B.; and Torrez, J. D., "Reliability Growth Study," Hughes Aircraft Company, Ground Systems Group, Fullerton, CA, Final Report No. TR 75-253, October 1975.
48. Shoeman, Martin L., Probabilistic Reliability: An Engineering Approach, McGraw-Hill, 1968.
49. Simoni, Arnold, "Component Reliability at Low Stress Levels and the Significance of Failure Mechanisms," IEEE Transactions on Parts, Materials and Packaging, Vol. PMP1, #1, New York, New York, June 1965, p. 303.
50. Singpurwalla, Nozer D., "Accelerated Life Testing, A Survey of Developments," 1972 NATO Conference Proceedings: Reliability Testing and Reliability Evaluation, California State University, Northridge, September 1972, p. VII-D.
51. "Stress Screening Experiment, Phase One," Hughes Aircraft Company, Report No. P76-385, October 1976.
52. "Stress Screening Studies," Hughes Aircraft Company, Culver City, CA, Report No. TIC 5/50.76-501, June 1976.
53. Vander Hamm, R. L., "Environmental Testing - The Key to High Reliability," Proceedings of the 1969 Annual Reliability and Maintainability Symposium, IEEE; New York, New York, 1969, p. 27.
54. Vanous, Donald D., "GARD - A New Era of Component Testing," IEEE Transactions on Parts, Materials and Packaging, Vol. PMP 1, #1, New York, New York, June 1965, p. 320.
55. Yadav, R. P. S., "A Reliability Model for Stress vs Strength Problem," Microelectronics and Reliability, Vol. 12, #2, Pergamon Press, Elmsford, New York, April 1973, p. 119.

## APPENDIX A

### A. 1 UCLA-Biomedical Computer Program (BMD)07R)

In the computation of Test Strength for Card/Unit Level and Equipment/System Level, and determining the Test Strength Relationship to Number of Subcontractors, the BMD07R computer program was used to find various parameter values. BMD07R is a nonlinear least squares regression program that incorporates with the canned routine the user supplied functional equation, partial derivatives, parameters and variables. Sample functions and variable values along with estimates for the parameters are used in an iteration scheme designed to converge to the parameter values. The UCLA BMD07R program is accessed by the IBM 370 system through use of the catalogued procedure, BMDT, which provides the necessary job control language. These programs use the FORTRAN IV language.

### A. 2 General Curve Plotting (QKPLOT):

A curve plot was given in Figure 4. 1 under the Computation of Test Strength for Card/Unit section. QKPLOT, or "Quick Plot" is a computer subroutine available in the IBM 370 Scientific Subprogram Library. A user can implement it by making a call to QKPLOT while providing the necessary points to be plotted. A logarithmic X (time) and linear Y = F(X) (test strength) set of axes was chosen and more than a thousand points were plotted. This program can be accessed using FORTRAN IV language.

### A. 3 Multiple Linear Regression (MLRG): Least Squares Fit and Analysis

Regression analyses were performed in the sections on Computation of Test Strength for Equipment/System Level and Conversion of MTBF to Defects, in which the computer program, MLRG, was used. MLRG, is a subroutine available in the IBM 370 Scientific Subprogram Library. A call to this routine with a set of observations of dependent and independent variables will cause it to compute the coefficients of a multiple linear equation expressing the dependent variable as a function of the independent ones. MLRG also calculates a set of statistical quantities such as the coefficient of multiple determination which provides a measure of the least-squares fit. This program can be accessed using the FORTRAN IV language.

### A. 4 Least Squares Curve Fit (CURFIT)

In the sections on Conversion of MTBF to Defects and Determining Relationship of Test Strength to Number of Subcontractors and Percent Non-Standard Parts, the computer program, CURFIT was used for the regression analysis. This program is available on the Dartmouth Time Sharing System (DTSS) and is written in the BASIC language. Using the input data values for the independent and dependent variables, the routine fits them to six different curve types, with output information of the measure of fit and the equation coefficients. The six curves used are:

- (1)  $Y = A + BX$ , linear
- (2)  $Y = Ae^{BX}$ , exponential

- (3)  $Y = AX^B$ , power  
 (4)  $Y = A + B/X$ , hyperbolic  
 (5)  $Y = 1/(A + BX)$ , hyperbolic  
 (6)  $Y = X/(A + BX)$ , hyperbolic

A<sub>x</sub> 5 SDO Computer Program Printout

```

INTEGER T                                         00000010
INTEGER*2 TV(20,300),SEQ(20,300),TV1(17000),SEQ1(17000) 00000020
DIMENSION N(5),ADEF(5),F(11,4,5),R1(11,4,5),R2(11,4,5),CT(5,5), 00000030
+CR1(5,5),CR2(5,5),TS(20,300),TC(20,300),NL(20),NF(20), 00000040
+P(11,4,5),TS1(17000),TC1(17000),SCOST(5,5),IARRAY(20), 00000050
+PARRAY(11,4,5) 00000060
DATA NF,TS,TC,TS1,TC1/20*0,46000*0./ 00000070
DATA PARRAY/220*1.0/ 00000080
DATA F,F,R1,R2/880*1.0/ 00000090
DATA N/5*5/ 00000100
CALC DATA (NCYC,M,PDEF,CREQD,E,ITV,N,ADEF, 00000110
+P,F,R1,R2,CR1,CR2,FRF,FRM,LEVEL,ITYP, 00000120
+AMAX11,AMIN11,AMAX12,AMIN12,AMAX21,AMIN21,AMAX22,AMIN22,AMAX23, 00000130
+AMIN23,AMAX24,AMIN24,AMAX31,AMIN31,AMAX32,AMIN32,AMAX33,AMIN33, 00000140
+AMAX34,AMIN34,AMAX41,AMIN41,AMAX51,AMIN51,AMAX52,AMIN52) 00000150
LL=0 00000160
ADIN=0.0 00000170
DO 1 I=1,M 00000180
1 ADIN=ADIN+ADEF(I) 00000190
DIN=ADIN+PDEF 00000200
IF(FRF.EQ.0.0) GO TO 5 00000210
OPTS=0.0 00000220
OPTS2=0.0 00000230
HOURS2=0.0 00000240
DO 220 I=1,3 00000250
GO TO (201,202,203),I 00000260
201 I2=2 00000270
GO TO 210 00000280
202 I2=3 00000290
GO TO 210 00000300
203 I2=5 00000310
210 CONTINUE 00000320
READ(13,*END=220) PMAX,CYCMAX 00000330
IF (PMAX.EQ.0.0.OR.CYCMAX.EQ.0.0) GO TO 220 00000340
CALL SSPIORB(ITV,ITV,ITV,ITV,M,I2,PARRAY,1,CT,0.,0.,ITV, 00000350
+HOURS,AMAX11,AMIN11,AMAX12,AMIN12,AMAX21,AMIN21,AMAX22,AMIN22, 00000360
+PMAX,AMIN23,CYCMAX,AMIN24,AMAX31,AMIN31,PMAX,AMIN32,AMAX33, 00000370
+AMIN33,CYCMAX,AMIN34,AMAX41,AMIN41,PMAX,AMIN51,CYCMAX,AMIN52) 00000380
OPTSI=PARRAY(1,M,I2) 00000390
OPTS=OPTS2+OPTSI*(1.0-OPTS2) 00000400
OPTS2=OPTSI 00000410
HOURS1=HOURS 00000420
HOURST=AMAX1(HOURS1,HOURS2) 00000430

```

```

HOURS2=HOURST          000004
PARRAY(1,M,I4)=1.0     00000450
220 CONTINUE           00000460
    GO TO 240          00000470
230 OPTS=1.0           00000480
    HOURST=26280.       00000490
240 IF (OPTS.EQ.0.0) GO TO 230      00000500
    DREQD=HOURST*((10.**(-6))* (FRF-FRM))/OPTS 00000510
    SREQD=1.0-DREQD/DIN   00000520
    WRITE(6,*) HOURST,SREQD 00000530
5 DO 50 I1=1,M          00000540
    N1=N(I1)            00000550
    DO 52 I2=1,N1        00000560
    LL=LL+1              00000570
    MV=0                 00000580
    READ(10,*,END=7) A1,B1 00000590
    GO TO 8              00000600
7 A1=0.0                00000610
    B1=0.0                00000620
8 DO 103 KK4=1,ITV       00000630
    DO 103 KK3=1,ITV       00000640
    DO 102 KK2=1,ITV       00000650
    DO 101 KK1=1,ITV       00000660
    MV=MV+1               00000670
    DO 10 I=1,NCYC         00000680
10 PARRAY(I,I1,I2)=P(I,I1,I2) 00000690
    CALL SSPPROB(KK1,KK2,KK3,KK4,I1,I2,PARRAY,NCYC,CT,A1,B1,ITV,HGJRS, 00000700
    +AMAX11,AMIN11,AMAX12,AMIN12,AMAX21,AMIN21,AMAX22,AMIN22,AMAX23, 00000710
    +AMIN23,AMAX24,AMIN24,AMAX31,AMIN31,AMAX32,AMIN32,AMAX33,AMIN33, 00000720
    +AMAX34,AMIN34,AMAX41,AMIN41,AMAX51,AMIN51,AMAX52,AMIN52) 00000730
    CALL SCRFN(NCYC,M,N,PDEF,DIN,ADEF,PARRAY,F,R1,R2,CT,CRL,CR2,SS, 00000740
    +TCOST,I1,I2,SCOST,TCMIN,0.0) 00000750
    TS(LL,MV)=SS          00000760
    TC(LL,MV)=TCOST       00000770
101 CONTINUE             00000780
    IF(I2-4) 102,104,102 00000790
102 CONTINUE             00000800
    IF((I2-1)*(I2-5)) 103,104,103 00000810
103 CONTINUE             00000820
104 CONTINUE             00000830
    DO 12 I=1,NCYC         00000840
12 PARRAY(I,I1,I2)=1.0   00000850
    N1(LL)=MV              00000860
    IF(LL-2) 52,15,20      00000870
15 K1=N1(LL-1)           00000880
    GO TO 30              00000890
20 K1=K
30 K2=N1(LL)
    NSEQ=K1*K2
    DO 40 J1=1,K1
    DO 40 J2=1,K2
    U=TS(LL,J2)*(1.0- TS(LL-1,J1))+TS(LL-1,J1)

```

```

V=TC(LL-1,J1)+(1.-TS(LL-1,J1))*TS(LL,J2)*TC(LL,J2)          00000960
36 NF(LL)=NF(LL)+1                                              00000970
JJ=NF(LL)                                                       00000980
SEQ1(JJ)=J1                                                       00000990
TV1(JJ)=J2                                                       00001000
TS1(JJ)=U                                                       00001010
TC1(JJ)=V                                                       00001020
40 CONTINUE
CALL RANK(TS1,TC1,NSEQ,SEQ1,TV1)                                 00001030
QC=0.0                                                       00001040
QS=0.0                                                       00001050
K=0                                                       00001060
DO 51 I=1,NSEQ
IF((TS1(I).EQ.0.0).OR.(TC1(I).EQ.0.0)) GO TO 51               00001070
IF((TS1(I)-QS)/TS1(I).LT.E) GO TO 51                           00001080
IF((TC1(I)-QC)/TC1(I).LT.E) GO TO 51                           00001090
QC=TC1(I)
QS=TS1(I)
K=K+1
TC(LL,K)=TC1(I)                                                 00001100
TS(LL,K)=TS1(I)                                                 00001110
SEQ(LL,K)=SEQ1(I)                                               00001120
TV(LL,K)=TV1(I)                                                 00001130
TC1(I)=0.                                                       00001140
TS1(I)=0.                                                       00001150
SEQ1(I)=0                                                       00001160
TV1(I)=0                                                       00001170
51 CONTINUE
52 CONTINUE
50 CONTINUE
CALL SEARCH(LL,K,CREQD,SREQD,SEQ,TV,TC,TS,TARRAY,TCMIN,TSMAX,DIN) 00001230
CALL REPORT(M,NCYC,N,PDEF,DIN,ADEF,P,F,R1,R2,CT,CR1,CR2,IARRAY,
+A1,B1,ITV,TCMIN,TSMAX,                                         00001240
+AMAX11,AMIN11,AMAX12,AMIN12,AMAX21,AMIN21,AMAX22,AMIN22,AMAX23, 00001250
+AMIN23,AMAX24,AMIN24,AMAX31,AMIN31,AMAX32,AMIN32,AMAX33,AMIN33, 00001260
+AMAX34,AMIN34,AMAX41,AMIN41,AMAX51,AMIN51,AMAX52,AMIN52)        00001270
DEBUG SUBCHK
END
SUBROUTINE SCREEN(NCYC,M,N,PDEF,DIN,ADEF,P,F,R1,R2,CF,CR1,CR2,
+SS,TCOST,I1,I2,SCOST,TCMIN,FLAG)                                00001280
DIMENSION X(11,4,5),P(11,4,5),F(11,4,5),R1(11,4,5),Q(20),
+R2(11,4,5),                                             00001290
+TCOSTL(5,5),COSTL(5),N(5),CT(5,5),CR1(5,5),ADEF(5),           00001300
+CR2(5,5),SCOST(5,5)                                           00001310
DO 10 I=1,NCYC
DO 10 J=1,M
NJ=N(J)
DO 10 K=1,NJ
10 X(I,J,K)=0.0
X(1,1,1)=ADEF(1)+PDEF
DO 35 J=1,M
NJ=N(J)

```

```

DO 35 K=1,NJ          00001480
IF(K-1) 20,20,32      00001490
20 IF(J-1) 35,35,31    00001500
31 X(I,J,K)=P(I,J-1,N(J-I))*X(I,J-I,N(J-1))+ADEF(J) 00001510
GO TO 35              00001520
32 X(I,J,K)=P(I,J,K-1)*X(I,J,K-1)                      00001530
35 CONTINUE            00001540
SUMX=0.                00001550
DO 95 I=1,NCYC        00001560
X(I+1,1,1)=0.0         00001570
50 DO 51 J=1,M         00001580
NJ=N(J)                00001590
DO 51 K=1,NJ           00001600
51 X(I+1,1,1)=X(I+1,1,1)+(1-P(I,J,K))*F(I,J,K)*(1-R1(I,J,K))*X(I,J,K) 00001610
DO 94 J=1,M           00001620
NJ=N(J)                00001630
DO 94 K=1,NJ           00001640
IF(X(I,J,K).LT.1.0) GO TO 94   00001650
IF(J.EQ.1.AND.K.EQ.1) GO TO 94   00001660
IF(J.GT.1.AND.K.EQ.1) GO TO 90   00001670
IF(J.GE.1.AND.K.GT.1) GO TO 80   00001680
GO TO 94               00001690
90 X(I+1,J,1)=P(I+1,J-1,N(J-1))*X(I+1,J-1,N(J-1))+
+(1-P(I,J,1))*(1-F(I,J,1))*(1-R2(I,J,1))*X(I,J,1) 00001700
GO TO 94               00001710
00001720
80 X(I+1,J,K)=P(I+1,J,K-1)*X(I+1,J,K-1)+(1-P(I,J,K))*(1-F(I,J,K))*
*(1-R2(I,J,K))*X(I,J,K) 00001730
00001740
94 CONTINUE            00001750
95 CONTINUE            00001760
SUM=0.0                00001770
DO 110 I=1,NCYC        00001780
NM=N(M)                00001790
SUM=SUM+X(I,M,NM)*P(I,M,NM) 00001800
110 CONTINUE            00001810
DOUT=SUM                00001820
SS=1.0-DOUT/DIN        00001830
DELTA=DIN-DOUT         00001840
ACOST=0.0               00001850
TCOST=0.0               00001860
DO 120 J=1,M           00001870
DO 120 K=1,5             00001880
120 SCOST(J,K)=0.0       00001890
IF(FLAG.NE.0.0) GO TO 123   00001900
DO 200 I=1,NCYC         00001910
CR=CR1(I1,I2)*F(I,I1,I2)+CR2(I1,I2)*(I.0-F(I,I1,I2)) 00001920
COST=CT(I1,I2)+CR*(I.0-P(I,I1,I2))*X(I,I1,I2) 00001930
200 ACOST=ACOST+COST     00001940
GO TO 205               00001950
126 DO 160 J=1,M         00001960
DO 160 K=1,5             00001970
SUMTS1=0.0               00001980
SUMTS2=0.0               00001990
DO 130 I=1,NCYC         00002000

```

```

        SUMTS1=SUMTS1+(1.-P(I,J,K))*X(I,J,K)          00002010
130  SUMTS2=SUMTS2+X(I,J,K)                      00002020
        SCOST(J,K)=SUMTS1/SUMTS2                  00002030
160  TOCOST=TOCOST+SCOST(J,K)                  00002040
        SUMTC=0.0                                00002050
        DO 140 J=1,M                            00002060
        DO 140 K=1,5                            00002070
        SCOST(J,K)=SCOST(J,K)*(TCMIN/TOCOST)    00002080
140  SUMTC=SUMTC+SCOST(J,K)                  00002090
        TOCOST=SUMTC                         00002100
        GO TO 300                           00002110
205  IF(DIN=DOUT) 210,210,220                00002120
210  DELTA=1.0                            00002130
220  TOCOST=ACOST/DELTA                  00002140
300  RETURN                               00002150
        DEBUG SUBCHK                         00002160
        END                                  00002170
        SUBROUTINE RANK(TS1,TC1,NSEQ,SEQ1,TV1) 00002180
        INTEGER*2 TV1(17000),SEQ1(17000)      00002190
        DIMENSION TS1(17000),TC1(17000)      00002200
        N2=NSEQ                           00002210
        M1=N2                            00002220
650  M1=INT(M1/2.)                         00002230
        IF(M1.EQ.0) GO TO 830               00002240
        H1=N2-M1                          00002250
        J=1                                00002260
690  I=J                                00002270
700  L1=I+M1                           00002280
        IF(TC1(I).LE.TC1(L1)) GO TO 800   00002290
        A1=TC1(I)                         00002300
        B1=TS1(I)                         00002310
        A2=SEQ1(I)                         00002320
        B2=TV1(I)                         00002330
        TC1(I)=TC1(L1)                   00002340
        TS1(I)=TS1(L1)                   00002350
        SEQ1(I)=SEQ1(L1)                 00002360
        TV1(I)=TV1(L1)                   00002370
        TC1(L1)=A1                        00002380
        TS1(L1)=B1                        00002390
        SEQ1(L1)=A2                        00002400
        TV1(L1)=B2                        00002410
        I=I-M1                           00002420
        IF(I.GE.1) GO TO 700              00002430
800  J=J+1                            00002440
        IF(J.LE.H1) GO TO 690            00002450
        GO TO 650                         00002460
        DEBUG SUBCHK                     00002470
830  RETURN                            00002480
        END                                00002490
        SUBROUTINE SEARCH(LL,N0,CREQD,SREQD,SEQ,TV,TC,TS,IARRAY,TOMIN,
+TSMAX,DIN)                         00002500
        INTEGER T,S,S1,S2                  00002510
        INTEGER*2 TV(20,300),SEQ(20,300)  00002520
                                            00002530

```

```

DIMENSION TC(20,300),TS(20,300),IARRAY(20)          00002540
DO I I=1,N0                                      00002550
TCTOT=TC(LL,I)*DIN*TS(LL,I)                      0000256
1 TC(LL,I)=TCTOT                                 00002570
IF(CREQD) 5,5,10                                00002580
5 IF(SREQD) 400,400,1S                           00002590
10 K=N0                                         00002600
7 IF(TC(LL,K)-CREQD) 20,20,25                  00002610
15 K=1                                         00002620
17 IF(TS(LL,K)-SREQD) 30,20,20                  00002630
20 S1=SEQ(LL,K)                                00002640
T=TV(LL,K)                                     00002650
IARRAY(LL)=T                                    00002660
TCMIN=TC(LL,K)                                00002670
TSMAX=TS(LL,K)                                00002680
K=1                                         00002690
GO TO 40                                       00002700
25 K=K-1                                       00002710
IF(K) 200,200,7                                00002720
30 K=K+1                                       00002730
IF(K=N0) 17,17,200                            00002740
40 I=LL-1                                      00002750
45 IF(I-1) 300,300,100                           00002760
100 S=SEQ(I,S1)                                00002770
T=TV(I,S1)                                    00002780
S2=S1                                         00002790
S1=S                                         00002800
IARRAY(I)=T                                  00002810
I=I-1                                         00002820
GO TO 45                                       00002830
200 WRITE(6,201)                                00002840
201 FORMAT(/IX,'REQUIREMENT CANNOT BE MET')
STOP
300 IARRAY(1)=S                                00002850
400 CONTINUE
    DEBUG SUBCHK
    RETURN
    END
    SUBROUTINE SSPROB(K1,K2,K3,K4,I1,I2,P,NCYC,CT,A1,B1,ITV,HOURS,
+AMAX11,AMIN11,AMAX12,AMIN12,AMAX21,AMIN21,AMAX22,AMIN22,AMAX23,
+AMIN23,AMAX24,AMIN24,AMAX31,AMIN31,AMAX32,AMIN32,AMAX33,AMIN33,
+AMAX34,AMIN34,AMAX41,AMIN41,AMAX51,AMIN51,AMAX52,AMIN52)
    DIMENSION P(11,4,5),CT(5,5)
    GO TO (10,20,30,40,50),I2
10 CONTINUE
C TEST ONE: CONSTANT TEMPERATURE(CT)
    E1=EXP(.0122*((AMAX11-AMIN11)*K1+AMIN11*ITV-AMAX11)/(ITV-1.)+
+273.0))
    E2=EXP(-((AMAX12-AMIN12)*K2+AMIN12*ITV-AMAX12)*.0000263*E1/
+(ITV-1.))
    PFT=.6*(1.0-E2)
    SS=1.0-PFT
    HOURS=((AMAX12-AMIN12)*K2+AMIN12*ITV-AMAX12)/(ITV-1.)

```

```

    IF(B1) 11,11,105                               00003070
11 GO TO(101,102,103,104),I1                  00003080
101 CT(1,I2)=(0.15)*HOURS                      00003090
      GO TO 60                                     00003100
102 CT(2,I2)=(0.15)*HOURS                      00003110
      GO TO 60                                     00003120
103 CT(3,I2)=HOURS                            00003130
      GO TO 60                                     00003140
104 CT(4,I2)=HOURS                            00003150
      GO TO 60                                     00003160
20 CONTINUE                                     00003170
C TEST TWO: CYCLED TEMPERATURE(CYT)
    TE=((ABS((AMAX21-AMIN21)*K1+AMIN21*ITV-AMAX21)/(ITV-1.)-25.)+ 00003180
    +ABS((AMAX22-AMIN22)*K2+AMIN22*ITV-AMAX22)/(ITV-1.)-25.0))+ 00003190
    +50.0)/2.0                                    00003200
    E1=EXP(.0122*(TE+273.0))                     00003210
    TT=((AMAX23-AMIN23)*K3+AMIN23*ITV-AMAX23)*4.5/(ITV-1.)          00003220
    E2=EXP(-((AMAX24-AMIN24)*K4+AMIN24*ITV-AMAX24)*TT*.0000263*E1/ 00003230
    +(ITV-1.))                                     00003240
    PFDT=.8*(1.0-E2)                            00003250
    IF ((AMAX21-AMIN21)*K1+AMIN21*ITV-AMAX21.EQ.(AMAX22 00003260
    +AMIN22)*K2+AMIN22*ITV-AMAX22) PFDT=0.        00003270
    SS=1.0-PFDT                                00003280
    IF((AMAX23-AMIN23)*K3+AMIN23*ITV-AMAX23) 5,5,7 00003290
7 HOURS=((AMAX21-AMIN21)*K1+AMIN21*ITV-AMAX21+(AMAX22-AMIN22)*K2+ 00003300
    +AMIN22*ITV-AMAX22)*((AMAX24-AMIN24)*K4+AMIN24*ITV-AMAX24)/ 00003310
    +(60.*((AMAX23-AMIN23)*K3+AMIN23*ITV-AMAX23)*(ITV-1.))          00003320
      GO TO 8                                     00003330
      5 HOURS=0.                                 00003340
8 IF(B1)21,21,105                           00003350
21 GO TO(201,202,203,204),I1                  00003360
201 CT(1,I2)=(0.15)*HOURS                      00003370
      GO TO 60                                     00003380
202 CT(2,I2)=(0.15)*HOURS                      00003390
      GO TO 60                                     00003400
203 CT(3,I2)=HOURS                            00003410
      GO TO 60                                     00003420
204 CT(4,I2)=HOURS                            00003430
      GO TO 60                                     00003440
30 CONTINUE                                     00003450
C TEST THREE: VIBRATION(VIB)
    TE=ABS((AMAX33-AMIN33)*K3+AMIN33*ITV-AMAX33)/(ITV-1.))+25.0 00003460
    E1=EXP(.0122*(TE+273.0))                     00003470
    E2=EXP(-(.0000789*((AMAX31-AMIN31)*K1+AMIN31*ITV-AMAX31)*((AMAX32- 00003480
    +AMIN32)*K2+AMIN32*ITV-AMAX32)*((AMAX34-AMIN34)*K4+AMIN34*ITV- 00003490
    +AMAX34)*E1)/(ITV-1.)**3)                   00003500
    PFV=.2*(1.0-E2)                            00003510
    SS=1.0-PFV                                00003520
    HOURS=((AMAX32-AMIN32)*K2+AMIN32*ITV-AMAX32)*((AMAX34-AMIN34)*K4+ 00003530
    +AMIN34*ITV-AMAX34)/(60.*((ITV-1.)**2))           00003540
    IF(B1) 31,31,105                           00003550
31 GO TO(301,302,303,304),I1                  00003560
301 CT(1,I2)=(0.15)*HOURS                      00003570
                                              00003580
                                              00003590

```

GO TO 60	00003600
302 CT(2,I2)=(0.15)*HOURS	00003610
GO TO 60	00003620
303 CT(3,I2)=HOURS	00003630
GO TO 60	00003640
304 CT(4,I2)=HOURS	00003650
GO TO 60	00003660
40 CONTINUE	00003670
C TEST FOUR: CONSTANT POWER(CP)	00003680
HOURS=((AMAX41-AMIN41)*K1+AMIN41*ITV-AMAX41)/(ITV-1.)	00003690
IF(I1.GT.2) GO TO 41	00003700
C CARD/MODULE LEVELS	00003710
PWR=664.965/(((AMAX41-AMIN41)*K1+AMIN41*ITV-AMAX41)/(ITV-1.)+	00003720
+664.965)	00003730
SS=PWR	00003740
IF(B1) 42,42,105	00003750
42 GO TO(401,402),I1	00003760
401 CT(1,I2)=(0.15)*HOURS	00003770
GO TO 60	00003780
402 CT(2,I2)=(0.15)*HOURS	00003790
GO TO 60	00003800
41 CONTINUE	00003810
C EQUIPMENT/SYSTEM LEVELS	00003820
PWR=3049.875/(((AMAX41-AMIN41)*K1+AMIN41*ITV-AMAX41)/(ITV-1.)+	00003830
+3049.875)	00003840
SS=PWR	00003850
IF(B1) 43,43,105	00003860
43 GO TO(403,404,403,404),I1	00003870
403 CT(3,I2)=HOURS	00003880
GO TO 60	00003890
404 CT(4,I2)=HOURS	00003900
GO TO 60	00003910
50 CONTINUE	00003920
C TEST FIVE: CYCLED POWER(CYP)	00003930
HOURS=((AMAX51-AMIN51)*K1+AMIN51*ITV-AMAX51)*((AMAX52-AMIN52)*K2+	00003940
+AMIN52*ITV-AMAX52)/(ITV-1.)**2	00003950
IF(I1.GT.2) GO TO 51	00003960
C CARD/MODULE LEVELS	00003970
R1=664.965/(((AMAX51-AMIN51)*K1+AMIN51*ITV-AMAX51)/(ITV-1.)+	00003980
+664.965)	00003990
PWRC=R1**((AMAX52-AMIN52)*K2+AMIN52*ITV-AMAX52)/(ITV-1.))	00004000
SS=PWRC	00004010
IF(B1) 52,52,105	00004020
52 GO TO(501,502),I1	00004030
501 CT(1,I2)=(0.15)*HOURS	00004040
GO TO 60	00004050
502 CT(2,I2)=(0.15)*HOURS	00004060
GO TO 60	00004070
51 CONTINUE	00004080
C EQUIPMENT/SYSTEM LEVELS	00004090
R1=3049.875/(((AMAX51-AMIN51)*K1+AMIN51*ITV-AMAX51)/(ITV-1.)+	00004100
+3049.875)	00004110
PWRC=R1**((AMAX52-AMIN52)*K2+AMIN52*ITV-AMAX52)/(ITV-1.))	00004120

```

SS=PWRC 00004130
IF(B1) 53,53,105 00004140
53 GO TO(503,504,503,504),I1 00004150
503 CT(3,I2)=HOURS 00004160
GO TO 60 00004170
504 CT(4,I2)=HOURS 00004180
GO TO 60 00004190
105 CT(I1,I2)=B1*HOURS+A1 00004200
60 CONTINUE 00004210
DO 70 I=1,NCYC 00004220
D=1.0-(1.0-SS)*P(I,I1,I2) 00004230
70 P(I,I1,I2)=D 00004240
D_BUG SUBCHK 00004250
RETURN 00004260
END 00004270
SUBROUTINE REPORT(M,NCYC,N,PDEF,DIN,ADEF,P,F,R1,R2,CT,CRL,CR2,
+ARRAY,A1,B1,ITV,TCMIN,TSMAX, 00004280
+AMAX11,AMIN11,AMAX12,AMIN12,AMAX21,AMIN21,AMAX22,AMIN22,AMAX23,
+AMIN23,AMAX24,AMIN24,AMAX31,AMIN31,AMAX32,AMIN32,AMAX33,AMIN33,
+AMAX34,AMIN34,AMAX41,AMIN41,AMAX51,AMIN51,AMAX52,AMIN52) 00004290
REAL LCOST(5) 00004300
DIMENSION NP(5,5,5),N(5),P(11,4,5),F(11,4,5), 00004310
+R1(11,4,5),R2(11,4,5),CR1(5,5),CR2(5,5),CT(5,5),SCOST(5,5), 00004320
+TEST(5),IR(5),IARRAY(20),ANP(5,5,5),ADEF(5) 00004330
REWIND 10 00004340
DATA NP/125*1/ 00004350
DATA LCOST/5*0./ 00004360
DATA TEST/'CT ','CYT ','VIB ','CP ','CYP '/ 00004370
DATA ANP/125*0./ 00004380
I=0 00004390
DO 65 I1=1,M 00004400
DO 65 I2=1,5 00004410
READ(10,*) A1,B1 00004420
I=I+1 00004430
IR(1)=IARRAY(I) 00004440
GO TO (30,50,50,20,30),I2 00004450
20 NP(I1,I2,1)=IARRAY(I) 00004460
GO TO 60 00004470
30 DO 33 K=2,3 00004480
IR(K)=MOD(IR(K-1),ITV**((3-K)) 00004490
IF(IR(K-1)) 32,32,31 00004500
31 NP(I1,I2,4-K)=INT(FLOAT(IR(K-1))/(FLOAT(ITV))**((3-K)+.999999)) 00004510
GO TO 33 00004520
32 NP(I1,I2,4-K)=ITV 00004530
33 CONTINUE 00004540
GO TO 60 00004550
40 DO 43 K=2,4 00004560
IR(K)=MOD(IR(K-1),ITV**((4-K)) 00004570
IF(IR(K-1)) 42,42,41 00004580
41 NP(I1,I2,5-K)=INT(FLOAT(IR(K-1))/(FLOAT(ITV))**((4-K)+.999999)) 00004590
GO TO 43 00004600
42 NP(I1,I2,5-K)=ITV 00004610
43 CONTINUE 00004620

```

```

      GO TO 60                               00004660
50 DO 53 K=2,5                           00004670
      1R(K)=MOD(IR(K-1),ITV**(5-K))        00004680
      IF(IR(K-1)) 52,52,51                  00004690
51 NP(I1,I2,6-K)=INT(FLOAT(IR(K-1))/(FLOAT(ITV)**(5-K)+.999999)) 00004700
      GO TO 53                                00004710
52 NP(I1,I2,6-K)=ITV                      00004720
53 CONTINUE                                00004730
      GO TO 60                                00004740
60 CALL SSPROB(NP(I1,I2,1),NP(I1,I2,2),NP(I1,I2,3),NP(I1,I2,4),I1,I2,00004750
      +P,NCYC,CT,A1,B1,ITV,HOURS,           00004760
      +AMAX11,AMIN11,AMAX12,AMIN12,AMAX21,AMIN21,AMAX22,AMIN22,AMAX23, 00004770
      +AMIN23,AMAX24,AMIN24,AMAX31,AMIN31,AMAX32,AMIN32,AMAX33,AMIN33, 00004780
      +AMAX34,AMIN34,AMAX41,AMIN41,AMAX51,AMIN51,AMAX52,AMIN52)       00004790
65 CONTINUE                                00004800
      CALL SCREEN(NCYC,M,N,FDEF,DIN,ADEF,P,F,R1,R2,CT,CRL,CR2,        00004810
      +SS,FOOST,I1,I2,SCOST,TCMIN,1.0)      00004820
      DO 70 I1=1,M                            00004830
      DO 70 I2=1,5                            00004840
70 LCOST(I1)=LCOST(I1)+SCOST(I1,I2)        00004850
      IQ=0                                     00004860
      WRITE(6,100)                            00004870
      DO 90 I1=1,M                            00004880
      DO 90 I2=1,5                            00004890
      ANP(I1,4,1)=((AMAX41-AMIN41)*NP(I1,4,1)+AMIN41*ITV-AMAX41)/ 00004900
      +(ITV-1.)                                00004910
      ANP(I1,1,1)=((AMAX11-AMIN11)*NP(I1,1,1)+AMIN11*ITV-AMAX11)/ 00004920
      +(ITV-1.)                                00004930
      ANP(I1,1,2)=((AMAX12-AMIN12)*NP(I1,1,2)+AMIN12*ITV-AMAX12)/ 00004940
      +(ITV-1.)                                00004950
      ANP(I1,5,1)=((AMAX51-AMIN51)*NP(I1,5,1)+AMIN51*ITV-AMAX51)/ 00004960
      +(ITV-1.)                                00004970
      ANP(I1,5,2)=((AMAX52-AMIN52)*NP(I1,5,2)+AMIN52*ITV-AMAX52)/ 00004980
      +(ITV-1.)                                00004990
      ANP(I1,2,1)=((AMAX21-AMIN21)*NP(I1,2,1)+AMIN21*ITV-AMAX21)/ 00005000
      +(ITV-1.)                                00005010
      ANP(I1,2,2)=((AMAX22-AMIN22)*NP(I1,2,2)+AMIN22*ITV-AMAX22)/ 00005020
      +(ITV-1.)                                00005030
      ANP(I1,2,3)=((AMAX23-AMIN23)*NP(I1,2,3)+AMIN23*ITV-AMAX23)/ 00005040
      +(ITV-1.)                                00005050
      ANP(I1,2,4)=((AMAX24-AMIN24)*NP(I1,2,4)+AMIN24*ITV-AMAX24)/ 00005060
      +(ITV-1.)                                00005070
      ANP(I1,3,1)=((AMAX31-AMIN31)*NP(I1,3,1)+AMIN31*ITV-AMAX31)/ 00005080
      +(ITV-1.)                                00005090
      ANP(I1,3,2)=((AMAX32-AMIN32)*NP(I1,3,2)+AMIN32*ITV-AMAX32)/ 00005100
      +(ITV-1.)                                00005110
      ANP(I1,3,3)=((AMAX33-AMIN33)*NP(I1,3,3)+AMIN33*ITV-AMAX33)/ 00005120
      +(ITV-1.)                                00005130
      ANP(I1,3,4)=((AMAX34-AMIN34)*NP(I1,3,4)+AMIN34*ITV-AMAX34)/ 00005140
      +(ITV-1.)                                00005150
      GO TO (101,102,103,104,105),I2          00005160
101 IF(NP(I1,1,2).EQ.1) GO TO 76          00005170
      GO TO 106                                00005180

```

```

102 IF(NP(I1,2,3).EQ.1) GO TO 76          00005190
    GO TO 106
103 IF(NP(I1,3,2).EQ.1) GO TO 76          00005200
    GO TO 106
104 IF(NP(I1,4,1).EQ.1) GO TO 76          00005210
    GO TO 106
105 IF(NP(I1,5,1).EQ.1) GO TO 76          00005220
    GO TO 106
    76 DO 77 K=1,4                         00005230
    77 ANP(I1,I2,K)=0.                      00005240
106 IF(I1-IQ) 80,80,75                     00005250
    75 WRITE(6,110) I1,LCOST(I1)            00005260
    80 WRITE(6,120) I2,TEST(I2),(ANP(I1,I2,K),K=1,4),SCOST(I1,I2)
    90 IQ=I1
        WRITE(6,130) FOOST
100 FORMAT('1',28X,'T E S T   D E S C R I P T I O N ',/
+39X,'PARAMETER VALUE'/10X,'TEST SEQUENCE',5X,'TYPE',2X,/
+'NO. 1',2X,'NO. 2',2X,'NO. 3',2X,'NO. 4',2X,'TOTAL COST'/
+7X,66('_')//)
110 FORMAT(' ',8X,'LEVEL',2X,'NO. ',I2,39X,F12.2)      00005340
120 FORMAT(' ',12X,'TEST NO. ',I2,3X,A4,1X,4F7.2,F12.2) 00005350
130 FORMAT(' ',7X,66('_')//,' ',8X,'TOTAL',' COST',43X,F12.2//',',/
+61X,12('_')/',' ',61X,12('_'))                  00005360
    WRITE(6,140) TSMAX
140 FORMAT(' ',8X,'TOTAL SEQUENCE TEST STRENGTH=',F8.6) 00005370
    DEBUG SUBCHK
    RETURN
    END
    SUBROUTINE DATA (NCYC,M,PDEF,CREQD,E,ITV,N,ADEF,
+P,F,R1,R2,CR1,CR2,FRF,FRM,LEVEL,ITYP,          00005380
+AMAX11,AMIN11,AMAX12,AMIN12,AMAX21,AMIN21,AMAX22,AMIN22,AMAX23, 00005390
+AMIN23,AMAX24,AMIN24,AMAX31,AMIN31,AMAX32,AMIN32,AMAX33,AMIN33, 00005400
+AMAX34,AMIN34,AMAX41,AMIN41,AMAX51,AMIN51,AMAX52,AMIN52)      00005410
    DIMENSION N(5),ADEF(5),P(11,4,5),              00005420
+F(11,4,5),R1(11,4,5),R2(11,4,5),CR1(5,5),CR2(5,5)          00005430
    READ(4,*) NCYC,M,NPARTS,CREQD,E,ITV,FRF,FRM          00005440
    READ(4,*),END=2) PDEF
    IF(PDEF) 2,2,1
2 PDEF=0.01*NPARTS
    G1 TO 3
1 PDEF=PDEF*NPARTS
3 DO 5 I=1,M
    READ(4,*),END=4) ADEF(I)
    IF(ADEF(I)) 4,4,5
4 ADEF(I)=(7./(3.*M))*PDEF
    GO TO 7
5 ADEF(I)=ADLF(I)*NPARTS
7 WRITE(6,98)
    WRITE(6,90) NCYC,M,PDEF,CREQD,E,ITV,FRF,FRM
    WRITE(6,99)
    WRITE(6,91) (I,N(I),ADEF(I),I=1,M)
    DO 30 I1=1,M
    DO 30 I2=1,S

```

```

      READ(11,* ,END=10) P(I,I1,I2)          00005720
      IF (P(I,I1,I2)) 10,10,20
10 DO 15 I=1,NCYC
15 P(I,I1,I2)=.75
      GO TO 30
20 DO 25 I=1,NCYC
25 P(I,I1,I2)=P(I,I1,I2)
30 CONTINUE
      WRITE(6,92)
      WRITE(6,*) ((P(I,J,K),K=1,5),J=1,M)
      DO 40 I1=1,M
      DO 40 I2=1,5
      READ(8,* ,END=45) F(I,I1,I2)
      IF (F(I,I1,I2)) 45,45,47
45 DO 46 I=1,NCYC
      F(I,1,I2)=1.0
      F(I,2,I2)=0.23
      F(I,3,I2)=0.43
46 F(I,4,I2)=0.65
      GO TO 40
47 DO 48 I=1,NCYC
48 F(I,I1,I2)=F(I,I1,I2)
49 CONTINUE
      WRITE(6,93)
      WRITE(6,*) ((F(I,J,K),K=1,5),J=1,M)
      DO 80 I1=1,M
      DO 80 I2=1,5
      READ(9,* ,END=55) R1(I,I1,I2),R2(I,I1,I2)
      IF (R1(I,I1,I2)) 55,55,60
55 DO 56 I=1,NCYC
56 R1(I,I1,I2)=0.5
      GO TO 61
60 DO 57 I=1,NCYC
57 R1(I,I1,I2)=R1(I,I1,I2)
61 IF (R2(I,I1,I2)) 65,65,70
65 DO 66 I=1,NCYC
66 R2(I,I1,I2)=0.5
      GO TO 80
70 DO 67 I=1,NCYC
67 R2(I,I1,I2)=R2(I,I1,I2)
80 CONTINUE
      WRITE(6,94)
      WRITE(6,*) ((R1(I,J,K),K=1,5),J=1,M)
      WRITE(6,95)
      WRITE(6,*) ((R2(I,J,K),K=1,5),J=1,M)
      DO 300 I1=1,M
      DO 300 I2=1,5
      READ(4,* ,END=100) B2,B3
      IF (B2) 100,100,105
100 GO TO (101,102,103,104),I1
101 CR1(I,I2)=0.5
      GO TO 150
102 CR1(2,I2)=9.46

```

```

GO TO 150                                00006250
103 CRL(3,I2)=51.5                      00006260
    GO TO 150                                00006270
104 CRL(4,I2)=63.0                      00006280
    GO TO 150                                00006290
105 CRL(I1,I2)=B2                        00006300
150 IF (B3) 200,200,110                  00006310
200 GO TO (201,202,203,204),11          00006320
201 CR2(1,I2)=0.1                      00006330
    GO TO 300                                00006340
202 CR2(2,I2)=3.67                     00006350
    GO TO 300                                00006360
203 CR2(3,I2)=45.5                     00006370
    GO TO 300                                00006380
204 CR2(4,I2)=57.0                     00006390
    GO TO 300                                00006400
110 CR2(I1,I2)=B3                      00006410
300 CONTINUE                               00006420
    WRITE(6,96)                            00006430
    WRITE(6,*) ((CRL(J,K),K=1,5),J=1,M) 00006440
    WRITE(6,97)                            00006450
    WRITE(6,*) ((CR2(J,K),K=1,5),J=1,M) 00006460
DO 89 I=1,NCYC                           00006470
DO 89 J=1,M                                00006480
NJ=N(J)                                    00006490
DO 89 K=1,NJ                                00006500
    F(I,J,K)=F(I,J,K)                    00006510
    R1(I,J,K)=R1(I,J,K)                    00006520
89 R2(I,J,K)=R2(I,J,K)                    00006530
C TEST ONE: CONSTANT TEMPERATURE        00006540
READ(12,*) AMAX11,AMIN11                 00006550
IF(AMAX11.NE.0..OR.AMIN11.NE.0.) GO TO 1001 00006560
AMAX11=70.                                 00006570
AMIN11=-55.                               00006580
1001 READ(12,*) AMAX12,AMIN12            00006590
IF(AMAX12.NE.0..OR.AMIN12.NE.0.) GO TO 1002 00006600
AMAX12=170.                               00006610
AMIN12=0.                                 00006620
1002 CONTINUE                               00006630
C TEST TWO: CYCLED TEMPERATURE          00006640
READ(12,*) AMAX21,AMIN21                 00006650
IF(AMAX21.NE.0..OR.AMIN21.NE.0.) GO TO 2001 00006660
AMAX21=70.                                00006670
AMIN21=25.                                00006680
2001 READ(12,*) AMAX22,AMIN22            00006690
IF(AMAX22.NE.0..OR.AMIN22.NE.0.) GO TO 2002 00006700
AMAX22=-50.                               00006710
AMIN22=-25.                               00006720
2002 READ(12,*) AMAX23,AMIN23            00006730
IF(AMAX23.NE.0..OR.AMIN23.NE.0.) GO TO 2003 00006740
AMAX23=10.                                00006750
AMIN23=0.                                 00006760
2003 READ(12,*) AMAX24,AMIN24            00006770

```

IF(AMAX24.NE.0..OR.AMIN24.NE.0.) GO TO 2004	00006780
AMAX24=64.	00006790
AMIN24=10.	00006800
2004 CONTINUE	00006810
C TEST THREE: VIBRATION	00006820
READ(12,*) AMAX31,AMIN31	00006830
IF(AMAX31.NE.0..OR.AMIN31.NE.0.) GO TO 3001	00006840
AMAX31=6.	00006850
AMIN31=1.	00006860
3001 READ(12,*) AMAX32,AMIN32	00006870
IF(AMAX32.NE.0..OR.AMIN32.NE.0.) GO TO 3002	00006880
AMAX32=2.	00006890
AMIN32=0.	00006900
3002 READ(12,*) AMAX33,AMIN33	00006910
IF(AMAX33.NE.0..OR.AMIN33.NE.0.) GO TO 3003	00006920
AMAX33=25.	00006930
AMIN33=25.	00006940
3003 READ(12,*) AMAX34,AMIN34	00006950
IF(AMAX34.NE.0..OR.AMIN34.NE.0.) GO TO 3004	00006960
AMAX34=64.	00006970
AMIN34=10.	00006980
3004 CONTINUE	00006990
C TEST FOUR: CONSTANT POWER	00007000
READ(12,*) AMAX41,AMIN41	00007010
IF(AMAX41.NE.0..OR.AMIN41.NE.0.) GO TO 4001	00007020
AMAX41=168.	00007030
AMIN41=0.	00007040
4001 CONTINUE	00007050
C TEST FIVE: CYCLED POWER	00007060
READ(12,*) AMAX51,AMIN51	00007070
IF(AMAX51.NE.0..OR.AMIN51.NE.0.) GO TO 5001	00007080
AMAX51=8.	00007090
AMIN51=0.	00007100
5001 READ(12,*) AMAX52,AMIN52	00007110
IF(AMAX52.NE.0..OR.AMIN52.NE.0.) GO TO 5002	00007120
AMAX52=168.	00007130
AMIN52=3.	00007140
5002 CONTINUE	00007150
98 FORMAT('1',28X,'TABLE A - PROGRAM DATA'/' ',80(' ')/' ', +4X,'NCYC',11X,'M',6X,(PDEF.NPARTS)',4X,'CREQD',10X, +'E',11X,'ITV',10X,'FRF',10X,'FRM'/' ', +130(''-')//)	00007160 00007170 00007180 00007190
99 FORMAT('1',24X,'TABLE B - TEST AND ASSEMBLY DATA'/' ',80(' ')/' ', +4X,'ASSEMBLY LEVEL',2X,'NUMBER OF SCREENS',2X, +'EXPECTED NUMBER OF ASSEMBLY DEFECTS'/' ',4X,14(' '_'),2X,17(' '_'),2X00007220 ,35(' ')//)	00007200 00007210 00007220 00007230
90 FORMAT(' ',5X,I2,12X,I1,6X,F13.0,F13.2,5X,F8.6,5X,I3,5X,2F13.2)	00007240
91 FORMAT(' ',7X,I2,14X,I2,22X,F9.0)	00007250
92 FORMAT('1',14X,'TABLE C - DETECTION PROBABILITIES FOR TEST', +' EQUIPMENT'/' ',80(' ')//)	00007260 00007270
93 FORMAT('1',4X,'TABLE D - FRACTION OF DEFECTS REQUIRING ', +'PRIMARY LEVEL (CARD ASSY) REVORK'/' ',80(' ')//)	00007280 00007290
94 FORMAT('1',20X,'TABLE E-FRACTION OF DEFECTS REMOVED AT'/' ',	00007300

```
+19X,'THE PRIMARY LEVEL PER REWORK/RETEST CYCLE'          00007310
+/' ',80(' ')//)                                            00007320
95 FORMAT('1T,14X,'TABLE F - FRACTION OF DEFECTS REMOVED AT ', 00007330
+'THE HIGHER'/' ',20X,'ASSEMBLY LEVEL PER REWORK',        00007340
+'RETEST CYCLE'/' ',80(' ')//)                                00007350
96 FORMAT('1',18X,'TABLE G - PRIMARY LEVEL REWORK COST PER', 00007360
+' CYCLE'/' ',80(' ')//)                                    00007370
97 FORMAT('1',13X,'TABLE H - HIGHER ASSEMBLY LEVEL REWORK ', 00007380
+'COST PER CYCLE'/' ',80(' ')//)
      RETURN
      END
```